### **EE 8002 DESIGN OF ELECTRICAL APPARATUS**

## <u>UNIT I</u>

### **DESIGN OF FIELD SYSTEM AND ARMATURE**

Prepared by Dr . T. Dharma Raj Asso.Prof /EEE 1.1 Major considerations in Electrical Machine Design.

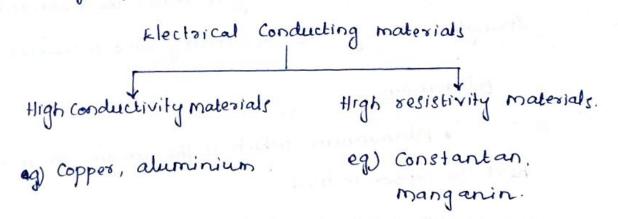
The major considerations to evolve a good design are i) cost ii) Durablety iii) Compliance with performance Criteria as Jaid down in Specifications

#### 12 Electrical Engineering Materials.

The Electrical Engineering materials are divided into following groups

- a) Electrical conducting Materials
- b) Electrical Carbon Malerials
- c) Magnetic Materials
- d) Insulating Materials.

a) <u>Flectrical Conducting Materials</u> The Electrical Conducting materials are divided into two groups.



#### High conductivity materials.

These materials are used for making all types of windings required in electrical machines, apparatus and devices. These materials should have the least possible resistivity.

The fundamental Requisements of high conductivity materials are

- i) least resistivity
- ii) high conductivity
- iii) low temperature coefficient of resistance.
- iv) rollability and drawability
- v) adequate resistance to corresion
- vi) good weldablity and salderability.

#### Copper

Hard drawon copper wires are used electrical machines as wire drawing increases the mechanical Strength although the resistivity also increases a little.

#### Aluminium

\* Aluminium which is the conductor material heat to copper is used

\* Freely available conductor material.

\* Pure Aluminium is softer than copper and therefore it can be rolled into thin sheets. High resistivity materials.

\* Used for making resistances and heating device. \* conductors of high resistance are used where it is desired to dissipate electrical energy into heat.

Materials of high registivity are primarily alloys of different metals. They are nickel, silver and iron This alloys can be classified according to their purpose. The three categories are.

i) The first group consists of materials used the precision.
 Work ie) measuring instruments, standard resistances and boxes.
 The material used for this group is Manganin. it consists
 of (u → 86%, Mn → 12%, Ni → 2%.
 (capper) (Magganin) (Nickel)

ii) The second group consists of materials used for Rheostats and control devices.

The material used for this group is constantan. It Consists of cu bo to 65% and 40 to \$. % of Nickel.

iii) The third group consists of materials used for heating devices ie) electric furnaces, loading Rheostals.

The material used for this group is alloys of nickel, chrominum and iron called Nichrome and alloy of aluminium iron and chromium.

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\* Aluminium cannot be drawn into very fine wires, due to its low mechanical Strength.

\* For induction motors with power outputs upto looker, aluminium can be used as material for bars and squirrel cage. Die cast aluminium windings are extensively used for rotors of induction motors.

\* Super enamelled aluminium wires are used for Stator windings of small induction motors.

\* Aluminium is also used for windings of transformers. due to decrease of overall cost of the transformer.

\* The use of aluminium in transformer tanks instead of Steel tanks reduces the weight and stray load loss.

Alloys of copper copper silver alloy Bronze Brass i) cadmium copper i) used in the manufacture i) used in turbo \* 1618 used for making contact alternators, because of electrical apparetus wires and commutator segments of its resistance bo ii) it generally contains \* it is also used for Cage thermal chortening 66% cu and 34% zinc. windings. and creep.

ii) Beryllium Copper

\* It is used for current Carrying Springs, brush holders, sliding contacts and knife switch blades.

# b) Electrical Carbon materials.

Flectrical Carbon materiale are manufactured from graphile and other forms of Carbon coal etc.

The conductivity of carbon used is slightly less than metals and alloys and these fore it is used for making brushes for electrical machines.

Brush Carbons are after graphited to raise the Conductivity of the brushes and reduces their hordness.

### c) Magnetic Materials

Fill magnetic materials posses magnetic properties to a greater (0x) a lesser degree. The magnetic properties of materials are characterized by their relative permeablity. Based on relative permeablity, the materials are classified into three types.

Ferromagnetic materials : The relative permeablily is much greater that unity and these permeablity values are dependent upon the magnetizing force.

Paramagnetic materials : These materials having the only Slightly tess than unity. The value of Susceptibility is thus positive for these materials.

Diamagnetic materials: These materials having dis only slightly less than unity. Both paramagnetic and Diamagnetic materials the value of permeablily is independent of the magnetising force.

Types of magnetic materials.

soft magnetic materials

Hard magnetic materials

> Solid cose materials

-> Electrical Sheet and Strip

-> Special purpose alloys

Materials which are easy to magnetise and demagnetise are called <u>soft magnetic materials</u>. These materials are used for making temporary magnets. These materials are used for making temporary magnets. Solid core materials are used for parts of magnetic

Circuite Carrying steady flux such as corres of de electro. Magnets, relays and field frames of de machines.

Electrical sheet materials are used for the magnetic circuits of electrical machines and the cores of the transformers was iron with a low content of Carbon. and other impurities. This had one major disadvantage. that of ageing

Ageing is the term used to denote the deterioration of magnetic performance in service, caused by increase in Coexcive force and hysteresis loss which inturn caused Cumulative overheating and subsequent breakdown.

In votating electrical machines, we use of steels with low silicon content are termed as dynamo grade steel.

The sheet steels "possessing higher silicon content (4-5%) silicon are called transformer grade steels.

The cold Rolled Grain oriented Steel (CR410) is manufactured by a series of Cold reductions and intermediate annealings. This Cold reduced material has strong directional magnetic properties, the rolling direction being the direction of highest permeability. This direction is also the direction of lowest hysteresis loss

### Hard magnetic materials

materials which retain their magnetism and are difficult to demagnetise are called hard magnetic materials. These material retain their magnetism even after the removal of the applied magnetic field.

d) Insulating materials.

An ideal insulating material should have the following properties

\* high dielectric strength

\* gh resistivity or specific resistance.

- \* low dielectric hysteresis
- \* good thermal Conductivity
- \* high degree of thermal stability.

The following insulating materials are used in modern electrical machines.

mica, micafolium, Fibrous glass, At Asbestos, Cotton fibre, Synthetic Resin, Petroleum based mineral oils.

Applications of insulating material.

- \* wives for magnet coils and windings of machines.
- \* lamitions
- \* machines and transformers.

Insulating material for laminations

The following are the common insulating materials for Laminations.

i) Insuline

This is a kaolin mixture which is sprayed on to one or both sides of the lamination.

ii) Oxide.

A natural oxide <del>cor</del> coating is formed on the Sheets during hot rolling process, but this insulation Cannot be depended upon as it may be inadequate.

iii) Varnish.

it makes the laminations rust proof and it is

not affected by temperature produced in electrical Machines.

#### Insulating materials for machines

De and AC motors and generators for industrial purposes are usually to insulated with class A or E materials, but turbo generators, traction motors and aircraft machines are insulated with class B materials to enable higher operating temperatures to be used for the purpose of obtaining larger output from a given frame Size.

Insulating materials for transformers.

\* Fibrous materials are employed for both air cooled and oil cooled transformers.

\* Cotton or oiled cambric is used for taping the Coils of air cooled transformer.

\* Synthetic resin bonded paper is used for the insulation between core and coils and also blue primary and secondary winding.

packing blue coils.

# Classification of Insulating Materials

class	Temperature	materials
Class y (o)		Cotton, Silk, paper, wood Cellulose

Class A materials of class 105° c impreganeted with natural resing Cellulose. CLASS E 120°C Synthetic resign enamels, Cotton and paper lamination mica glass, fibre, 130°C CLASS B with suitable bonding Substances. 155 C material of class B with Class F bonding Substances. Glass Fibre and a Class H 180 c material with Silicon resins mica, Ceramics, glass, Quarte above 180 c class c without binders with ray have a rapid paper of which for the

and at poons is could and an prover and

Magnetic eiscuit Calculations

9) Magnetic ciacuít

The path of Magnetic flyz is Called Magnetic Ciscuit.

consider a coil with n no :. ~ I of turns is wounded on a core ф 18 Connected to Supply Voltage'V' The current flows through the Coil Creates the momf and this mong produce a flux in the Important terms ysed in magnetic ciscuit i) MMF( magnetic motive force). It is the ability of coil to produce magnetic flux mme=NI, also 2 -> 1 Its unit is Amperel turns (AT). i) Magnetic field Intensity (or) Magnetising fosce ( It or at) It is defined as monf per unit length of flux path (Horat) = mmf From Equation () Flux density B  $H = \frac{NI}{I}$ B= \$/A 4=BA  $H = \frac{43}{1}$ Reluctance S H= BAS S= L LA  $H = \frac{BAL}{4A} \Rightarrow H = \frac{B}{4}$ 

b) mmf Calculation for Aligap

mmf Calculation for Air gap is not easy. because of the following challenges.

i) one on both of the bron surfaces around the aisgap may be slotled

→ Flux tends to concentrate on teeth

> Mon uniform distribution of flux in the air gap.

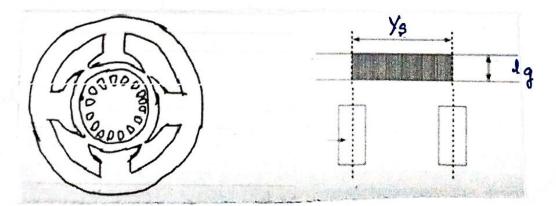
ii) There are radial ventilating ducts for cooling > Contraction of flux in the axial direction

11) Salient pole machines

> non uniform ais gap.

Because of the above Challenges, the reluctance of the aisgap is not ceniform, so it varies depends on the type of slots used.

<u>Case</u>!: Smooth iron Surfaces on both sides of the angap (Flux distribution is uniform)

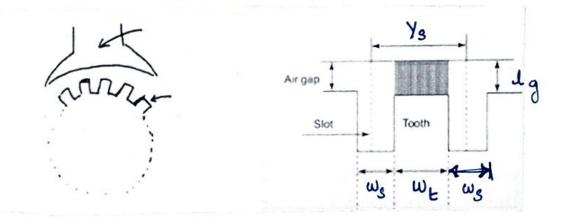


Reluctance of airgap  $\delta q = \frac{lq}{4A} = \frac{lq}{40A} = \frac{lq}{40A}$ 

From the above figure

A= Lys

Reluctance of algap of Smooth Armature Sg = lg Ho LYS.



Here we see flux is concentrated on toothwidth Therefore the effective area of flux is decreased ie) The reluctance of aig gap is increased.

 $W_s = \delta lot \ width (00) \delta lot opening$  $<math>w_t = tooth \ width$   $Y_s = \delta lot pitch = \frac{\pi D}{s}$ From the figure,  $Y_s = w_t + w_s/2 + w_s/2$ 

#### $Y_{g} = \omega_{t} + \omega_{s}$

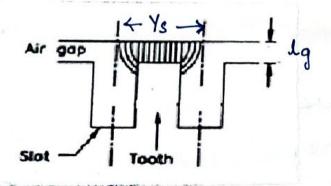
Reluctance of avagap of slotted Armature Sg = ig

From the figure,  $A = LY_s$   $Y_s' = W_t$ (or.)  $Y_s' = Y_s - W_s$ 

Reluctance of airgap of flotted Armature

$$Sg = \frac{Jq}{UoLY_s} = \frac{Jg}{UoL(Y_s - W_s)}$$

In slotted Armature, the flux will not pass like a Straight line towards the tooth. But in practical, the flux would fringe around the tooth and this fringing of flux could increase the area of Cross Section of flux path.



From the Figure,  $A = LY_s'$   $Y_s' = \omega_t + \delta \omega_s$   $= \omega_t + \delta \omega_s + \omega_s - \omega_s$  $= (\omega_t + \omega_s) + \delta \omega_s - \omega_s$ 

$$Y_{s}^{1} = Y_{s} + \omega_{s}(s-1)$$
  
 $Y_{s}^{1} = Y_{s} - \omega_{s}(1-s)$   
 $Y_{s}^{1} = Y_{s} - \omega_{s}k_{cs}$ 

where kes is the Caster's Coefficient for slots. It depends on the ratio Slot width gap length.

For parallel sided open slots (Induction motor)

$$k_{es} = \frac{2}{\pi} \left[ tas' y - \frac{1}{y} \log \sqrt{1 + y^2} \right]$$
(radian  
(radian  
mode) Where  $y = \frac{w_s}{alg}$   
Therefore,  $A = L(Y_s - w_s k_{es})$ 

Reluctance of air gap with Slotted almature is  

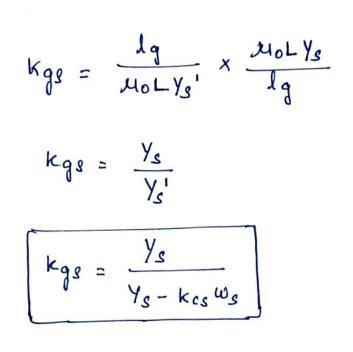
$$S_g = \frac{l_q}{a_{10} \perp \gamma_s^1} = \frac{l_g}{\mu_{0} \perp (\gamma_s - k_{cs} \omega_s)}$$

Gap contraction factor for slots is (kgs) is defined

$$kgs = \frac{lq}{lq}$$

$$Kgs = \frac{lq}{lq}$$

$$HoL Ys$$



<u>Case-3</u>: Effect of Ventilating Ducts Air gop Core stock Ducts, Wg wide

From the figure, it is seen that Length of the Core is reduced due to the presence of radial Ventialating Ducts

Eartes Coefficient of Ducts depends on the ratio Duct width Aisgaplength

The empirical formula used for parallel sided open. Slots is

$$k_{cd} = \frac{2}{\pi} \left[ \tan^{2} y - \frac{1}{y} \log \sqrt{(1+y^{2})} \right]$$
  
where  $y = \frac{\omega_{d}}{a \lg_{1}}$ 

Gap contraction factor for ducts (kgd) is defined as the ratio of

kgd = Reluctance of airgap of Slotted armature with ducts

Peluctance of aixgap of Slotted armature

without Ducts.

$$K_{gd} = \frac{J_{g}}{J_{0}L'Y_{s}'}$$

$$M_{0}L'Y_{s}'$$

$$kgd = \frac{lg}{u_{0L}'Y_{s}'} \times \frac{u_{0L}Y_{s}'}{lg}$$

$$k_{gd} = \frac{L}{L^{1}}$$

$$k_{gd} = \frac{L}{L - k_{cd} n d \omega d}$$

Total gap contraction factors (kg) is defined as the ratio of

Reluctance of airigap of slotted Armature kg = \_\_\_\_\_ with Ducts

Reluctance of alsgap of Smooth Armature

= 683

$$x_{g} = \frac{d_{g}}{d_{0}L'Y_{s}'}$$

$$\frac{d_{g}}{d_{0}L'Y_{s}}$$

$$\frac{d_{g}}{d_{0}L'Y_{s}}$$

$$kg = \frac{lg}{H_{0L}'Y_{s'}} \times \frac{H_{0}LY_{s}}{lg}$$

$$k_{g} = \frac{L}{L'} \times \frac{Y_{s}}{Y_{s}'}$$

$$k_{g} = k_{gd} \times k_{gs}$$

when both statog and rotog are slotted, then

where

kgss → gap Contraction factor for stator flots. kgsr → gap Contraction factor for rotor blots. Therefore this gap contraction factor which explains the relation b/w reluctance of Smooth armature and the reluctance Slotted almature with and without ducts. By using this factor, we can easily find the mmf required for air gap.

Calculation for mmf

we know that

a

mmf/unit length is called magnetising force (Horat)

$$t \circ H = \frac{mmf}{l}$$
$$= \frac{\phi S}{l}$$
$$= BAL$$
$$Jur$$

$$f = \frac{B}{u}$$

mmf/unit length = B

 $mmf = \frac{BL}{\mu} \implies AT = \frac{BL}{\mu}$   $mmf = \frac{BL}{\mu} \implies BL = \frac{BL}{4\pi x \overline{10} \mu_r}$   $(AT) \qquad Ho \mu_r \qquad A\pi x \overline{10} \mu_r$ 

By using the above Concept,

The mont Required for algap of Smooth growth reactive  $AT_{g} = \frac{Bav lg}{Fagar}$ For alg  $H_{r} = 1$ 

45xio xI

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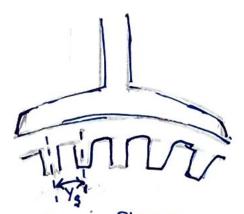
ATg = 8,00,000 Bauly > Fog Smooth Armature

But for Slotted Armature, the most Required for air gap is greater than the most Required for 8 mooth armature. Therefore most Required for air gap of Slotted Armature is greater that most Required for airgap of Smooth Armature.

MMF Required for airgap of Slotted Armature is = kg x mmf Required for airgap of Smooth Armature

ATg = 8,00,000 kg Bauly > For Blotted

Due to Slotting in the Armature, the aisgap length aig is increased but gap area is decreased. Therefore the effective air gap length is given by kglg



From the above figure Als gap area/pole =  $Slots/pole \times Slot pitch \times Core length$   $Ag = \frac{S}{P} \times Y_S \times L$   $Ag = \frac{S}{P} \times Y_S \times L$  or  $Ag = \overline{TDL}$   $\frac{Ag}{P} = \frac{S}{P} \times Y_S \times L$  or  $Ag = \overline{TDL}$   $\frac{TD}{P} = \frac{TDL}{P}$ To case of Salient pale machies, the flux density is not uniform over the sis gap. At the pole centre, the flux density is Maximum, hence Daw is replaced by Bg.

Mmf Required for airgap of Salient pole Machine

Calculate the mmf Required for the airgap of a machine having Core length of 0.32 m including 4 ducts of 10 mm each. pole arc = 0.19 m, 8lot pitch = 65.4 mm, Slot opening = 5mm, airgaplength = 5mm, flux per pole = 52 m cob. Given Carter's Coefficient is 0.18 for opening/gap = 1 and is 0.28 opening/gap is 2.

## Aliver data

L= 0.32 m, nd = 4, Od = 10 mm, pole are = 0.19 m. Ys = 6504 mm, edg = 5 mm, aisgap length = 5 mm, \$\overline = 52 m wb, for opening | gap = 1, then Carter Coefficient 18 0.18. For opening | gap = 2, then Carter Coefficient = 0.28. To find

MMF Raquired for air gap of DC machine Solution

$$AT_{g} = 8,00,000 Bg kg kg.$$

$$AT_{g} = 8,00,000 Bg \times 5 \times 10^{-3} \times kg \rightarrow (1)$$

we know that

$$K_{f} = \frac{Poloanc}{Polepitch} = \frac{Bau}{Bg}$$

$$\frac{0.19}{TDL} = \frac{Pa}{TDL}$$

$$\frac{\pi D}{P} = \frac{Bau}{Bg}$$

$$0.19 \text{ Bg} = \frac{P \phi}{T DL} \times \frac{T D}{P}$$

0.19  $Bg = \frac{4}{L}$   $Bg = \frac{4}{0.19 \times L}$  $Bg = \frac{52 \times 10^{3}}{0.19 \times 0.32}$ 

$$Bg = 0.8553 \text{ cob}/m^2$$

be know that

$$kq = \frac{L}{L'} * \frac{\gamma_s}{\gamma_s'} \longrightarrow (2)$$

we know that

$$L' = L - k_{cd} nd \omega_{d}$$

$$L' = L - k_{cd} \times 4 \times 10 \times 10^{3} \longrightarrow \textcircled{3}$$

$$k_{cd} \text{ depends on the sation duct opening ais gap longth}$$

$$\underbrace{\text{Duct openining}}_{ais gap longth} = \underbrace{\frac{10 \times 10}{5 \times 10^{3}}}_{5 \times 10^{3}} = 2$$

$$\underbrace{\text{Criven Contex Coefficient is e, for opening/aisgap longth}}_{\circ k_{cd} = 0.28$$

$$F_{\text{Rom oquation }}^{\text{FRom oquation }}_{L' = 0.32 - (0.28 \times 4 \times 10 \times 10^{-3})}_{L' = 0.3088 \text{ m}}$$

we know that

ſ

 $Y'_{s} = (65 \cdot 4 \times 10^{-3}) - (k_{cs} \times 5 \times 10^{-3}) \rightarrow (4)$ 

Slot opening = 
$$\frac{5 \times 10^3}{5 \times 10^3} = 1$$

Given Carter Coefficient is 0.18, fog opening = 1 aizgaplongth

From Eq. (4)  $Y_{s}' = (6504 \times 10^{3}) - (10018 \times 5 \times 10^{3})$  $Y_{s}' = 0.0645$ , m

$$k_{g} = \left(\frac{0.32}{0.3088}\right) \times \left(\frac{(65.4 \times 10^{-3})}{0.0645}\right)$$

# Kg = 1.0507

Substitute kay and by Value in aquation () ATg = 8,00,000 × 0.8553 × 5x10 × 21.0507 ATg = 3594.7541, AT

Ang = 3594.7541, AT.

The Statog of a machine has a smooth Surface but its rotor has open type flots with flot width = tooth width = 12 mm and the length of all gap is 2mm Find the effective length of airgap, if the carter's coefficient is equal to 1 . There are no 1+ 5lg/ws radial ducts.

Given data

 $W_{g} = W_{t} = 12 \times 10^{3} \text{ m}$ ,  $L_{g} = 2 \times 10^{3} \text{ m}$ ,  $k_{eg} = \frac{1}{1 + 5 L_{g}} | w_{s}$ To find

Effective length of air gap (ly) Solution

The effective length of aig gap is given by lg = kglg  $lg' = kg \times (2 \times 10^3) \longrightarrow (1)$ where kg = kgs x kgd -> 2 Given there is no radial Ducts, i. kgd = 1  $k_{gs} = \frac{\gamma_s}{\gamma_s}$  $k_{gs} = \frac{Y_s}{Y_{s-k_{es}} \omega_{s}}$ > 3

$$Y_{g} = \omega_{L} + \omega_{g}$$

$$= (12 \times 10^{3}) + (12 \times 10^{3})$$

$$Y_{g} = 24 \times 10^{3}, m$$
Criven 'k<sub>cs</sub> =  $\frac{1}{1 + 54g/\omega_{g}}$ 

$$k_{cs} = \frac{1}{1 + (\frac{5 \times 2 \times 10^{3}}{12 \times 10^{3}})}$$

$$k_{cs} = \frac{1}{1 + (\frac{10}{12})} \implies k_{cs} = 0.5455$$
From Equation (3)
$$k_{qs} = \frac{(24 \times 10^{3})}{((24 \times 10^{3}) - (0.5455 \times 12 \times 10^{3}))}$$

$$k_{qs} = \frac{1.375}{1.375}$$
From Equation (2)
$$k_{q} = 1.375 \times 1 \implies k_{q} = 1.375$$
From Equation (1)
$$dq^{1} = 1.375 \times (2 \times 10^{3}) = 0.0028 m$$

$$dq^{1} = 2.8 mm$$
Answer

The effective air gap length (lg) = 2.8 mm

Determine the air gap length of a DC machine from the following particulars. gross length of core = 0.12m, no: of ducts = 1 and is 10 mm coide, Slot pitch = 25 mm, Slot width = 10 mm, Carter's Coefficient for flots and ducts = 0.82. The gap density at Pole centre = 0.7 cob/m<sup>2</sup>. field mmf / pole = 3900 AT mmf Required for iron parts of magnetic circuit. Fiven data

 $L = 0.12 \text{ m}, \ \Pi d = 1, \ \omega d = 10 \times 10^{3} \text{ m}, \ \forall s = 25 \times 10^{3} \text{ m}, \ \omega s = 10 \times 10^{3} \text{ m}, \ k_{cs} = k_{ed} = 0.32, \ Bg = 0.7 \ \omega b \text{ m}^{2}, \ \text{AT}_{f} = 3900 \text{ AT}, \ \text{Mmf} \ \text{Required for Mon part} = 800 \text{ AT} \ To find$ 

Salution

Mmf Required for airgap of DC Machine is given by (Salient pale M/c)

$$AT_{g} = \frac{R_{i}}{R_{i}} = \frac{AT_{g}}{R_{i}} = \frac{AT_{g}}{R_{i}} = \frac{AT_{g}}{R_{i}} = \frac{AT_{g}}{R_{i}} = \frac{AT_{g}}{R_{i}} \xrightarrow{\text{AT}_{g}} \xrightarrow{\text{AT}_{g}}$$

fretal mmf/pale = 3900 > mong Required for air, gap. A Mrs pequired for thon parts = 800 AT

From the Figure,

Field mmf | pole = mmf Required for als gap + mmf Required for iten parts

3900 = mmf Required for all gap + 800 Mmf Required for Air gap (ATy)= 3900 - 800

ATg = 3100 AT

We know that  $kg = kgs \cdot kgd \rightarrow (2)$   $kgs = \frac{Y_s}{Y_s'} = \frac{Y_s}{Y_s - k_{cs} \omega_s}$   $kgs = \frac{25 \times 10^3}{(25 \times 10^3) - (D \cdot 32 \times 10 \times 10^3)}$  $kgs = 1 \cdot 1468$ 

From Equation 2

From Equation ()

$$lg = 0.0047 m = 4.7 mm$$

A 15 kw, 230V, 4 pole de machine has the following data

Armature diameter = 0.25 m, armature Core length = 0.125 m, length of airgap at pole centre = 2.5 mm, flux per pale = 11.7 x 10<sup>3</sup> wb, ratio of <u>Pole arc</u> = 0.66 Pale pitch calculate the mmf Required for the airgap for i) if the armature surface is treated as smooth. ii) if the armature surface is treated as smooth. ii) if the armature is slotted and the gap Contraction factor is 1.18

Given data Po=15 kw, V=230V, P=4, D=0.25m, L=0.125m, dg=2.5mm=2.5x10<sup>3</sup>m, φ=11.7x10<sup>3</sup>,  $k_f = \frac{Pole are}{Pole pitch} = \frac{Bao}{Bg} = 0.66$ ,  $k_g = 1.18$ To find i) mong Required for smooth Armature. ii) mong Required for Slotted Armature. Solution i) MMF Required for airgap of Smooth Armature ATg= 8,00,000 Bglg. ATg= 8,00,000 x Bg x 2.5 x 10 -> 1

Given

$$\frac{B_{av}}{B_{g}} = 0.66$$

$$\frac{P.\phi}{\pi DL} = 0.66$$

$$B_{g}$$

Weknow that Baw = <u>P\$</u> TDL

$$\left(\frac{4 \times 11.7 \times 10^{3}}{5 \times 0.25 \times 0.125}\right) = B_{g}$$

$$\frac{0.4767}{0.66} = Bg$$

$$Bg = 0.7223 \quad \text{wb} m^2$$

From Equation (1)

 $AT_{g} = 8,00,000 \times 0.7223 \times 2.5 \times 10^{3}$  $AT_{g} = 1444.5481$  BT

ii) MMF Required for Aisgap of Slotted Armature  

$$PTg = 8,00,000$$
 Bg kg lg  
 $PTg = 8,00,000 \times 0.7233 \times 1.18 \times 2.5 \times 10^{-3}$   
 $PTg = 1704.628$ , AT

### Answeas

i) mmf Required for aisgap of Smooth Armature ATg = <u>1444.5481</u>, AT ii) mmf Required for aisgap of Slotted Armature ATg = <u>1704.628</u>, AT calculate the mmg Required for airgap of a DC machine with an axial length of 20 cm (No ducts) and a pole are of 18 cm, the 8lot pitch is 27 mm, slot opening = 12 mm, airgap = 6mm and the useful flux/pale = 27 m cob. Take carter's coefficient for 8lot as 0.3.

Given data

 $L = 20 \text{ cm} = 20 \times 10^{2} \text{ m}, \text{ pale arc} = 18 \text{ cm} = 18 \times 10^{3} \text{ m}$   $Y_{s} = 27 \times 10^{3} \text{ m}, \ \omega_{s} = 12 \times 10^{3} \text{ m}, \ \text{Lg} = 6 \times 10^{3} \text{ mm}, \ \text{q} = 27 \times 10^{3} \text{ m}$ Kes = 0.3

To find

. Mmf Required of aisgap of DC machine Salution

> MMF Required for sisgap of DC machine (Salient pole)

We know that

$$kg = kgs \times kgd$$

$$kg = \frac{Y_s}{Y_s'} \times \frac{L}{L'}$$

$$k_{g} = \frac{V_{s}}{(Y_{s} - k_{cs}\omega_{s})} \times \frac{L}{(L - K_{cd} nd \omega d)}$$
  
To this problem, ducts are not used, Hence  $n_{d} = 1$   

$$k_{g} = \frac{(27 \times 10^{3})}{((27 \times 10^{3}) - (0.3 \times 12 \times 10^{3}))} \times \frac{20 \times 10^{2}}{(20 \times 10^{2} - 0)}$$
  

$$k_{g} = 1.1538 \times 1$$

$$\frac{18 \times 10^2}{\pi D} = \frac{P \phi}{\pi D L}$$

$$\frac{\pi D}{P} \qquad Bg$$

$$B_{g} \times 18 \times 10^{2} = \frac{Pq}{\pi DL} \times \frac{\pi D}{P}$$

$$Bg \times 18 \times 10^{-2} = \frac{\Phi}{L}$$

$$B_{q} = \frac{\phi}{18 \times 10^{2} \times L}$$

$$B_{q} = \frac{27 \times 10^{3}}{18 \times 10^{2} \times 20 \times 10^{2}} \Rightarrow \frac{27 \times 10^{3}}{18 \times 20 \times 10^{4}}$$

$$Bg = 0.75 \text{ wb}/m^2$$

From Equation (1)

$$AT_{g} = 8,00;000 \times 1.1538 \times 0.75 \times 6 \times 10^{3}$$
  
$$AT_{g} = 4153.6800, AT$$

Answers

Mmf Requiered for Airgap for DC machine ATg = 4153,6800, AT Estimate the effective gap area / pole of a 12 pole Slipring Induction motor with following data.

Staton bore = 0.7 m, Core length = 0.3 m, No of Staton Blots = 90, Staton Blot opening = 2 mm, roban Slots = 120, roban Blot opening = 2 mm, Carten's Coefficient for ducts = 0.68, Carter's Coefficient for Blots = 0.46; No =: of Ventilating ducts = 2 each on Staton and roban, width of each ventilating duct = 10 mm, airgap length = 0.95 mm.

Given data

P=12,  $D=0.7m, L=0.3m, S_{s}=90$ ,  $\omega_{ss}=3x10m$   $S_{7}=120, \omega_{sy}=3x10^{3}m, k_{cd}=0.68, k_{cs}=0.46$ ,  $n_{d}=3$ , on Staton and solog,  $\omega_{d}=10x10^{3}m$ ,  $l_{g}=0.95x10^{3}mm$ .

To find

kg

$$A_{g} \Rightarrow Gap area / pole$$

$$A_{g} = \frac{\pi DL}{P}$$

$$= \frac{\pi \times 0.7 \times 0.3}{12}$$

$$A_{g} = 0.05 \text{ fo}, m^{2}$$

we know that

$$kg = kgs \times kgd \longrightarrow (2)$$

For Induction motor, Both Stator and rotas have

Poton Blots.

$$k_{gss} = \frac{y_{ss}}{y_{ss}}$$

$$k_{gss} = \frac{\gamma_{ss}}{\gamma_{ss} - k_{cs} \omega_{ss}} \rightarrow \textcircled{4}$$

We kno that  $Y_{ss} = W_{ts} + W_{ss} = \frac{TD}{s_s}$  From the given data

$$S_{S} = \frac{TD}{S_{S}}$$

$$Y_{ss} = \frac{\pi \times 0.7}{90}$$

$$Y_{88} = 0.0244$$
, m.

From Equation (4)

$$k_{gss} = 0.0244$$
  
(0.0244 - (0.46 ×  $3 \times 10^{-3}$ ))

$$k_{gg_{\pi}} = \frac{Y_{g_{\pi}}}{Y_{g_{\pi}}}$$

$$k_{gsr} = \frac{Y_{sy}}{(Y_{sy} - k_{cs} \omega_{y})}$$

$$Y_{Sy} = W_{Sy} + W_{Sy} = \frac{T D_y}{S_y}$$

$$D_{r} = D - 2 lg$$

$$D_{r} = 0.7 - (2 \times 0.95 \times 10^{-3})$$

$$D_{r} = 0.6981 m$$

Yen = 0.0183

kgsg = 1.0816

From Equation 3 Kgs = 1.0599×1.0816 Kgs = <u>1.1463</u>

$$kgd = \underline{L}$$

$$= \underline{L}$$

$$L - kcd D d \omega d$$

$$k_{gd} = \frac{0.3}{0.3 - (0.68 \times 3 \times 10 \times 10^{-3})}$$

From Equation 3

From Equation ()

$$Ag = 0.0447, m^2$$

Answers

A 175 MVA, 20 pole water wheel generator has a Corre length of 1.72 m and a diameter of 6.5 m. The Stator Slots (open) have a width of 22 mm. the Slot pitch being 64 mm and the air gap length at the centre of pole is 30 mm. There are 41 radial Ventilating ducts each 6mm wide. The total mmf pole is 27000 A, The mmf required for the air gap is 87% of the total mmf per pole. Estimate the average flux density in the air gap, if the field form factor is 0.7.

Given data

 $Q_{1} = 175 \text{ MYA}, P = 20, L = 1.72 \text{ m}, P = 6.5 \text{ m}, W_{S} = 22 \text{ mm}$ =  $22 \times 10^{3}$ ,  $Y_{S} = 64 \text{ mm} = 64 \times 10^{3} \text{ m}$ ,  $lg = 30 \text{ mm} = 30 \times 10^{3} \text{ m}$  $nd = 41, Wd = 6 \times 10^{3} \text{ m}$ , Mmf | pole = 27000 A,  $AT_{g} = 87^{\circ}/_{0} \text{ of total mmf} | pole, K_{f} = 0.7$ To find

Solution

$$AT_q = 8,00,000 \text{ kg Bglg} (rog Solient pole m/c)$$

$$B_q = \frac{AT_q}{8,00,000 \times \text{ kg x lg}}$$

$$B_q = \frac{AT_q}{8,00,000 \times \text{ kg x 30 x 10}^3} \rightarrow (1)$$

Given 
$$AT_g = 87\%$$
 of total mmf/pole  
 $AT_g = 0.87 \times 27,000$   
 $AT_g = \frac{23490}{7}, AT$ 

we know that

$$k_g = \frac{L}{L'} \times \frac{\gamma_s}{\gamma_s'}$$

$$k_{g} = \frac{1072}{L'} \times \frac{64 \times 10^{3}}{y_{g}'} \rightarrow \textcircled{2}$$

$$L' = L - k_{cd} n_{d} \omega_{d}$$

$$L' = 1072 - (k_{cd} \times 41 \times 6 \times 10^{3}) \longrightarrow (3)$$

$$k_{cd} = \frac{2}{\pi} \left[ \tan^{2} y - \frac{1}{y} \log \sqrt{(1+y^{2})} \right]$$

For ducts, 
$$y = \frac{\omega_d}{alg}$$
  
 $y = \frac{6 \times 10^3}{2 \times 30 \times 10^3} = 0.1$ 

$$k_{cd} = \frac{2}{\pi} \left[ \frac{\tan^{-1} \cos 1}{4 \pi} - \frac{1}{\cos 1} \log \sqrt{(1+0.1^{2})} \right]$$
  
=  $\frac{2}{\pi} \left[ \cos \cos 997 - (\cos \cos \cos 22) \right]$   
 $k_{cd} = 0.0497$ 

From Equation (3)  
L' = 1.72 - (0.0497 x 41 x b x 10<sup>3</sup>)  
L' = 1.7078, m.  

$$y'_{s} = y_{g} - K_{cs} \omega_{g}$$
  
 $y'_{s} = (64 x 103) - (k_{cs} \times 22 x 103) \rightarrow (4)$   
 $k_{cs} = \frac{2}{\pi} \left[ Lao^{-1}y - \frac{1}{y} \log \sqrt{(1+y^{2})} \right]$   
FOR Blots,  $y = \frac{\omega_{g}}{2\lambda_{g}}$   
 $y' = \frac{22 \times 10^{3}}{2 \times 30 \times 10^{3}} = 0.3667$   
 $k_{cg} = \frac{2}{\pi} \left[ Lao^{-1} 0.3667 - \frac{1}{0.3667} \log \sqrt{(1+0.3667^{2})} \right]$   
 $k_{cg} = \frac{2}{\pi} \left[ 0.3515 - (2.7270 \times 0.0274) \right]$   
 $k_{cg} = \frac{0.1762}{4}$   
From Equation (4)

 $Y_{s}' = (64x10^{3}) - (0.1762 \times 22x10^{3})$ 

$$Y_{s}' = 0.0601$$

From Equation (2)

$$k_{g} = \frac{1072}{107078} \times \frac{64 \times 10^{-3}}{0.0601}$$

$$kq = 1.0071 \times 1.0649$$
  
 $kq = 1.0725$ 

From Equation (1)  $B_{g} = \frac{23490}{8,00,000 \times 1.0725 \times 30\times 10^{3}}$   $B_{g} = \underbrace{0.9126}_{1.00}, \underbrace{00}_{m^{2}}$ From we know that form factor  $k_{f} = \frac{B_{av}}{B_{g}}$ 

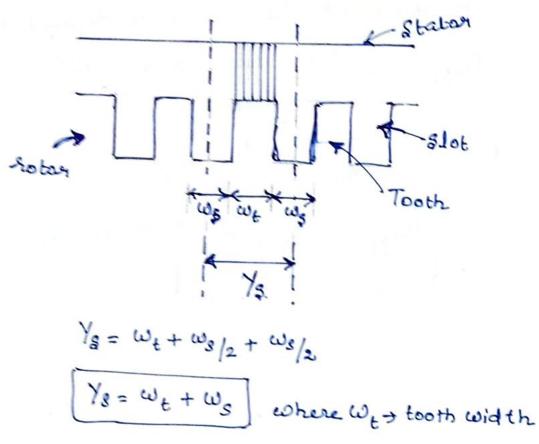
... 
$$Bay = Bg \times k_f$$
  
 $Bay = 0.9126 \times 0.7$   
 $Bay = 0.6388, \omega b/m^2$ 

Ang

$$B_{av} = 0.6388$$
,  $\omega b | m^2$ 

### Real and apparent flux densities

Consider a rotating machine, the status flux passes through the airgap and enter to the rotor.



Ws > Slot width (ba) Tooth Area (AL) = Liwe where L-length of the core. Ys -> Slot pitch. Slotopening

From this figure, we observed that the major part of the flux from Statas passes through the tooth.

At lower flux densities, the flux passing through dlot can be neglected. But at higher flux densities, flux from statos Choose an alternate path through Slot due to Magnetic Saturation at booth Hence the real flux passing through the beath

is always less than the total (Apparent) flux

As a result, the real flux density in the tooth is always less than the apparent flux density

$$Bapp = \frac{\Phi_t + \Phi_A}{A_t} \rightarrow (1)$$

where  $\varphi_t \rightarrow flux$  over the tooth  $\varphi_A \rightarrow Flux$  over the flot gap.  $A_t \rightarrow Brea of tooth = L w_t$ 

The real flux density is defined as

$$Breal = \frac{\Psi_E}{A_E} \rightarrow (2)$$

Relation b/w Bapp and Bread  
From Equation ()  

$$Bapp = \frac{\Phi_t + \Phi_a}{\Phi_t}$$

$$Bapp = \frac{\Phi_t + \Phi_a}{\Phi_t}$$

$$Bapp = \frac{\Phi_t}{\Phi_t} + \frac{\Phi_a}{\Phi_t}$$

From Equation (2) Bapp = Breat + Pa At Bapp = Breat +  $\frac{\Phi_A}{A_1} \times \frac{A_a}{A_a}$ Bapp = Breat + da \* Aa Aa A. Bapp = Breat + (Bair x k) where  $k = \frac{A_{\alpha}}{A_{L}}$ At > tooth Area = Liwt Li > net 1800 length = Sf L > (without duct) = Sf (L-nd wd) -> with Duct Aa > Ais gap quea \_ Staton rotan From the figure A Aa = Area of abcdefgha = Area over the Slot pitch - Tooth Area = Area of abgha - Tooth Area Aa = LYs - Liwe

we know that B=UH= Mat = Mourat FOR als, B= Moatreal = B= Ar=1 Ba= 4x10 atreal Bapp = Breal + (4x10 atreal x K)

X

atreal -> real Magnetising force.

Calculate the apparent flunc density at a particular Section of a tooth from the following data

Pooth width = 12 mm, Slot width = 10 mm, Gross Core length = 0.32 m, No of Ventilating ducts = 4 and each 10 mm wide. real flux density = 2.2 cob/m<sup>2</sup>. Permeability of teeth Corresponding to real flux density = 31.4 × 10<sup>-6</sup> H/m. Stacking factors = 0.9

#### given data

 $w_{t} = 12 \text{ mm} = 12 \times 10^{-3} \text{ m}, \quad w_{s} = 10 \text{ mm} = 10 \times 10^{-3} \text{ m},$   $L = 0.32 \text{ m}, \quad nd = 4, \quad wd = 10 \text{ mm} = 10 \times 10^{-3} \text{ m}, \quad \text{Breal} = 2.2$  $w_{b}/m^{2}, \quad u_{t} = 31.4 \times 10^{-6} \text{ H/m}, \quad s_{f} = 0.9$ 

To find

Salution

$$Bapp = 2 \cdot 2 + (4\pi x 10^7 \times at_{real} \times k) \rightarrow 1$$

we know that  $B = \mathcal{U} H$ (or)  $B = \mathcal{U} at$   $\therefore Breal = \mathcal{U} at real$  $at_{real} = \frac{Breal}{\mathcal{U}}$   $at_{real} = \frac{2 \cdot 2}{31 \cdot 4 \times 10^6}$ 

at real = 
$$\frac{10063.6943}{AT/m}$$
  
 $A = \frac{Aa}{A_{t}} \rightarrow \textcircled{2}$   
 $A_{a} \rightarrow gap area over the Slot pitch$   
 $A_{t} \rightarrow tooth Area$   
 $A_{t} \rightarrow J$ 

Aa (gap area over the Slot pitch) = Total area over the Slot pitch - Tooth area

(Area of abcdefgha) = (Area of abgha) - (Area of Thereforme C defc)

$$A_a = LY_s - Li \omega_E$$

$$M_a = (0.32 \times Y_g) - (Li \times 12 \times 10^3)$$

$$\frac{Y_s}{V_s}$$

$$\frac{Y_s}{V_w}$$

$$\frac{Y_s}{W_E}$$

From Figure  $Y_s = \omega_t + \omega_s/2 + \omega_s/2$ 

$$Y_{g} = \omega_{t} + \omega_{g}$$

$$Y_{g} = (18 \times 10^{-3}) + (10 \times 10^{-3})$$

$$Y_{g} = (22 \times 10^{-3}) m$$

$$L^{2} = S_{g} (L - n_{d} \omega_{d})$$

$$= 0.9 (0.82 - 4 \times 10 \times 10^{-3})$$

$$L^{2} = 0.252, m$$

$$R_{q} = (0.38 \times 22 \times 10^{-3}) - (0.252 \times 12 \times 10^{-3})$$

$$R_{q} = (0.040, m^{2})$$

$$R_{t} = 0.0040, m^{2}$$

$$R_{t} = 0.0030, m^{2}$$
From Equation (2)  

$$k = \frac{0.0040}{0.0030}, m^{2}$$
From Equation (2)  

$$k = \frac{0.0040}{0.0030}$$

$$K = 1.8228$$
From Equation (3)  

$$R_{q} = 2.2 + (4\pi \times 10^{-1} \times 70063.6943 \times 1.3228)$$

$$Bapp = 2.2 + (4\pi \times 10^{-1} \times 70063.6943 \times 1.3228)$$

$$Bapp = 2.3165, \omega b | m^{2}$$

calculate the apparent flux density at a section of teeth of an armature of DC machine from the following data

flot pitch = 24 mm, flot width = tooth width=12mm dength of armature core including 5 ducts of 10mm each is 0.38 m. Iron Stacking factor = 0.92, tooth flux density in teeth at that section is 2.2 wb/m<sup>2</sup> for which the mmf is 70,000 AT/m.

given data

 $Y_{s} = 84 \text{ mm} = 24 \text{ m}, \quad \omega_{t} = \omega_{s} = 12 \text{ mm} = 12 \times 10^{3} \text{ m}$   $L = 0.38 \text{ m}, \quad Dd = 5, \quad \omega_{d} = 10 \text{ mm} = 10 \times 10^{3} \text{ m}, \quad S_{f} = 0.92$ Breal = 2.2  $(0b/m^{2}), \quad atreal = 70,000 \text{ BT/m}$ To find

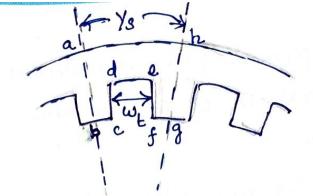
Apparent flux density (Bapp)

Solution

Bapp = 2.2 + (45x10 x 70,000xk) -> ()

$$k = \frac{A_a}{A_b} \rightarrow \textcircled{2}$$

 $A_a \rightarrow gap area in a dlot pitch$  $<math>A_t \rightarrow both area.$ 



Aa (gap grea in the Slotpitch) = Total Area in the Slot pitch - tooth Area

Area of abcdefgha = Area of abgha - Area of defed

$$\begin{aligned} & A_{a} = LY_{s} - Li\omega_{t} \\ & B_{a} = (0.88 \times 24 \times 10^{3}) - (Li \times 12 \times 10^{3}) \rightarrow (3) \\ & Li = S_{f}(L - n_{d}\omega_{d}) \\ & = 0.92(0.88 - 5 \times 10 \times 10^{3}) \\ & Li = 0.3036, m. \end{aligned}$$

From 3

$$A_{a} = (0.38 \times 24 \times 10^{3}) - (0.3036 \times 12 \times 10^{3})$$

$$A_{a} = 0.00548 , m^{2}$$

$$A_{t} = Li \times \omega_{t}$$

= 0.3036x 12x103

$$h_t = 0.00364$$
, m<sup>2</sup>

From 2

$$k = \frac{Aa}{At} = \frac{0.00548}{0.00364}$$

K = 1.5042

From Equation ()

 $Bapp = 2.2 + (4\pi \times 10^{-7} \times 70,000 \times 1.5042)$  $Bapp = 2.3823 \quad \omega b/m^2$ 

Answers

Apparent flux density (Bapp) = 2.3323 cob/m<sup>2</sup>

Find the permeability at the root of the beeth of a Dc machine armature from the following data Slot pitch = 2.1 cm, tooth width at the root=1.07 cm Gross length = 32 cm, Stacking factor = 0.9, read flux density at the root of the tooth = 2.25 tesla, apparent flux density at the isoot = 2.36 tesla Given data

 $Y_{s} = 2 \cdot 1 \text{ cm} = 2 \cdot 1 \times 10^{2} \text{ m}, \quad \omega_{t} = 1 \cdot 07 \text{ cm} = 1 \cdot 07 \times 10^{2} \text{ m}$  $L = 32 \text{ cm} = 32 \times 10^{2} \text{ m}, \quad s_{f} = 0 \cdot 9, \quad \text{Breal} = 2 \cdot 25, \quad \text{Bapp} = 2 \cdot 36 \quad \text{tesla}$ 

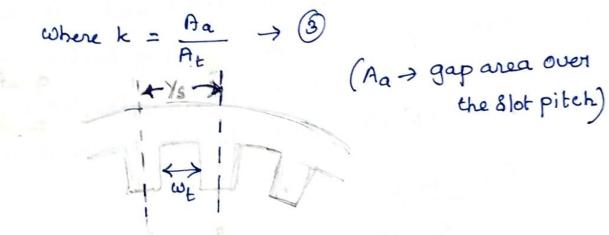
To find

Permerbility of the root of the teeth (u)

Permant O'L Com h	We know that
Permeability (11)= Bread	B=~H
atreal	07
	B=Mat
$\mu = \frac{2 \cdot 25}{at_{real}} \rightarrow 0$	: u= at

we know that

atreal = 
$$\frac{Bapp - Breal}{4\pi \times 10^7 \times K} = \frac{2 \cdot 36 - 2 \cdot 25}{4\pi \times 10^7 \times K} \rightarrow 2$$



From the Figure  $A_a = Total area over the Slot pitch - teeth Area$   $A_a = LY_s - Li \omega_t$   $A_a = (3axio^2 x 2 \cdot 1x io^2) - (Li x 1 \cdot 07x io^2)$   $Li = S_f(L - h_d \omega_d)$   $Li = 0.9 (32xio^3 - 0)$  Li = 0.288, mIf there is no duct then  $n_d = 0$ 

From Equation (

$$\begin{aligned} & \Pi_{a} = (32 \times 10^{2} \times 2.1 \times 10^{2}) - (0.288 \times 1.07 \times 10^{2}) \\ &= (32 \times 2.1 \times 10^{4}) - (0.288 \times 1.07 \times 10^{2}) \\ & \Pi_{a} = 0.0036, m^{2} \\ & \Pi_{t} = Area \text{ of the tooth} = Li \omega_{t} \\ & \Pi_{t} = 0.288 \times 1.07 \times 10^{2} \\ & \Pi_{t} = 0.0031, m^{2} \end{aligned}$$

From Equation (3)  

$$k = \frac{0.0036}{0.0031}$$

k = 1.1682

From Equation 2

$$at_{real} = \frac{2.36 - 2.25}{4\pi x 10^7 \times 1.1682}$$

at real = 74931.7058, AT/m

From Equation (B)  $\mathcal{U} = \frac{2 \cdot 25}{74931 \cdot 7058}$ 

 $M = 0.00003 = 3 \times 10^5$ , H/m

#### Flux leakage (Magnetic leakage)

The flux which passes through unwanted path is Called the leakage flux. It is impossible to confine all the magnetic flux is a given path.

The leakage flux does not contribute to either transfer or conversion of energy. The leakage flux affects the following performance

i) Excitation demand of salient pale machines

i) Performance of ac machines depends on the leakage reactance.

iii) Voltage regulation of generators and transformers

iv) Commutation Conditions in DC machines.

v) Stray Load losses

21110

Called Loog

vi) Circulating Currents in transformer tank walle.

For magnetic circuit calculations, deakage Coefficient is introduced in order to take into account the leakage flux. The leakage Coefficient is defined as the ratio of total flux to useful flux

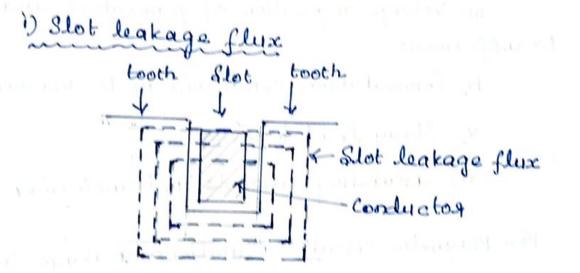
Leakage Coefficient C1 = Total flux Useful flux

> CI = Useful flux + Leakage flux Useful flux.

#### Types of Leakage flux

The different types of armature leakage fluxes

- i) Slot leakage flux
- ii) Tooth tap leakage flux
- III) Zigzag leakage flux
- iv) Overhang leakage flux
- V) Differential leakage flux
- vi) Skew leakage flux.
- vii) Peripheral leakage flux



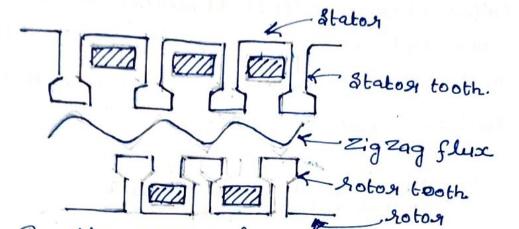
The flux that Crosses the flot from one tooth to the next and returning through Ison are called Slot leakage flux.

ii) Tooth top leakage flux

The flux flowing from top of one tooth to the top of another tooth is called tooth top

leakage flux. This lookage flux is considered only the machine having large airgap length.

iii) Ziz Zag Leakage flux



The flux passing from one tooth to another in a zigzag fashion is Called zigzag leakage flux.

The magnitude of this flux depends on the length of aisgap and the relative positions of the tips of stator and rotog booth.

### iv) Overhang leakage flux

The conductors which connects the two coil Sides of a coil are called overhang. The fluxes produced by the overhang portion of the armature winding are called overhang leakage flux.

# V) Differential leakage flux

This leakage flux is due to dissimilar mmf distribution in the Stator and rotar.

## Vi) Skew leakage flux

A twist provided on the rotor of Induction motor to eliminate harmonic torques and noise is Called Skewing. This Skewing reduces the mutual flux and thus Creating a difference b/w total flux and mutual flux. This difference is called Skew leakage flux.

Vii) Pheripheral leakage flux

The fluxes flowing Circumferentially around the air gap linking with any of the Winding is called peripheral leakage flux.

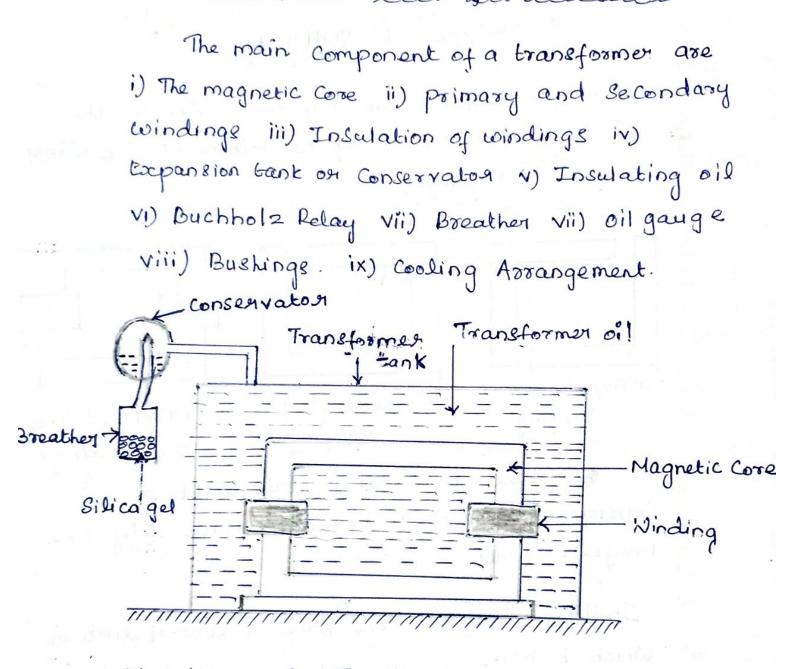
#### **EE 8002 DESIGN OF ELECTRICAL APPARATUS**

#### <u>UNIT II</u>

#### **DESIGN OF TRANSFORMER**

Prepared by Dr . T. Dharma Raj Asso.Prof /EEE

#### Constructional Details of a Transformer



Magnetic core (or) Transformer core

grader .

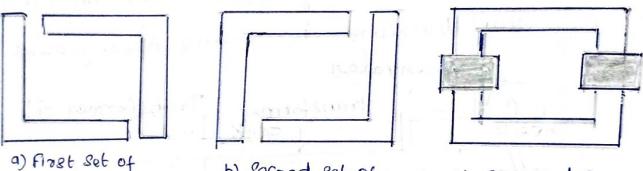
The Magnetic core is made by a good magnetic material like Cast iron (or) Silicon Steel . The magnetic Core are generally laminated with thickness of 0.35mm to 0.5mm. The laminations are insulated from each other by Coating with a thin coat of Varnish.

Dury Contraction

The two types of Magnetic Core are

a) core type b) Shell type

Corretype Magnetic Corre has two limbs for the windings and is made up of joining two L ----Shaped Stampings.

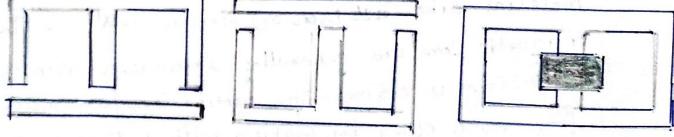


Stampings Stampings

c) Finished Core with windings.

Here the windings surround a Considerable part of core and have only one Magnetic path.

Shell type magnetic core have a Central limb in Which both the windings are wounded. and is made up of joining E and I Shaped Stampings



Here the Core Surrounds the Considerable Part of Gove Windings. and have two magnetic paths. In both types of Cores, the joints are staggered due to following advantages

i) Avoid Continuous gap blus joints Causing increase in the magnetising Current.

ii) To increase the mechanical strength of the core

iii) Avoid undue humming noise.

# Insulation of windings

Paper is still used as the basic conductor insulation. Enamel Insulation is used as the interturn insulation for low Voltage transformers. For power transformers enamelled Copper with paper insulation is also used.

## Expansion tank of Conservator

It is an Small auxillary tank mounted above the transformer and connected to the main tank by a pipe. It function is to keep the transformer tank full of oil during expansion and contraction of the Coil when Subjected to Change in temperature.

## Insulating oil og Transformer oil

It protect the paper from dist and moisture

and removes the heat produced in the core and coils. The oil must posses the following properties.

i) High dielectric Strength

ii) Free from inorganic acid, alkali and corrosive Sulphur to prevent injury to the Conductor or Insulation

iii) Low viscosites to provide good heat

iv) Crood Resistance to emedsion

Buchbo Buchholz Pelay

It is an gas operated veloy placed the inside the pipe which connects the tank and Conservator.

This relay will give an alarm incase of minor fault and in case of severe fault this relay disconnect the transformer from the supply mains

#### Breather

The breather is connected one side of the conservator and it is filled with some drying agent such as calcium chloride (or) silica got to prevent the entry of the moisture inside the transformer tank. The drying agent is replaced periodically as routine mainteinance.

## Oil gauge

Every transformer have a oil gauge to indicate the oil present inside the tank. The oil gauge is provided with an alarm contact which gives an alarm, when the oil level has dropped below the permissible height due to oil leak on due to any

## Bushings

Connections from the transformer windings. are brought out by means of bushings. Bushings are fixed on the transformer tank.

# Cooling Arrangement

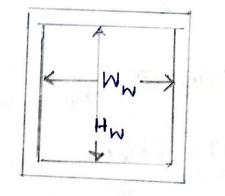
The various methods of cooling employed in a transformer are

- a) oil immersed natural cooled transformers
- b) oil immersed forced aig Cooled they's
- c) oil immersed water cooled transformers
- d) oil immersed forced oil cooled transformers
- e) Air blast transformers.

#### Output Equation of transformer

a) Single phase transformer Consider an ideal transformer, the output Power is given by Q = EpIpX10<sup>3</sup> KVA (08) EgIsX10<sup>3</sup> KVA  $\rightarrow 0$ Where Ep, Es are the erof induced in the primary and secondary windings. Ip, Is are the primary and secondary currents From Emf Equation Ep = 4.44 f  $\phi_m$  Tp, Volts  $\rightarrow 0$ From Equation (1)

 $Q_{1} = 4.44 \int \Phi_{m} T_{p} I_{p} \times 10^{3}$ , kv A $Q_{1} = 4.44 \int \Phi_{m} AT \times 10^{3}$ , kv B



From the Figure

window Space factor =  $\frac{\text{Copper area in the window}}{\text{Total area in the window}}$  $K_{W} = \frac{A_{c}}{A_{c}} \rightarrow ③$ 

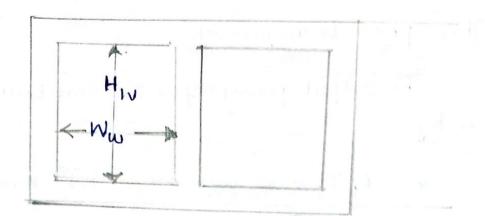
where Aw = Hw. Ww Copper Area in the window (Ac) = Copper Area in the primary winding + Copper area in the secondary wdg Ac = ( Primary turns x area of Conductors in primary wdg) + (secondary turns × area of conductors in secondary wdg) Ac= Tpap + Is as -> (A) For an ideal transformer, Current density is same for both sides, then  $\delta = \frac{I_{p}}{a_{p}} = \frac{I_{s}}{a_{s}}, \quad a_{p} = \frac{I_{p}}{\delta}, \quad a_{s} = \frac{I_{s}}{\delta}$ From (A) Ac = Tp Ip + Tg Is neglecting the magnetizing mmf, then Tp Ip = Tg Ig = AT  $A_{c} = \frac{AT}{s} + \frac{AT}{s} \Rightarrow \frac{2AT}{s} \rightarrow (5)$ Substitute (5) in (3) KN = 2AT 8 Aw AT= SkwAw From Equation () Qr = 4.44 f \$m & Kw Aw x103, KVA Q = 2.22 f \$m & kw Aw X103, KVA

b) Three phase transformer

The Output power of 34 Ideal transformer is given by  $Q = 3 E_p T_p \times 10^3$  (er)  $3 E_s I_s \times 10^3$  kvA  $\rightarrow$  (f) where  $E_p$ ,  $E_s$  are the emf induced (phase in the Primare and Recordary windings  $T_p$ ,  $I_s$  are the primary and secondary current/ phase. For star connection. Line Voltage (VL) =  $\sqrt{3}$  phase voltage (Vph) Line (urrent ( $T_L$ ) = phase current ( $T_ph$ ) For Delta Connection.

Line Voltage (VL) = Phase Voltage (Vph) Line Current (IL) = J3 phase Current (Iph) From emf Equation

hase.



From the figure

Window Space factor (Kw) = Copper area in the window Total area in the window

$$k_{\rm W} = \frac{A_c}{A_{\rm LO}} \rightarrow \textcircled{3}$$

Aw= Hw Ww

From the Figure, It is observed that each window daw have two limbs. Each Limb Corries a primary and secondary winding for a phase.

... Therefore Copper area in the window Ac = (Copper area of the primary and becondary winding In the first limb) + (Copper area of the primary and secondary winding in the second limb. Ac = (T

$$A_{c} = (T_{p} \cdot a_{p} + T_{g} a_{g}) + (T_{p} a_{p} + T_{g} a_{g})$$
$$A_{c} = 2 T_{p} a_{p} + 2 T_{g} a_{s} \rightarrow (4)$$

For an ideal transformer, Current density is same for primary and secondary, then  $\delta = \frac{I_p}{a_p} \neq \frac{I_s}{q_s}$  $a_p = \frac{T_p}{e}$ ,  $a_s = \frac{T_s}{s}$ From equation @  $A_{c} = \frac{2 \operatorname{Tp} \operatorname{Ip}}{e} + \frac{2 \operatorname{Ts} \operatorname{Is}}{8}$ neglecting magnetising mmf, then TpIp=TsIs=AT  $A_{c} = \frac{2AT}{g} + \frac{2AT}{g}$  $A_c = \frac{4AT}{s} \rightarrow 5$ Substitute (5) in (2)  $k_w = \frac{4AT}{8A_w}$ AT = KWSAW > 6 Substitute eq (6 in (2), we get Q= 3x4.44 f dm Kw8Aw x10, kvn

Q= 3.33 f dm Kw8 Awx10, KVA 2 at Capital Handwis Sty was und only it's

1, Show that output of the 3¢ core type transformer is given by Qr = 5.23 f Bm Hd<sup>2</sup> Hwx 10<sup>3</sup> kvA, where f is the frequency in Hz, Bm is the flux density in the core, H is the magnetic potential gradient in Limb A/m, d is the effective diameter of the Core?

The basic output equation of transformer is

$$Q_{r} = 3 E_{p} I_{p} x_{10}^{3} = 3 E_{s} I_{s} x_{10}^{3} k_{VA} \rightarrow (1)$$

30

where Ep= 4.44 f Tp \$, Voits.

$$B_{m=\frac{\Phi}{A_i}} \Rightarrow \phi = A_i B_m$$

From Equation (1)

$$Q = 3 \times 4.44 \text{ f} T_p \text{ Ai Bm } I_p \times 10^3 \text{ kva}$$
  
 $\rightarrow 2$ 

criver d'is the effective diameter of the Core

ie) 
$$A_i = \frac{\pi d^2}{4} \rightarrow 3$$

Substitute 3 in 2

 $Q = 13.32 \text{ f Bm } \frac{\pi d^2}{4} (T_p I_p) \times 10^3 \text{ kva}$   $Q = 13.32 \text{ f Bm } \frac{\pi d^2}{4} \text{ AT } \times 10^3 \text{ kva}$   $\longrightarrow$ 

A 30 core transformer has 3 limbs. Each limb. Consists of one primary and secondary winding.

Mmf acting in limb = mmf acting in the primary t mmf acting in the secondary

= Tp Ip + Te Is

Far an ideal transformer, TPIP=TeIs=AT

: mmf acting in the limb = AT + AT= 2AT

From the given daba H=AT/L ie) H= mmf acting in the limb length of the limb

 $H = \frac{2AT}{H_{W}}$ 

$$AT = \frac{H \times H_{\omega}}{2}$$

From (1)

Hence peroved.

Volt perturn interms of output Equation

he know that Ep= 4.44 f dm Tp  $E_t = 4.44 f \phi_m \rightarrow \mathbb{O}$ The output Equation a is given by Q= EpIpX103, KVA Q = 4.44 f om Tp Ip X 103, KVA Q= 4.44 f dm AT x 103, KVA het we take  $r = \frac{q_m}{r}$  $AT = \frac{\Phi m}{r}$ Q= 4:44 f Om 4m × 103, KVA Q= 4.44 f 4m x 10 % KYA  $\Phi_{m}^{2} = \frac{Q_{1}g}{444f \times 15^{3}}$ 

From Eq D

$$E_{L} = 4 \cdot 44 f \sqrt{\alpha}, \quad \sqrt{\frac{\gamma}{4 \cdot 44 f \times 10^{3}}}$$

$$E_{L} = \sqrt{\alpha}, \quad \sqrt{\frac{(4 \cdot 44 f)^{2} \gamma}{4 \cdot 44 f \times 10^{3}}}$$

$$E_{L} = \sqrt{\alpha}, \quad \sqrt{4 \cdot 44 f \times 10^{3}}$$
Volte.

1

The tratio of flux to full load mmf in a 400 kVA, 50 Hz, 10 core type transformer is 2-9×10<sup>6</sup>. Calculate the net ison area and window area of the transformer. Maximum flux density in the core is 1.3 wb/m<sup>2</sup>, Current density is 2.7 A/mm<sup>2</sup> and window space factor is 0.26. Also calculate the full load mmf.

Given data  $f = \frac{\Phi}{BT} = 2.4 \times 10^{6}, \quad @i = 400 \text{ kvB}, \quad f = 50 \text{ Hz}, \quad 14 \text{ core}$   $B_m = 1.3 \quad \omega b / m^2, \quad \delta = 2.7 \text{ A} / mm^2, \quad k_w = 0.26$ To find i) Net ison Area (Ai)

ii) Window Area (Aw)

iii) Fatt Full Load mmf (AT)

Solution

i) net 1200 Area (Ai)

$$B_{m} = \frac{\Phi_{m}}{A_{i}} \quad (o_{A}) \quad S_{f} = \frac{A_{i}}{A_{gi}}$$

From the given data, select  $B_m = \frac{\Phi_m}{A}$ 

$$A_{i} = \frac{\Phi_{m}}{B_{m}}$$

$$A_{i} = \frac{\Phi_{m}}{1\cdot 3} \rightarrow (1)$$

1.1

We know that  

$$E_{p} = 4 \cdot 44 \text{ f } \phi_{m} T_{p}, \text{ Volk}$$

$$E_{p} | T_{p} = 4 \cdot 44 \text{ f } \phi_{m}$$

$$E_{t} \Rightarrow \text{ Volt} | \text{ turn}$$

$$E_{t} = 4 \cdot 44 \text{ f } \phi_{m}$$

$$\phi_{m} = \frac{E_{t}}{4 \cdot 44 \text{ f }} = \frac{E_{t}}{4 \cdot 44 \text{ x 50}} \longrightarrow (2)$$

From the given data, we can express the Et interms of Qu

$$E_{t} = \sqrt{Q} \cdot \sqrt{4.44 f * x 10^{3}}$$

$$E_{t} = \sqrt{A00} \times \sqrt{(4.44 \times 50 \times 2.4 \times 10^{5} \times 10^{3})}$$

$$E_{t} = 14.5986$$
, Volts

From Equation (2)

$$\Phi_{m} = \frac{14.5986}{4.44 \times 50}$$

$$\Phi_m = 0.0658$$
, uab

From Equation ()

$$Ai = \frac{0.0658}{1.3}$$

$$A_{i} = 0.0506$$
, m<sup>2</sup>

ii) Area of the window (Bw)

Q in kVA = 2.22 f & S ko A to X10, KVA

Aw = 400 2.22× 50× 0.0658× 2.7×10× 0.26×10

$$A_{w} = \underbrace{0.0780}_{m} m^{2}$$

iii) Full load mmf (AT)

Criven 
$$\frac{\Phi_m}{AT} = 2.4 \times 10^6$$

$$AT = \frac{\Phi_m}{a \cdot 4 \times 10^6}$$

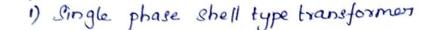
$$PT = \frac{0.0658}{2.4 \times 10^6} = 27416.6667$$
 AT

$$\frac{Answers}{A_{t}} = \underbrace{0.050b}_{0.050b}, m^{2}$$

$$A_{w} = \underbrace{0.0780}_{0.0780}, m^{2}$$

$$A_{T} = \underbrace{27416.6667}_{0.0780}, A_{T}.$$

Overall Dimensions of a transformer.



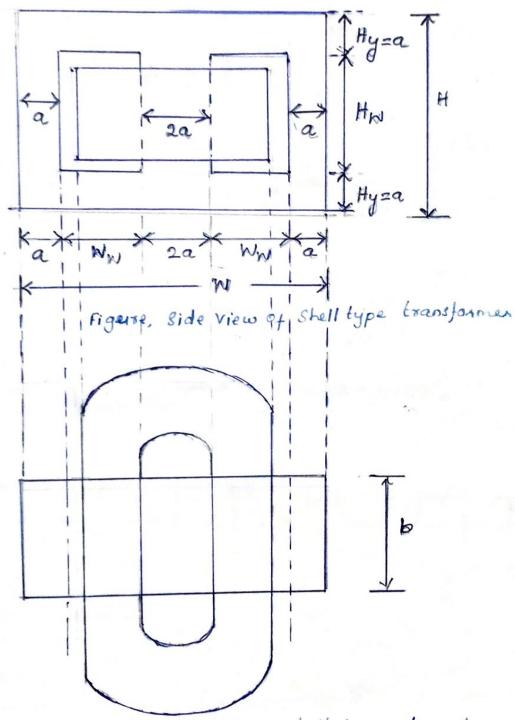
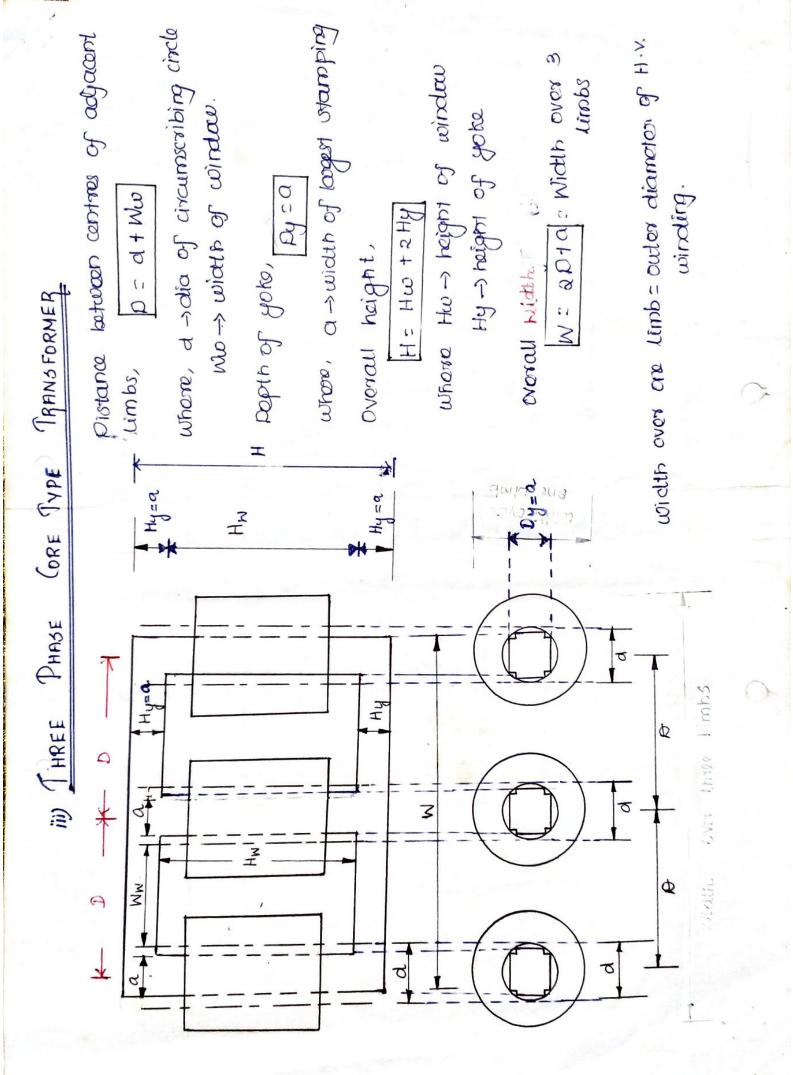


Figure Top view of Shell type toansformer

Depth of the Yoke Dy = b, Height of Yoke Hy = a Overall width W = 2Nw + 4a Overall Height H = 2Hy + Hw = Hw + 2a

10

(5)



Calculate the main dimensions and winding details of a 100 kVR, 2000/ 400 V, 50 ltz, 14 Shell type togs. Assume Voltage per turn is 10 V, flux dessily in the Core is 1.1 wb/m², Current densily is 2 A/mm², Window Space factors is 0.83, ratio window height to window space factors is 0.83, ratio window height to window space factors is 3 and the ratio of love depth to the width of central limb is 2.5 and the Stacking factors is 0.9.

Criven data

 $\begin{array}{l} (V = 100 \, \text{kVA}, \, V = \, 2000 \, \text{q00} \, \text{V}, \, \text{f} = 50 \, \text{H}_2, \, 1 \, \text{q} \, \text{shell type}, \\ E_{\text{f}} = 10 \, \text{V}, \, \text{B}_{\text{m}} = 1 \cdot 1, \, \text{g} = \, 2 \, \text{A} / \text{mm}^2 = \, \frac{2}{15^6} = \, 2 \, \text{x10}^6 \, \text{A} / \text{m}^2, \\ \hline 10^6 \, \text{K}_{\text{W}} = 0.33, \, \frac{\text{H}_{\text{W}}}{\text{W}_{\text{N}}} = 3, \, \frac{\text{b}}{2a} = \, 2 \cdot 5, \, \text{S}_{\text{f}} = 0.9 \end{array}$ 

To find

Main dimensions and winding details.

Solution

1) Main dimensions

The output Equation of 1¢ shell type the is  $Q_{1} = 2.22 \text{ fdm S kw Awx 10^{3}, kvA.}$   $A_{w} = \frac{Q}{2.22 \text{ fdm S km $ 10^{3}}}$  $A_{w} = \frac{100}{2.22 \text{ fdm S km $ 10^{3}}}$ 

2.22× 50× 0 12 × 2×10 × 0.33 × 103

From (3)  $\Phi_{m} = \frac{10}{4.44 \times 50}$  $\Phi_{m} = \frac{0.045}{0.045}$  wb

From ()

$$A_{\omega} = \frac{100}{2 \cdot 22 \times 50 \times 0.045 \times 2 \times 10^6 \times 0.33 \times 10^3}$$

$$A_{w} = 0.0303 \text{ m}^{2}$$

We know that

Hw=3NW From eq (A) Aw=3WNNW

$$W_N^2 = \frac{A_W}{3} = \frac{0.0303}{3} = 0.0101$$

$$N_{\rm N} = 0.1006 \, {\rm m}$$

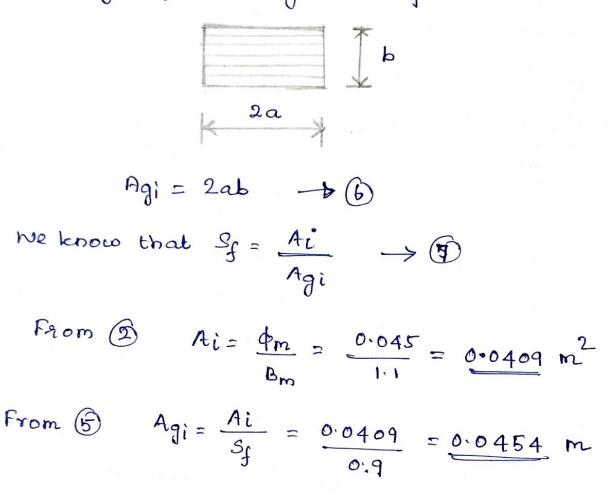
From eq (1) Missing in the

$$H_{N} = \frac{A\omega}{N\omega} = \frac{0.0303}{0.1006} = 0.3013 \text{ m}$$

Criven  $\frac{b}{2a} = 2.5$ 

$$\begin{array}{ccc} \vdots & a = \underline{b} & \longrightarrow & \textcircled{5} \\ & 5 & 5 \end{array}$$

In Shell type tfr, Rectangular shape core is used.



From (6) 2ab = 0.0454 -> (8)

Substitute 5 in equation (8)

$$2 \times \frac{b}{5} \times b = 0.0454$$

$$b^2 = \frac{0.0454 \times 5}{2} = 0.1135$$
  
b = 0.3369 m

From (5)

a = 0.0674 m

8) antidoripa di Carabilidadas

1811 0 = 1× A240 0 7

Determine the dimension of the core and yoke for a 200 KVA, 50 Hz, 10 Core type transformer. A stepped Core is used with distance between adjacent limbs segual to 1.6 times the width of core laminations. Assume Voltage per turn 14V, maximum flux densiby 1.1 wb/m window Space factor 0.32, current density 3 A/mm<sup>2</sup> and stacking factor = 0.9. The net iten area is 0.56d in a cruciform core, where d is the diameter of Circumscribing circle. Acto Also the widht of largest Stamping is 0.85d.

Que 200 kVA, f = 50 Hz,  $D = 1.6 \times a$ ,  $E_{\text{L}} = 14 \text{ V}$ , Bm=1.1 cob/m2, Kw=0.2, 8=3 A/mm2, Sf=0.9, Ai=0.56d2, a=0.85d

To find

Dimension of the core and yoke

Solution

Output Equation of 10 transformen is Qr=2.22fda8kwAwx10, kVA Aw = \_\_\_\_ Q

2.22 f 9 8 Kwx 103

## 200

We know 
$$B_m = \frac{\Phi_m}{A_i}$$
  
 $\Phi_m = B_m A_i$   
 $\Phi_m = 1.1 A_i \rightarrow @$   
From the given data  
 $A_i = 0.56 d^2 \rightarrow @$ 

Also we know that

render Ersten

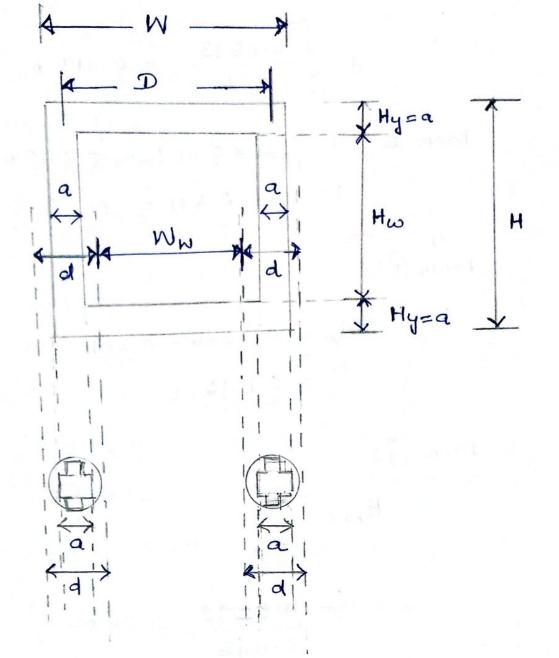
Print Strate

$$A_{\omega} = \frac{200}{2.22 \times 50 \times 0.063 \times 3 \times 10^6 \times 0.32 \times 10^3}$$

. .

$$A_{10} = 0.0298 m^2$$

$$A_{W} = H_{N} N_{N} \longrightarrow (4)$$



9

From the figure

$$W_{10} = D - d_{12} - d_{12}$$

$$W_{10} = D - d. \rightarrow (5)$$

viven D=10bq and a=0.85d -> (6)

From 2

$$A_{i} = \frac{\Phi_{m}}{11} = \frac{0.063}{11} = 0.0573 \,\mathrm{m}^{2}$$

From 3

0.0573 = 0.56 d<sup>2</sup>

 $d = \int \frac{0.0573}{0.56} = 0.319 \text{ m}$ 

From (a)  $q = 0.85 \times 0.319 = 0.2712 \text{ m}$  $D = 1.6 \times 0.271 = 0.4336 \text{ m}$ 

From 5

$$W_{W} = 0.4836 - 0.319$$
  
= 0.1146 m

From (4)

$$H_{N} = \frac{A_{10}}{W_{10}}$$
  
=  $0.0298 = 0.26 \text{ m}$   
 $0.1146$ 

From figure

$$W = D + 2a$$
  
= 0.4336 + (2×0.2712)  
$$W = 0.976 m$$
  
$$H = Hw + 2Hy$$
  
= Hw + 2a  
= 0.26 + (2×0.2712)  
$$H = 0.8024 m$$

Answers

1

$$A_{10} = 0.0298 m^{2}$$

$$H_{10} = 0.26 m$$

$$W_{10} = 0.1146 m$$

$$D = 0.4836 m$$

$$d = 0.319 m$$

$$q = 0.2712 m$$

$$H = 0.8024 m$$

$$W = 0.976 m$$

$$H_{11} = D_{11} = q = 0.2712 m$$

Determine the main dimensions of the core of a 5 kvA, 11000/4001, 50 Hz, 10 Corretype distribution transformer The net Conductor area in the window is 0.6 times the net coross Section area of iron in the core. The Core is of square Cross Section, mascimum fluse density is 1 wb/m<sup>2</sup>, Current density is 1.4 A/mm<sup>2</sup>. Window Space factor is 0.2. Height of the window is 3 times its width.

Crives daba

Qr=5kVA, 11000/400, f=50 ltz, 1¢ core type, Ac=0.6 Ai, Core  $\Rightarrow$  square, Bm=1 cob/m<sup>2</sup>, 8=1.4 A/mm<sup>2</sup>  $k_{w} = 0.2$ ,  $l_{tw} = 3 W_{w}$ 

To find

Main Dimensions

Salution

The output Equation of 10 core type trans transformer is given by

$$Aw = \frac{\omega}{2\cdot22} f \phi m \delta k w \times 10^3$$

$$A_{w} = \frac{5}{2.22 \times 50 \times \phi_{m} \times 1.4 \times 10^{6} \times 0.2 \times 10^{3}}$$

$$\Phi_{m} \xrightarrow{B_{m}} B_{m} = \frac{\Phi_{m}}{A_{i}}, \quad \Phi_{m} = B_{m}A_{i} \rightarrow \textcircled{2}$$

$$\Phi_{m} \xrightarrow{E_{t}} E_{t} = 4.44f \Phi_{m}, \quad \Phi_{m} = \frac{E_{t}}{4.44f}$$

Criven Ac= 0.6 Ai

$$ti = \frac{Ae}{0.6}$$

From 2

$$A_{c} = K_{N} A_{W}$$
  
 $A_{c} = 0.2 A_{W}$ 

From eq 3

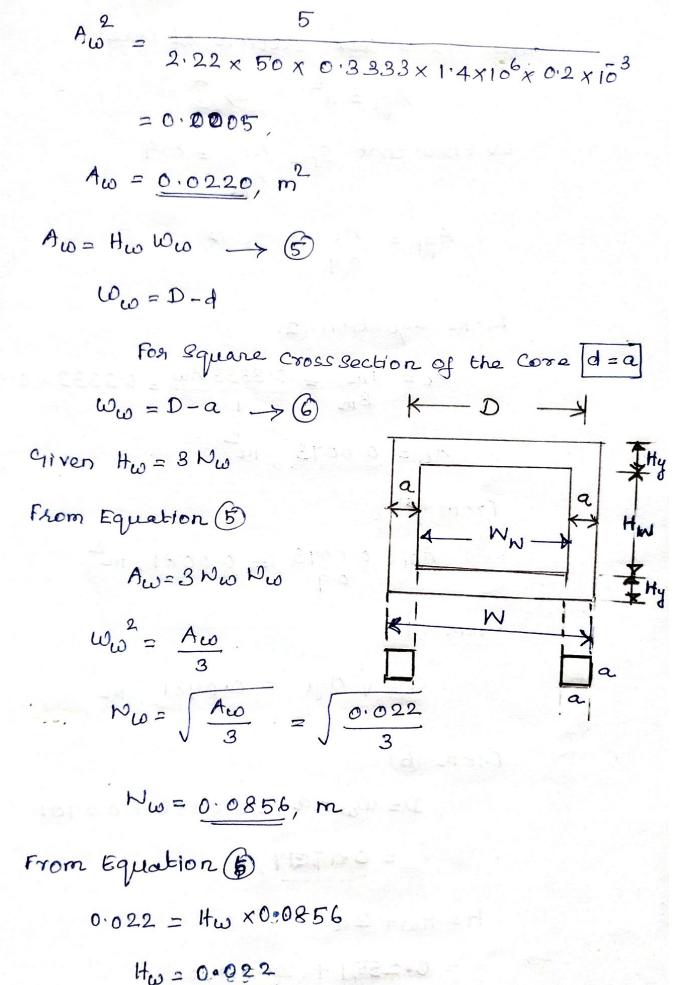
$$P_m = \frac{1 \times 0.2 A_{10}}{0.6}$$

$$\Phi_{m} = 0.3333 A_{\omega} \longrightarrow \Phi$$

From eq ()

 $A_{\omega} = \frac{1}{2 \cdot 22 \times 50 \times 0.3333} A_{\omega} \times 1.4 \times 10^{6} \times 0.2 \times 10^{3}$ 

5



$$H_{\omega} = 0.2511$$
, m.

For square cross section of core

We know that  $S_f = \frac{Ai}{Agi} = 0.9$ 

$$Agi = \frac{Ai}{0.9} \longrightarrow \textcircled{B}$$

From equation 3

 $A_{i} = \frac{\Phi_{m}}{B_{m}} = \frac{0.3333 \, A_{10}}{1} = 0.3333 \times 0.022$   $A_{i} = \frac{0.0073}{1}, m^{2}$ 

From (8)

$$Ag_{i} = \frac{0.0073}{0.9} = \frac{0.0081}{0.0081}, m^{2}$$

From D

$$a = \sqrt{Agi} = 0.0901$$
, m

From 6

D= Ww + a = 0.0856 + 0.0901

 $H = H_{W} + 2a$ = 0.2571 + (2×0.0901) H = 0.4373 m

$$W = D + a$$
  
= 0.1757 + 0.0901  
= 0.2658, m

 $A_{\omega} = 0.022, m^{2}$   $H_{\omega} = 0.2571, m$   $N_{\omega} = 0.0856, m$  a = 0.0901, m D = 0.1757, m N = 0.2658, m H = 0.4373, m  $H_{\omega} = 0.9901, m$ 

Ans

Calculate the KVA output of the 14 transformer from the following data

Corre height to the distance between corre = 2.8, diameter of ciacumsribing ciacle to the Distance between corre centre is 0.56. Net iron area to the area of circum cribing Circle is O.T, Current density = 2-3 A/mm2 Frequency 18 50 Hz, window space factor is 0.27. flux densily In the core is 2 wb/m² and Distance b/co core centre 13 004 m.

Criven

core height to the distance b/w core centre Hw = 2.8 Diameter of circumscribing circle to the Distance between core centre d = 0.56 Net iron area (Ai) = 0.7 Area of Ciacumscribing Circle 8 = 2.3 A/mm<sup>2</sup> kw= 0.27  $f = 50 H_2$ Bm = 12 wb/m2 D= 0.4 m To find Qr in KVA

Solution

2

For 1 & transformer  

$$\begin{aligned}
& \varphi = 2 \cdot 22 \oint \phi_m & k_{10} A_{10} \times 10^3 \text{ kVD} \\
& \varphi = 2 \cdot 22 \times 50 \times \phi_m \times \frac{2 \cdot 3}{10^6} \times 0.27 \times A_{10} \times 10^8 \text{ kVD} \\
& \varphi = 2 \cdot 22 \times 50 \times \phi_m \times \frac{2 \cdot 3}{10^6} \times 0.27 \times A_{10} \times 10^8 \text{ kVD} \\
& \to 0
\end{aligned}$$

$$\begin{aligned}
& F_t = 4 \cdot 44 \oint \phi_m & (0.9, ) B_m = \frac{\phi_m}{A_i} \\
& \phi_m = B_m \times A_i \\
& \phi_m = B_m \times A_i \\
& \phi_m = 1 \cdot 2 \times A_i \longrightarrow \textcircled{O}
\end{aligned}$$

$$\begin{aligned}
& Criven \quad A_i \\
& Area of CiscumScribing Ciscle
\end{aligned}$$

$$\begin{aligned}
& A_i \\
& \pi (\frac{4}{2})^2 = 0.7 \\
& A_i = 0.7 \times \frac{\pi}{4} \times d^2 \longrightarrow \textcircled{O}
\end{aligned}$$

$$\begin{aligned}
& \text{Given} \quad \frac{d}{D} = 0.56 \\
& d = 0.56 D \\
& = 0.56 \times 0.4 \\
& d = 0.224 m
\end{aligned}$$

From 3

$$A_i = 0.028 m^2$$

From 3

Aw= HWXNN > (1)

 $w_{10} = D - d = 0.4 - 0.224 = 0.176 m$ 

given 
$$\frac{H_{W}}{D} = 2.8$$

$$H_{w} = 2.8 \times 0.4 = 1.12 \text{ m}$$

From (4)

From eq (1)

Qr= 2.22 × 50 × 0.033 × 2.3 × 0.27 × 0.19 × 10 106

Qr = 448.12 KVA

calculate the approximate overall dimensions for a 200 kvA 6600/420 V, 50H2, 3¢ Core type transformer. The following data may be assumed as Emf | turn = 10V, maximum flux density is 1.3 Wb/m², Current density is 2.5 Almm², window Space factor is 0.3, Overall height is equal to overall width, Stacking factor is 0.9, use asstepped core.

Fog a three Stepped core: width of largest Stamping = 0.9d, and Net iron area = 0.6d<sup>2</sup>, where d is the diameter of circumscribing circle.

Criver

 $Q_{i} = 200 \text{ kvA}$ , Voltage rating = bboo|420 V, f = 50 Hz,  $3\phi$   $E_{t} = 10 \text{ V}$ ,  $B_{m} = 1.3 \text{ wb} m^{2}$ ,  $\delta = 2.5 \text{ A}[\text{mm}^{2}, \text{ kw} = 0.3]$ , H = W,  $S_{f} = 0.9$ , a = 0.9 d,  $A_{i} = 0.6 \text{ d}^{2}$ . To find

Overall dimensions?

Solution

Fora 30 transformer

$$A_{\omega} = \frac{Q_{\omega}}{3\cdot 33 \times f \times q_{m} \times 8 \times A_{\omega} \times k_{\omega} \times 10^{3}}$$

$$A_{w} = \frac{200}{3.33 \times \Phi_{m} \times 50 \times 2.5 \times 10^{6} \times 0.3 \times 10^{3}}$$

$$\Phi_{m} \longrightarrow B_{m} = \Phi_{m} \Rightarrow \Phi_{m} = B_{m} A_{i} \rightarrow \textcircled{2}$$

$$\Phi_{m} \longrightarrow E_{t} = 4.44f \Phi_{m} \Rightarrow \Phi_{m} = \frac{E_{L}}{4.44f} \rightarrow \textcircled{3}$$

SG

From 
$$\phi_m = \frac{E_L}{4.44f} = \frac{10}{4.44\times 50}$$

$$P_m = 0.045 \text{ cob}$$

From Equation ()

 $A_{\omega} =$ 

2.33 × 0.045 × 50 × 2.5 × 10 × 0.3 × 103.

$$Aw = 0.0356 m^2$$

$$A_{\omega} = H_{\omega} \omega_{\omega} \rightarrow (\mathbf{a})$$

We know that

$$\mathcal{W}_{\omega} = \mathcal{D} - d \rightarrow \mathfrak{F}$$

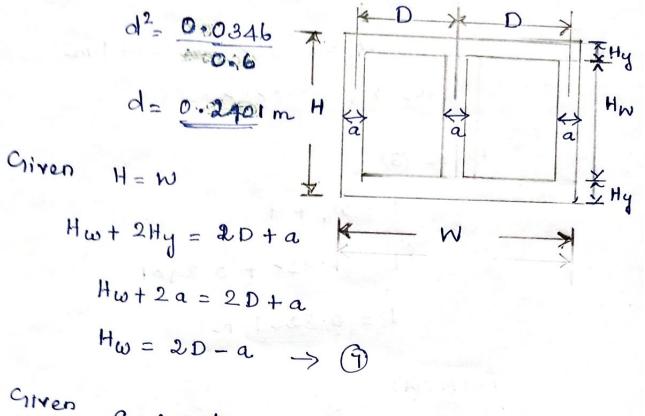
Given

$$A_i = 0.6 d^2 \rightarrow 6$$

From Equation (2)

$$A_i = \frac{\Phi_m}{B_m} = \frac{0.045}{1.3} = \frac{0.0346}{0.0346} m^2$$

From Equation ()



iven (

$$a = 0.9 d$$
  
 $a = 0.9 \times 0.2401$   
 $a = 0.2161 m$ 

From eq G

$$H_{w} = 2(w_{w} + d) - a$$

$$H_{W} = 2W_{W} + (2 \times 0.2401) - 0.2161$$
$$H_{W} = 2W_{N} + 0.2641$$

From eq (1)

$$0.0356 = (2W_{W} + 0.2641) W_{W}$$
  
 $0.0356 = 2W_{W}^{2} + 0.2641 W_{W}$ 



Ww=0.0828, -0.2149 m

take Ww = 0.0828 m

From (3)

 $D = \omega_{\omega} + d$ 

D=0.0828 + 0.2401

D = 0.3229 m

From A

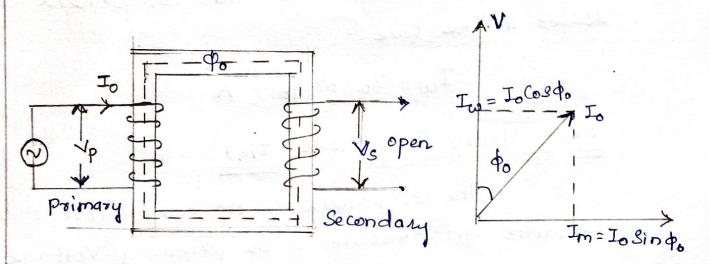
$$Hw = 0.43 m$$

## Answers

 $A_{w} = \underbrace{0.0356}_{Hw} m^{2}$   $H_{w} = \underbrace{0.43}_{Hw} m$   $W_{w} = \underbrace{0.43}_{Ww} m$   $D = \underbrace{0.0828}_{Ww} m$   $D = \underbrace{0.3229}_{Ww} m$   $a = \underbrace{0.2161}_{Ww} m$   $H = H_{w} + 2a = 0.43 * (2 * 0.2161) = \underbrace{0.8622}_{Ww} m$   $W = 2D + a = (2 \times 0.3229) + 0.2161 = \underbrace{0.8619}_{Ww} m$   $H_{y} = D_{y} = a = \underbrace{0.2161}_{Ww} m$ 

## No laad of a transformer

When there is no load, in Secondary side, the transformer is said to be under no load or open circuit. Since the primary winding is connected to the supply, a small current is drawn from the supply is called no load current.



a) ciscuit diagram c) Phasos diagram

B8 the primary winding is both Resistive and Inductive, the no load current lags the primary Voltage by an angle 90. This angle 90 is called no load power factor angle (08) hysteresis angle of advance.

The no load current Io can be resolved into two components like loss component and magnetising components.

Jo = Iio + Im

 $T_0 = \sqrt{I_m^2 + I_w^2}$ 

Loss Component (Iw)

The component of no load current (Iw) Which remains inphase with primary Voltage is Callec loss component. It is also called active (or) working (or) wattfull component. This component causes losses in the core.

Iw= Io cos po, A.

Magnetising Component (Im)

The component of no load current which remains quadrature with primary Voltage is Called magnetising Component. It is also called reactive (or) wattless component. This component causes flux in the core.

Im= Io Sinpo, A

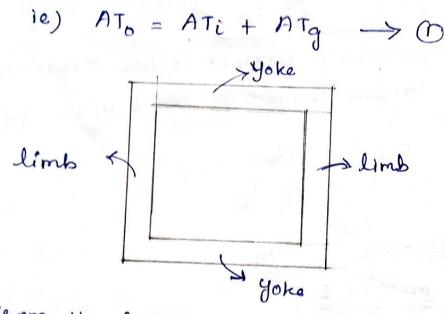
Noload current (Io) fog 1¢ core type transformer

A Calculation of magnetising component 
$$(I_m)$$
  
The magnetising mmf required at no load is  
given by ATo (or) ATm = ImTp  
 $I_m = AT_0$   
 $T_p$   
Where Tp is the number of primary turns.  
The R m.s value of magnetising current at no load  
is given by  
 $I_m = AT_0$ , A  
The magnetising current is not Sinuscidal, therefore  
the peak factor should be used in the place  $\sqrt{2}$ .  
 $I_m = AT_0$   
 $k_p$ . Tp  
Calculation of ATo  
Method I  
we know that mmf = Rdux x Reluctance.  
 $AT_0 = \Phi_m \times S$   
 $AT_0 = \Phi_m \times S$   
 $AT_0 = \Phi_m \times \frac{L_0}{L_1 + L_2} \Rightarrow AT_0 = \Phi_m \times \frac{L_0}{L_0 + L_3}$   
Where  $\Phi_m \Rightarrow$  flux in the Corre., wb

le -> length of the core = (2 × length of the dimb) + (2 × length of the Yoke)., m Mg > relative permeablity of the Core. No -> 47 × 10<sup>7</sup> Ac -> gret area of the Core, m<sup>2</sup>

## Method I

Mmf Required to produce the flux in the Core = mmf equired for iron path + mmf required for jointe (gap)



From the figure; mmf roquised for ison path is given by AT: = 2 atyly + 2 atyly > 2 aty > magnetizing force required for yoke, AT/m by > length of flix path in the yoke, m

ate > magnetizing force required for limb, AT/m li > length of flux path in the limb, m

Mmf Required for 
$$jaints(gap) = 8,00,000$$
 Bg/g/g  
Where  
Bg  $\Rightarrow$  maximum flux density,  $wb/m^2$   
dy  $\Rightarrow$  ais gap length, m  
kg  $\Rightarrow$  crap contraction factor  
 $= 1$ , for transformer.  
**b)** Calculation of loss component (Ind)  
The loss component at any load is given  
by  
 $Wo = IwVp$ , watts  
 $\boxed{Iw = \frac{Wo}{Vp}}$ , A  
where  
 $Mo \Rightarrow$  power loss at no load. (concloss)  
 $Vp \Rightarrow$  primary Voltage.  
Let  $Wg \Rightarrow$  Specific iron loss, watts | kg  
 $Di \Rightarrow$  Beigsity of iron, kg/m<sup>3</sup>  
 $Vi \Rightarrow$  Volume of iron in the Core, m<sup>3</sup>  
 $= Area of$  iron x length of iron in the Core

Power loss at no load = Ws x Di x Vi, Watts (No)

•••

A 14,400V, 50 Hz, transformer is built from Stampings have a relative permeablity of 1000, the length of the flux path is 2.5 m. The area of Cross section of the Core is 2.5 × 10<sup>3</sup> m<sup>2</sup> and the primary winding has soo tuans. Estimate the maximum flux density and the no load arrivent of the transformer. The Iron loss at the working flux density is 2.6 watts | kg. Iron weight is 5.8 × 10<sup>3</sup> kg/m<sup>3</sup>, stacking factor is 0.9.

## Given data

14, Ep=400V, f=50 Hz,  $M_{A}=1000$ , length of the flux path = 2.5 m, Area of Cross Section for the Core = 2.5 x  $10^{3}$  m<sup>2</sup>, Tp=800, Iron lass = 2.6 watts' /kg, Iron weight =  $5.8 \times 10^{3}$  kg/m<sup>3</sup>, Sf = 0.9 To find

- 1) Maximum flux density (Bm)
- ii) No load current (ID)

Solution

$$D = B_m = \frac{\Phi_m}{A_i}$$

we know that, Ep= 4.44 f \$m Tp

Ep/Tp= 4.44 f Pm

$$\frac{400}{800} = 4.44 \times 50 \times 4m$$

$$0.5 = 4.44 \times 50 \times 4m$$

$$\frac{0.5}{4m} = \frac{0.5}{4.44 \times 50}$$

$$\frac{1}{4m} = \frac{0.0023}{4.44 \times 50}$$

$$\frac{1}{4m} = \frac{0.0023}{4.44 \times 50}$$

$$\frac{1}{4m} = \frac{0.0023}{4.44 \times 50}$$

$$\frac{1}{4m} = \frac{0.0023}{4m}, \text{ obs.}$$

$$\frac{1}{4m} = \frac{0.0023}{4m}, \text{ obs.}$$

$$\frac{1}{4m} = \frac{0.0023}{4m}, \text{ obs.}$$

$$\frac{1}{4m} = \frac{0.0023}{4m}, \text{ m}^{2}$$
From Equation (1)
$$\frac{1}{4m} = \frac{1}{4m} \frac{1}{4m} \frac{1}{4m} \rightarrow \infty$$

$$\frac{1}{4m} = \frac{1}{4m} \frac{1}{4m} \frac{1}{4m} \rightarrow \infty$$

$$\frac{1}{4m} = \frac{1}{4m} \frac{1}{4m} \rightarrow \infty$$

.

$$I_{\omega} = \frac{P_i}{V_p(\Theta_{\mathcal{A}})E_p} \rightarrow \textcircled{A}$$

$$AT_{0} = \frac{0.0023 \times 2.5}{(4\pi \times 10^{7}) \times 1000 \times 0.0023}$$

AT. = 1989.4368, AT.

From Equation 3

$$T_{m} = \frac{1989.4368}{\sqrt{2} \times 800}$$

Given Iron Loss = 2.6 watts/kg and Iron weight =  $5.8 \times 10^3$  kg/m<sup>3</sup>.

Therefore Iron loss in watts (Pi) = Iron loss in watts/kg × Iron weight in kg/m × Volume af iron Iron Loss (Pi) in watts = 2.6 x  $(5.8 \times 10^{3}) \times A_{c} L_{c}$ = 2.6 x  $(5.8 \times 10^{3}) \times 0.0023 \times 2.5$ Pi = 86.71 watts

From Equation (a)  $I_{w} = \frac{86.71}{400} = 0.2168, A$ 

From Equation (2)  $I_0 = \sqrt{(0.2168)^2 + (1.7584)^2}$ 

 $I_0 = \underbrace{1.7717}_{A}$ 

Answers

$$B_{m} = 1 |\omega_{b}| m^{2}$$

$$I_{m} = 1.7584, A$$

$$I_{w} = 0.2168, A$$

$$I_{o} = 1.7717, A.$$

105-

Temperature rise in transformers

When the boarsformer is in woosking conditions, losses are occurs in the transformer core and windings are converted into thermal energy and couse heating of corresponding transformer parts.

The path of heat flow is

i) From the integnal most heated spots to the outer surfaces in contact with the oil (conduction)

ii) From the outer Surface to the oil that Cools it (convection)

iii) From the oil to the walls of the tank (convection)
iv) From the walls of the tank to the cooling medium
(by convection and radiation).

To keep the temperature rise within the safe limit, the heat produced should be transferred to a cooling medium like air or water.

when the transformer Hating is high, the natural aig cooling method is not effective and hence oil immersed type of cooling is employed.

The specific heat dissipation due to convection of oil is

 $\lambda_{conv} = 40.3 \left(\frac{\alpha}{H}\right) in w/m^2.$  c

Where

and temperature rise of the transformer above ambient temperature, °c

H -> Height of dissipating Surface, m

From the test desults, It has been observed that the heat dissipation of oil due to convection is to times higher than heat dissipation by air due to Convection. Therefore in transformer, oil is used as a Cooling medium.

The walls of the tank dissipate the heat by both radiation and convection. From the test results, the plain tank Surface dissipates the heat by radiation is 6 w/m<sup>2</sup> ° and convection is 6.5 w/m<sup>2</sup> ° respectively. The Specific has here in

The Specific heat dissipation by tank walls due to Convection and Radiation ( $\beta_{conv}$  and rad) = 12.5  $\omega/m^2$ °C

Temperature Dise  $(a) = \frac{Q_{t}}{\Lambda S_{t}}$  $Q_{t} \neq \text{total loss} = \text{Pi+Pc}$  $\Lambda \Rightarrow \text{Specific heat dissipation}$ 

= 12.5 w/m², °c.

St > Surface area of the tank.

Temperature rise 
$$(a) = \frac{P_i + P_c}{12.5 S_L}$$

# Design of tank with Cooling tubes

The transformers are provided with Cooling tubes to increase the heat dissipating areq. The tubes are mounted on the Vertical Sides of the transformer tank.

The heat dissipating surface area of the tank be  $S_t$  $S_t = 2(H_t \times L_t) + 2(H_t \times W_t)$ , m

Let,

The heat dissipated by the surface of the tank due to Convection and radiation =  $(6+6.5) S_{t}$ = 12.5 S<sub>t</sub>

The heat dissipating surface of the cooling tube be taken as  $xs_t \rightarrow 0$ 

But the increase in dissipation of heat is not proportional to increase in area, Because the tubes would block the heat dissipation by the tank due to radiation. On the other hand, the tubes will improve the Circulation of a oil due to more effective heads produced by columns of oil in tubes. Therefore the Specific heat dissipation due to convection is taken as 35% more than that without tubes.

Therefore the heat dissipation by the surface of the cooling tube due to convection = 6.5 × 135 × 25 = 8.8 x SE

Temperature rise in transformer with tank and Cooling tubes 1 = Total loss

Total heat dissipation

$$O(c) = P_i + P_c$$

Heat dissipation by tank and tubes

$$a = \frac{P_T}{12.5 s_t + 8.8 x_{s_t}}$$

 $O(12.5S_{t} + 8.8XS_{t}) = P_{T}$ 

of all in

 $12.5S_{t} + 8.8 \text{ Dc} S_{t} = -\frac{P_{T}}{2}$ on herause in arreal Bacause the 8.8 DCSt = - 12.53L based 12 dds  $\mathcal{R} = \frac{1}{8 \cdot 8 s_{t}} \left[ \frac{P_{T}}{a^{\circ}} - 12 \cdot 5 s_{t} \right]$ 

From Equation ()

The heat dissipating surface of the cooling tube =  $\frac{1}{8.8s_t} \left[ \frac{P_T}{a^\circ} - 12.5s_t \right] s_t$ 

Surface area of each tube = Tollt

Total number of tubes  $n_t = Total area of cooling tubes$ Area of each tube.

The Standard diameter of the Cooling tubes is 50 mm and the length of the tube depends on the height of the tank. The tubes are arranged with a Centre to Centre Spacing of IS mm. The tank of 1250 kvA, natural oil cooled transformed has the dimensions length, width and height as  $0.65 \times 1.55 \times 1.85$  m respectively. The load loss = 13.1 ku loss dissipation due to radiations  $6 \text{ M/m}^2$ ° c, loss dissipation due to convection =  $6.5 \text{ M/m}^2$ . The Improvement in convection due to provision of tubes =  $40^{\circ}/_{0}$ , temperature size is  $40^{\circ}$ c, length of each tube is 1 m, diameter, of each tube is 50mm Find the number of tubes for this Convection. Negled the top and bottom Surface of the tank as regards the cooling?

Given data

Qual 250 KVB, Dimensions = 0.65 × 1.55 × 1.85 m Total loss = 13.1 kW,  $\lambda con = 6.5 \text{ M/m}^2 c$ ,  $\lambda_{rad} = 6$   $\omega/m^2 c$ ,  $\alpha = 40^{\circ}c$ , Improvement in cooling due to tubes = 40°/.,  $L_{t} = 1m$ ,  $d_{t} = 50 \text{ mm}$ To find

no of tubes and arrangement of tubes Solution

Temperature rise  $(a) = \frac{Total loss}{(\lambda_t S_t + \Lambda_{ct} S_{ct})} \rightarrow 0$  $\lambda_t \rightarrow Specific heat dissipation of the tank by radiation and convection$ 

$$\lambda_{t} = 6 + 6.5 = 12.5 \text{ is} / m^{2} \circ_{c}$$

$$S_{t} \Rightarrow \text{Surface area of the trank}$$

$$S_{t} = 2.H_{T} (U_{T} + L_{T})$$

$$= 2 \times 1.85 (1.55 + 0.65)$$

$$S_{t} = \underbrace{\$.14}_{m} m^{2}$$

$$\lambda_{ct} \Rightarrow \text{Specific heat dissipation of Cooling tube}$$
by Convection
$$\lambda_{ct} = 6.5 \times \underbrace{140}_{100} = 9.1 \text{ is} / m^{2} \circ_{c}$$

$$S_{ct} \Rightarrow \text{Surface area of the Cooling tubes}$$

$$\text{Let us take } S_{ct} = \underbrace{\times S_{t}}_{m} \rightarrow \underbrace{\$}$$
From Equation (1)
$$40 = \underbrace{13.1 \times 10^{3}}_{101.15 + 14.014 \times}$$

$$40 (101.15 + 74.074 \times) = 13.1 \times 10^{3}$$

$$4070 + 2962.96 \times = 13.1 \times 10^{3}$$

$$3912.94 \text{ is} \quad \text{Area}$$

 $2962.96 x = (13.1 \times 10^3) - 4070$ 

2962.96 
$$X = 9030$$
  
 $X = 3.0476$   
From Eq (2)  
Surface area of the Cooling tabe( $\pounds_{CL}$ ) =  $3.0476x$   
 $8.14$   
 $= 24.8077$ , m<sup>2</sup>  
Area of each tabe =  $Td_{L}L_{L}$   
 $= T \times (50 \times 10^{3}) \times 1$   
 $= 0.1571$ , m<sup>2</sup>  
Total number of cooling ( $Ta_{L}$ ) =  $\frac{Total area of tabes}{area of each tabe}$ 

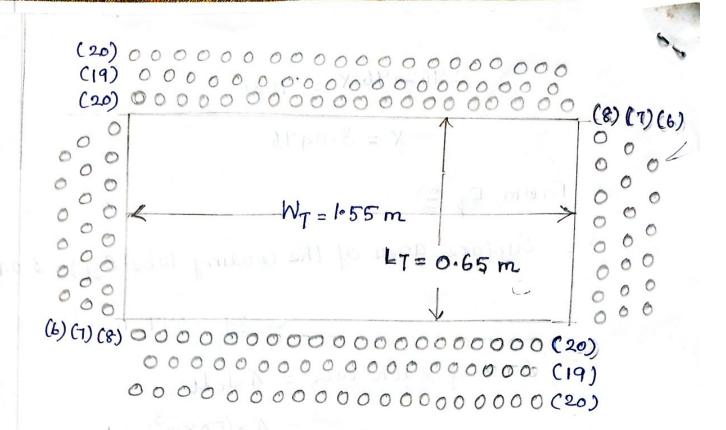
$$n_{t} = \frac{24.8077}{0.1571}$$

Sta Last

nt = 157.9102 2 158

This tubes are placed in vertical position along widthwise and length of the tark with a Centre to Centre specing of 45mm.

No of tubes along length wise = 0.65  $75 \times 10^3 = 8.6667$   $= 8 \ M$ No of tubes along width wise  $= \frac{1.55}{75 \times 10^3} = 20.6667$  $= 20 \ M$ 



Number of tubes in the first row along Widthwise and lingthwise = 2(20+8) = 56Number of tubes in the first row and second row along widthwise and lingthwise = 2(20+8)+2(19+7)= 108

Number of tubes in the first row, second row and third row along widthwise and length wise = 2(20+8) + 2(19+1) + 2(18+6) = 156Here Namber of tubes arranged is less than number of tubes required ... Number of tubes arranged in first, second and third row along widthwise and lengthwise = 2(20+8) + 2(19+1) + 2(20+6) = 160<u>Ans</u> The number of tubes provided = 160 //

#### **EE 8002 DESIGN OF ELECTRICAL APPARATUS**

#### <u>UNIT III</u>

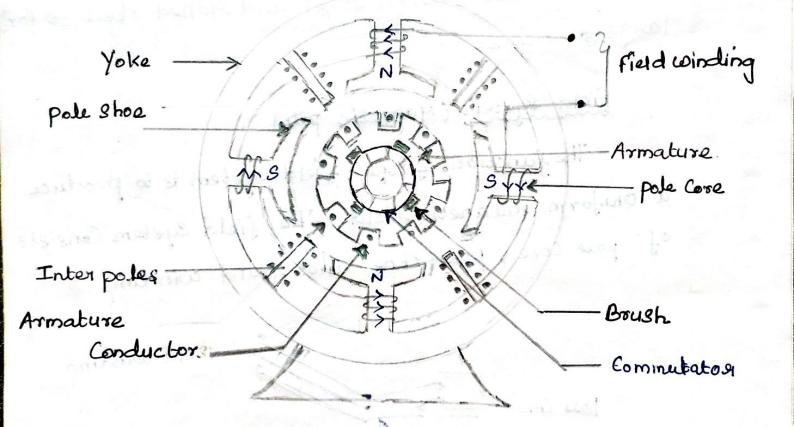
#### **DESIGN OF DC MACHINES**

Prepared by Dr . T. Dharma Raj Asso.Prof /EEE Constructional details of a Dc machine

A DC machine has two main parts; States, and Hotog.

The Staton is the non-notating part and Consists of field System, Moke, Interpoles and Brushes.

The rotag is the rotating part and consists of Armature System, Commutator and Bearings



Yoke

It is the outer layer of the DC machine which protects the parts of the machine against dust, moisture and Various gas like Soz, Acidic fumes entering into it. It provider a mechanical support to the poles It provides a path of magnetic flux produced by the field system.

The magnetic material used for making yoke is must to provide low reluctance path for magnetic flux. Hence the material selected was

Cast iron -> for Small machines

Cast steel, silicon steel, rolled stell > for Lange Machines.

Field System

The field system Consists of pole core, pole shoe and field windings.

Pole Core Carries the field winding and directs the magnetic flux through aig gap , armature and to mexit pole.

To reduce the core losses, the pole cores are made up of magnetic material Called Cast inon (or) Cast steel with laminated construction. These laminations are stacked and stamped together to form pole core, The pole core is fitted to the Yoke by means bolts or welding. The curried surface at the end of the pole core towards the armature is called pale shoe. It will serve two purposes.

i) It offer low reluctance for the magnetic flux, because of that magnetic flux are uniformly distributed over the air gap.

ii) It provides the mechanical support to the field windings.

also exciting which is Field winding is called exiting winding made up of copper. This field windings are wounded on around the pole core and Causing the pole core as electro magnet, when the field windings carrying current.

The field Coils are connected in Series in Such a way that adjacent poles have opposite polarity

## Inter poles

pally. mus

Interpoles are usally made by Cast iron or Cast steel and is placed between two adjacent main poles. Just like field winding, the interpoler also have exciting Coils which are connected in Series with the agmatuse. It will serve two purposes

i) Reduce the effect due to Armature Reaction ii) Improve the Commutation. Armature System

The armature System Consists Are of armature Core and armature windings

Armature core is in Cylindrical shape and having alteenate slot and teeth on its outerperiphery. The an slots are provided for place the armature windings.

The asimature cose along with asmatuse Conductors are rotated in the uniform magnetic field produced by the field system

To reduce the core losses, the 'Armature core are made up of magnetic material Called Cast iron (or) Cast steel with laminated construction. These laninations are stacked and stamped together to form Armature Core.

## Armature windings

Armature windings are usually made of Copper. There are two types of sommature windings namely (i) Lap winding and (ii) Nowe winding. In Lap a type Armature winding, the armature Coils are divided into non-number of parallel paths which equal to number of poles. This type

Selon rodel

of winding is preferred for low voltage, high arreal applications.

In wave type Armature winding, the armature Coils are divided into number of parallel path which is equal to 2. This type of winding is preferred for high voltage, low current applications.

#### Commutator

It is in the Cylindrical Structure and made up of Copper Segments. But these Copper segments are insulated from each other by thin Sheet of Mica.

Its function is to collect the current from the Armature conductors and convert this alternating Current into unidirectional Corr Current. So this Commutator is called Mechanical Rectifier.

### Bruches

It is in rectangular shape, made by Carbon or graphite. The Brushness are placed in brush holders. A spring action in the Brush holder maintain a proper Contact of Brush over the Rotating Commutators.

Its function is to collect the current from Commutator and make it available to the load in the case of DC Generator.

ii) It supplies the current from the external Circuit to Armature windings in case of DC motor.

## Bearings

Real Blocks Coppen Sognal and an

It function is to provide Smooth rotation of armature system. The different types of bearings are ball and holley bearings.

really again of a light in the

Because of their reliablity ball bearings are frequently used. sources reddes for de

insulated from earth other by this sheet of hises . Its foundation is to College the Current from Ar water Conductors and Convert Hill all with and bing suis de l'ansient que constant sa étue 1.60 Committation is called mechanical Rectifiers

Parsbarg

Je Br in rectanged as shaped, inde by rather an graphile. The bruckness we placed in bouch helders' A spring addien in the lower's harded maintain a paper Contact of brack over the Robath commutation.

The function is to callect the Garrent from Remainstate and make it considered to the land. in

#### Main Dimensions

The adjuncture diameters and armature Core length are known as the main dimensions of a Motating machine.

Total Loadings and Specific Loadings.

Total loadings can be divided into two types.

- (i) Total magnetic loading
- (ii) Total Electric loading

Total magnetic loading

The total flux around the armabure pheriphery at the air gap is called total magnetic logding.

Potal magnetic loading = P¢ where

P → nomber of poles.
\$\phi\$ → flupc per pole, ub

Total Electric Loading.

The total number of ampere conductors around the armature periphery is called total electric loading.

where

Iz → Current flow in each conductor, A z → total number of asmature Conductor

#### Specific Loadings

It can be divided into two types i) Specific Magnetic Loading ii) Specific electric Loading

Specific magnetic loading (Baw)

It is defined as the ratio of total flux around the aisgap to the Area of flux path at the aisgap.

 $B_{av} = \frac{\text{Total flux around the aisgap}}{\text{Area of flux path at the aisgap}}$  $B_{av} = \frac{P\phi}{\pi DL}$ 

Specific Electric Loading (ac)

It is defined as the ratio of total ampere Conductors to the Armature periphery at airgap  $AE = \frac{Total \ ampere \ Conductors}{Armature \ periphery \ at \ air \ gap}$ .  $ac = \frac{I_z \cdot z}{\pi D}$ . choice of specific magnetic loading in DC machine.

The Specific magnetic loading is defined at the ratio of Total flux around the alogap to Area of flux path around the air gap.

$$B_{av} = \frac{P\phi}{TDL}.$$

The following factors are to be Considered to choose the specific magnetic loading

- ) Flux density in teeth
- ii) Frequencey
- iii) Voltage

i) Flux density in teeth

If a high value of algap flux density (Baw) is used, the flux density in armature teeth also becomes high. The value of flux density at the root of teeth should not exceeds 2.2 wb/m<sup>2</sup>. Otherwise it may lead to

> increased than loss

→ higher ampere turns requires for passing the flux through teeth leading to increased copper losses and cost of copper. ii) Frequency

The frequency of flux reverse in the armature is given  $f = \frac{pn_s}{2}$ . Higher frequency loads to increased ison losses in the armature core and teeth. So there is a limitation in choosing higher Baw for a machine having higher frequency.

 $N_{g} = \frac{12 \circ f}{P} \implies f = \frac{N_{g} P}{120} = \frac{N_{g}}{60} \times \frac{P}{a} = \frac{Pn_{g}}{2}$ 

Iii) Voltage.

For high Voltage Voachines, Spaced required for insulation is large. Therefore for a given diameter width of the slot increases which inturn decreases the width of the seeth.

If width of the teeth reduces, then area of cross Section of teeth reduces, which increases the flux density of the teeth

$$B_{t} = \frac{\phi_{t}}{A_{t}}$$

Therefore for high voltage machine, low value of Baw is choosen, otherwise the flux density increases beyond the permissible limit.

The typical values of Bas varies from 0.4 to

### Choice of Specific electric loading in DC Machine.

It is defined as the ratio of total ampere Conductor to the armature peripherey at the air gap

$$ac = \frac{I_z \cdot z}{\pi D} \rightarrow 0$$

The following factors are to be Considered, while selecting the Specific electric loading

- i) Temperature rise
- ii) Speed of machine
- iii) Voltage
- iv) Size of the machine
- \*) Armature Reaction

Vi) Commentation

i) Temperature vise

A higher Value of "ac" results in a bright temperature rise of windings. Temperature rise depends on method of cooling and type of enclosure. ii) Speed of machine

In high Speed machines, the ventilation will be batter and more losses Can be dissipated, Hence higher value of ac can be used for higher Speed Machines.

### iii) Voltage

High Voltage machines require large Space for insulation, therefore there is less space for conductors which inturn reduces the area of the Conductors. Due to this reason, a small value of ac' should be used for high voltage machines.

iv) Size of the machine

In large size machines, there is more space for accommadating copper conductors. Therefore high Value of "ac" can be used.

N) Armature Reaction

A higher value of "ac" results in a higher armature mmf. Under loaded condition, this armature mmf affects the field mmf which inturn reduce the value of net flux. To compensate this, field mmf Should be increased. Thus overall cost of copper in the machine will increase.

VI) Commentation

From Equation (D, higher value of ac is achieved by either Using more ampere Conductor (or) with Small diameter.

For small diameter, the deeper slots are Used to accommadate this ampere conductors. Deeper Blots also give higher reactance Voltage. Higher reactance Voltage results in poor Commutation. Hence, higher 'ac' leads to poor commutation.

The typical value of 'ac' varies from 15,000 to 50,000 ampere conductors | metre. A 350 KW, 500V, 4500 rpm, 6 pole DC generator is built with the armature diameter of 0.87 m and Core length of 0.33 m. The lap wound armature has 660 Conductors. Calculate the specific electric and magnetic loading.

Given data

P=350 km, V=500V, N=450 spm, P=6, D=0.81m, L= 0.33 m, lap wound, Z=660

To find

i) specific electric loading (ac)

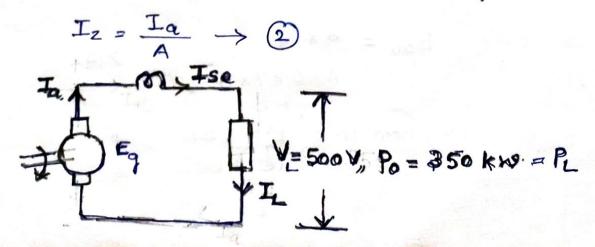
ii) Specific magnetic loading (Bar)

Salution

i) Specific electric loading  

$$ac = I_2, 2$$
  
 $T \cdot D$ 

Iz > Current through the parallel path



 $I_a = I_{se} = I_L \implies 3$ 

From the Circuit diagram

$$P_{L} = V_{L} I_{L}$$

$$350 \times 10^{3} = 500 \times I_{L}$$

$$I_{L} = \frac{350 \times 10^{3}}{500} = 700 \text{ A}$$

From 3

From 2

$$I_{z'} = \frac{700}{6} = 116.6667$$
 A

From D

$$ac = \frac{116.6667 \times 660}{\pi \times 0.87}$$

ac = 28,172.2623 ampère Conductor / metre.

") Specific Magnetic Loading (Bau)  

$$B_{av} = \frac{P\phi}{TDL}$$

$$B_{aw} = \frac{6 \times \phi}{\pi \times 0.87 \times 0.33} \rightarrow 4$$

We know that  $E_{g} = \frac{q \ge N}{60} \times \frac{P}{A}$ 

$$\begin{aligned}
\varphi &= \frac{F_q \times bo \times A}{Z \times N \times P} \\
\varphi &= \frac{F_q \times bo \times 6}{660 \times 450 \times 6} \rightarrow E
\end{aligned}$$
From the Cigcuit diagnam
$$F_q &= IaRa + I_{se}Rse + V_L \\
F_q &\simeq V_L \\
From E = 500 \times 0 \\
\varphi &= 500 \times 60 \times 6 \\
\hline
660 \times 450 \times 6
\end{aligned}$$

From (A)

$$B_{aw} = \frac{6 \times 0.1010}{\pi \times 0.87 \times 0.33}$$
  
$$B_{aw} = 0.6719, wb/m^{2}$$

#### Output Equation

The equation which describe the relationship between main dimension, Specific electric and magnetic loading and speed is known as output Equation.

Power developed by the asymptotic (Pa) in kill  
Pa = generated emf × Armature current x 10<sup>3</sup>.  
Pa = E × Ia × 10<sup>3</sup>, Kull 
$$\rightarrow 0$$
  
Chenorated emf, E =  $\frac{42NP}{60A} \rightarrow 2$   
Where, N  $\rightarrow$  Speed in spm,  $\therefore n = \frac{N}{60}$   
 $N = n \times 60 \rightarrow \otimes$   
where  
 $n \rightarrow$  Speed in sps  
Substitute (3) and (2) in equation (1)  
Pa =  $\frac{42NP}{2} \times I_0 \times 10^3$ 

$$= \phi z(n \times 60) P \times I_q \times 10^3$$

$$P_a = \frac{\varphi z n P}{A} \times Ia \times 10^{-3}, \text{ kne.}$$

Current in each Conductors 
$$(I_2) = \frac{T_a}{A} \rightarrow I_a = I_2 \cdot A \rightarrow \textcircled{0}$$
  
hence  $P_a = \frac{\varphi z_n P}{A} \times I_2 \cdot A \times I_0^{-3} \rightarrow \textcircled{0}$   
Therefore  $P_a = .(P\varphi)(I_2 \cdot Z) \times n \times I_0^{-3} \rightarrow \oiint{0}$   
Therefore  $P_a = .Total magnetic loading \times .Total electrical loading \times .Speed in .The storm is ... No know that
Specific magnetic loading  $Bau = \frac{P\varphi}{TDL}$   
 $P\varphi = Bau TDL$   
 $gpecific electric loading  $ac = Iz \cdot Z$   
 $I_2 \cdot Z = acTD$   
Substitute  $(P\varphi)and \, \oiint{1}(T_2 \cdot Z) \, \ln \, \textcircled{0}$   
 $Pa = (B_{au}TDL)(acTD) \times n \times I_0^{-3}$   
 $Pa = (O D^2 L n Bau ac \times I_0^{-3}), kus$   
 $Pa = Co D^2 L n$   
 $uhere (o = T^2 Bau ac \times I_0^{-3}), kus$   
 $P \Rightarrow Armature Diameter (er) Statos bora, m$   
 $L \Rightarrow hength of core, im$$$ 

ł

Estimate the main dimensions of 9 200 kW, 250 Volts, 6 poles, 1000 spm DC generator. The maximum Value of flux density in the air gap is 0.87 Wb/m<sup>2</sup> and the ampere Conductors per metre length of armature pheriphery are 31000, The ratio of Pole arc to pole pitch = 0.67 and the efficiency is 91 %. Assume that the ratio of length of core to pole pitch = 0.75.

Given data

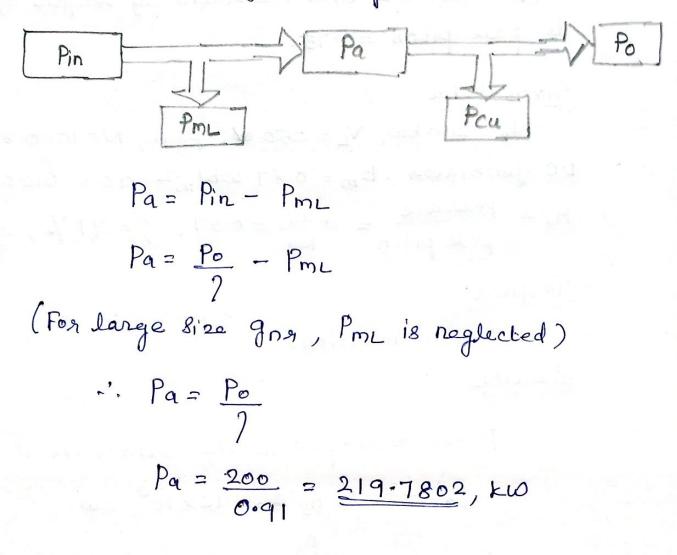
 $P_{0} = 200 \text{ kW}, \ V_{L} = 250 \text{ W}, \ P = 6, \ N = 1000 \text{ spm},$   $pc \text{ generator}, \ B_{m} = 0.87 \text{ wb} / m^{2}, \ qc = 31000,$   $k_{f} = \frac{Pole \text{ are}}{Pole \text{ pitch}} = \frac{Bav}{Bg} = 0.67, \ \gamma = 91^{\circ}/_{0}, \ \frac{L}{2} = 0.75$  To find

Main dimensions Solution

> Power developed in the armature is given by  $Pa = \pi^2 O^2 L ng Bau acx 10^8, kin$   $D^2 L = \frac{Pa}{\pi^2 ng Bau acx 10^3}$  $D^2 L = \frac{Pa}{\pi^2 ng Bau acx 10^3} \rightarrow 0$

Given  $\frac{Bav}{Bg} = 0.67$ Bav = 0.67 x 0.87 Bav = 0.5829,  $wb/m^2$ 

From the power flow diagram



From Equation ()

$$D^{2}L = \frac{219.7802}{\pi^{2}x \ 16.6667 \times 0.5829 \times 31000 \times 10^{3}}$$
$$D^{2}L = 0.0789, m^{3} \rightarrow 2$$

Seperation of D and L from D<sup>2</sup>L, based on the ratio L and is given L = 0.75  $\frac{L}{T} = 0.75$ L=0.752 L= 0.75 × 10 P  $L = 0.75 \times TD$ L= 0.3927, D → 3 From eq 2  $D \times 0.3927 D = 0.0739$  $D^{3} = \frac{0.0739}{0.3927} = 0.1882$ D = 0.5731, m From og (3) L= 0.3927 × 0.5731 L = 0.2250, m

A design is required for a 50 kw 4 pale, 600 rpm dc Shound generator, the full load terminal voltage being 220V. If the maximum gap density is 0.83 cob/m<sup>2</sup> and the armature ampere conductor per metre are 30,000. Calculate Suitable dimensions of gamature core to give a square pole face.

Assume that the full load armature Voltage drop is 3 percent of the rated terminal Voltage. and that the field current is I percent of rated full load current. Ratio of pale arc to pale pitch is 0.67.

### Given data

Po= 50 kne, P=4, N= 600 rpm, DC Shunt generates N\_= 220N, Bg = 0.83 Wb/m², 9c = 30,000 ampere Conductor | metre, Square pole, Iaka = 3% of VL  $I_f = 1^{\circ} /_{0} \text{ of } I_L$ ,  $K_f = \frac{Poleare}{Pole pitch} = \frac{Bav}{Bg} = 0.67$ .

To find

Main Dimensions (D and L)

#### Solution

The power developed in the armature is given by Pa= T<sup>2</sup> D<sup>2</sup> L ng Bau ac x 10°, kw

D<sup>2</sup>L = Pa  

$$\pi^2 \times n_s \times Bav \times ac \times 15^3$$

$$D^{2}L = \frac{P_{q}}{\pi^{2} \times (\frac{600}{60}) \times B_{av} \times 30000 \times 10^{-3}}$$

Given 
$$k_f = \frac{Bav}{Bg} = 0.67$$
  
 $Bav = Bgvx 0.67$   
 $Bav = 0.88 \times 0.67$   
 $Bav = 0.88 \times 0.67$   
 $Bav = 0.5561$ ,  $cob [m^2$ 

all the

From the power flow diagram Pin Pa Po Pn Pa Po Pa Po Pa

$$Pa = Pin + PmL$$

$$Po = \frac{Po}{7} - PmL$$

$$Pa = \frac{Po}{7} (neglect PmL if Po \ge 3.75 km)$$

$$(5 hp)$$

$$Pa = \frac{50}{7}$$

Also the Parise expressed as 
$$P_{a} = E_{g} T_{a} \times 10^{-3}$$
,  $kr_{9} \rightarrow 2^{\circ}$   
 $F_{g} = \frac{\varphi = N}{60} \times \frac{P}{A}$  and also  $E_{g} = V + I_{a} P_{a}$   
 $F_{g} = V + I_{a} P_{a}$   
 $= 320 + (3\% \text{ of } V_{L})$   
 $= 220 + (\frac{3}{100} \times 220)$   
 $= 220 + 6.6$   
 $F_{g} = 226.6$ ,  $V$   
 $I_{a} = T_{L} + I_{f}$ .  $\rightarrow 3$   
 $P_{L} = V_{L} I_{L}$   
 $I_{L} = \frac{P_{L}}{V_{L}} = \frac{50 \times 10^{3}}{220}$   
 $I_{L} = 227 \cdot 2727$ ,  $A$   
 $C_{Hiven} I_{f} = 1\%$  of  $I_{L}$   
 $I_{f} = 0.01 \times 227 \cdot 2127$   
 $I_{f} = \frac{2 \cdot 8727}{4}$ ,  $A$ 

From (3) Ia = 227.2727 + 2.2727

 $T_q = 229.5454$  A

From 2

$$P_{a} = 226.6 \times 229.5454 \times 10^{3}$$
, kw  
 $P_{a} = 52.0150$ , kw

From (

$$D^{2}L = \frac{52.015}{\pi^{2}x (\frac{600}{60}) \times 0.5561x 30000 \times 10^{3}}$$

$$D^{2}L = 2.0316, m^{3} \rightarrow \textcircled{4}$$

Dand L from D'L was seperated by ratio L. Given Square pole face, ie) pale arc = els glass.

$$L = \frac{\text{pole arc}}{\text{pole pitch}} = 0.67$$

$$L = 0.67 \text{ T}$$

$$L = 0.67 \times \text{TD}$$

$$P$$

$$L = 0.67 \times \text{T} \times \text{D}$$

$$4$$

L = 0.526 D> (5) From eq @  $D^2 x \ 0.526D = 0.0316$  $D^3 = \frac{0.0316}{0.526}$  $D^3 = 0.0601$ D = 0.3917, mfrom eq (5) L= 0.526× 0.3917 L = 0.2060, m

<u>Ans</u> D=0.3917, m 2060, m L = 0.2060, m Calculate the diameter and length of asimature for a 75kw, A pale, 1000 r.p.m., 220V Sheintmotors. Criven full load efficiency = 0.83, maximum gap flux deneity = 0.9 cob/m², Specific olectric loading = 80,000 ampere conductors per metre. field form factor = 0.7. Assume that the maximum efficienc, occurs at full load and the field Current is 2.5% of rated current. The pole face is square.

#### Crives data

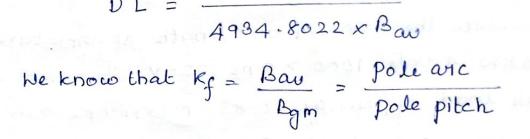
Po=7.5 kw, p=4, N=1000 spm, V=220, Shunt motor Ifull = 0.83, Bgm = 0.9 cob/m<sup>2</sup>, ac = 30,000, kf = 0.7 Maximum efficiency occurs at full load and field current = 2.5% of I\_1; pole face is square. To find

Diameter and length of Armature.

#### Salution

Power developed in the Armature is given by  

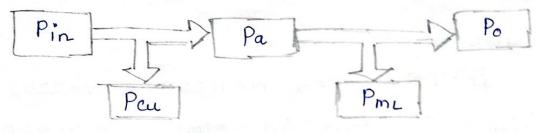
$$Pa = \pi^2 5^2 L n_s Bavac x \overline{10}^3, k_N \rightarrow \mathbb{O}$$
  
 $D^2 L = \frac{Pa}{\pi^2 n_s Bavac x \overline{10}^3}$   
 $D^2 L = \frac{Pa}{\pi^2 n_s Bavac x \overline{10}^3}$ 



Pa

$$Bav = k_f \times Bqm$$
$$= 0.7 \times 0.9$$
$$Bav = 0.63, cob | m^2$$

From Power flow diagram



 $P_a = P_o + P_{mL} \longrightarrow \textcircled{2}$ 

At maximum efficiency occurs at full load. At maximum efficiency.

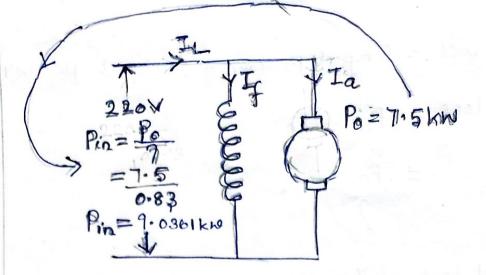
Variable loss = Constantloss, hence Pmc cannot be reglected.

Constant loss = field a loss + Friction and windag. Loss + iron loss

Friction and windage loss + iron loss = Constant loss field cu loss

PmL = Constant loss - field a loss >3

Potal loss = Input power - Output power Total loss = Pin - Po  $=\frac{Po}{7}-Po$ Total loss =  $\left(\frac{7.5}{0.82} - 7.5\right)$ Total loss = 1.5361, KN Also the total loss can be expressed as Total loss = Variable loss + Constant loss At maximenum efficiency, Variable loss = Constant loss ... Potal loss = Constant Loss + Constant Loss Constant loss =  $\frac{105361}{2}$  =  $\frac{105361}{2}$ Constant loss = 0.7681, kne. Field a loss depends on the Current flocos through the Resistance of the field winding. Field cu loss = If Rsh = Vsh If Given, If = 2.5 x IL Field a loss = Vsh x 2.5 x IL  $\rightarrow \bigcirc$ 



$$Pin = V_{L} I_{L}$$

$$I_{L} = \frac{Pin}{V_{L}} = \frac{9*0361 \times 10^{3}}{220}$$

$$I_{L} = 41.0732$$

A

From (1)

Field cy loss =  $220 \times \frac{2.5}{100} \times 41.0732$ = 225.9025, watts

= 0.2259, kw

from (3)

= 0.5422 kw.

From 3

= 8.0422, kw

From (1)

 $D^2L = 8.0422$ 4934.8022 × 0.63  $D^2L = 0.0026 m^3 \rightarrow 6$ Criven pole face is square, polearc = core length L = Corre length = Polearc = 0.7 Pole pitch pole pitch  $\frac{L}{T} = 0.7$ L = 0.77 L= 0.7 AD  $L = \left(\underbrace{O\cdot \mathcal{T} \times \mathcal{T}}_{A}\right) D$ L= 0.5498 D → (b) From Eq (5)  $D^2 \times 0.5498 D = 0.0026$  $D^3 = 0.0026$ 0.5498  $D = \sqrt[3]{0.0047} = 0.1679$ , m From Eq. 6 L= 0.5498 x 0.1679 = 0.0923, m

### Selection of poles in DC machine

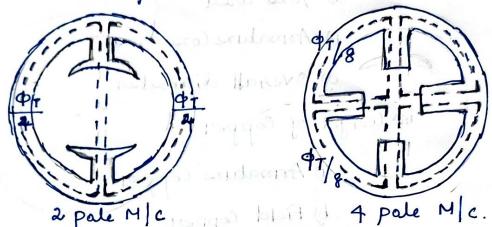
The factors to be considered for selection of poles are i) Frequency ii) height of ison parts a) Yoke area b) Armature Core area c) Overall diameter iii) Weight of Copper 9) Armature Copper b) Field Copper iv) Length of Commutator V) Labour Charges VI) Flash over vii) Distortion of field flux. Level esticite and side i) Frequency The frequency of flux Reversal is given by

 $f = \frac{Pn}{2}$ 

If we choose higher number of poles, the frequency will high. But for DC machine, frequency will be allowed Lies b/w 25 to 50 H2. Otherwise it would give rise to excessive iron loss in grmature core and teeth. 1) Weight of Iron parts

a) Yoke area

The total flux around the airgap is taken as  $P_T$ Consider a 2 pole and 4 pale DC machine



For 2 pole machine, the flux/pole is  $\frac{\Phi_T}{e}$ . But at the Yoke, this flux divides itself into two parts and therefore the yoke has to carry a flux  $\frac{\Phi_T}{e}$ 

For A pole machine, the flux/pole is  $\frac{\Phi_T}{4}$ , But at the yoke, this flux divides itself into two parts and therefore the yoke has to carry a flux of  $\frac{\Phi_T}{2}$ 

From this 2 pole and 4 pole machines, we conclude that, by using greater number of poles, the flux carried by the yoke decreases which inturn decreases the area of cross section of yoke.

If area of Cross Section of yoke decreases means the weight of yoke decreases with Selecting higher number of poles. b) Armature core area

Consider a 2 pole and 4 pole machine Shown in above figure

For 2 pole machine, Flux carried by the Armature is \$+T

For A pale machine, Flux corried by the Armature is  $\frac{\phi_T}{q}$ 

From this we conclude, if we using higher number of poles, Flux in the armature core reduces which inturn reduces the area of cross section of Armature Core.

ie) weight of Armature Core decreases.

But in DC machines. The armature is the rotating past, hence we consider the effect of frequency of flux Reversal.

The equation which relates the poles and frequency is given by

$$f = \frac{Pn}{2}$$

If we choose higher number of poles, the frequency increases which inturn increases the iron losses. But here

Eddy current loss per with higher number of poles. Hysteresis loss tes with choosing higher number of poles. c) overall diameter

If we choose higher number of poles, the armature mmf/pole coill decreases.

If armature mmf/pole is reduced, the number of turns in the field winding required is reduced, which in turn reduces the height of the field winding which allow us to reduce the height of the pole.

If height of the pale reduces the Overall diameter of the machine reduces.

ie) weight of machine reduces with selecting higher no: of poles.

iii) weight of copper a) weight of Armature Copper and all and the Consider a 2 pole and 4 pole DC machine.

4 inactive Copper

Active Copper

N

2 pole M/c 4 pole M/c Active copper > conductor inside the slot and responsible fog emp production

13

NS

FInactive

Copper

Inactive copper -> Conductor which Lies outside the Slot and used for end connection

From this figure, we understood that inactive copper length decreases with increase in poles. That means the weight of copper in Armature is reduced by using higher no: of pales.

#### b) weight of field capper

If we choose higher number of poles, the field monf pole reduces, which allow us to reduce the area of Cross Section of pole which inturn reduces the length of mean turn of field winding.

If length of mean turn of field winding reduces, then weight of field copper reduces and also the power loss in the field winding reduces.

af crizas

# iv) hength of commutator

Length of Commutator is determined by thickness of brush. The thickness of brush depends on Gas current carried by the brush arm. We know that 2 parallel paths meets at each brush arm. The efore

The current carried by the brush asm  $I_b = 2I_z = 2I_a$ 

For Lap winding (A=P), If we choose higher number of poles, the current carried by the brush arm decreases, which allow us to reduce the area thickness of brush.

Reduction in bough thickness which leads to

reduction in length of commutatory.

V) Labour charge

Labour charge increase with choosing higher number of poles

109 geo Hell 43 areises i

Vi) Flash over

We know that No: of brushes = NO: of poles if we choose higher number of poles, then number of brush arm increases, that means Distance b/w adjacent brush arm decreases leads to the possibility of flash over.

Vii) Distortion of field flux

When we select more number of poles, the armature mmf/pole decreases, that means the effect of armature flux over field flux decreases.

Choosing higher number poles reduces the effect Of Armature Reaction

is Carrent Carried by the boots name Is - 2 - 2 - 4-

For Leg (sinding (A=F), If we choole higher purch

poses, the Conserved Conserved by the backets completes

where us to reduce the area of the third

Reduction in Lough thickness which had

# Selection of number of Poles,

Cruiding factors for selection of number of poles.

1) The frequency of flux reversals in the armature Core generally lies between 25 to 50 Hz.

$$f = \frac{Pn}{2}$$
where  $n \rightarrow speed$  in  $rps$ 

$$P \rightarrow No: \circ of poles.$$

2) The value of current parallel path  $(I_z)$  is limited to about 200A. Thus the current per brush 94m  $(I_b)$  should not be more than 400A.

3) The armature mmf should not be excessively large.

Armature mmf/pole ATa = ac c/2

ATa = aCTD

	2 P
outpeut (kw)	Armature mmf/pole
UP 100	5,000 (08) les
100 to 500	5,000 to 7,500
500 to 2000	4,500 to 10,000
OVER 1500	upto 12,500

If more than one poles satisfies the above Conditions, then select large number of poles. This results in reduction in iron and copper. Determine the main dimensions, humber of poles and the length of Ais gap of a book w, 500 v, 900 v.p.m. Dc generator. Assume average gap density as 0.6 Wb/m<sup>2</sup> and ampere conductor / metre as 35000. The ratio of palearic to polepitch is 0.75 and the efficiency is 91%. The following are the dusign constraints : peripheral speed \$ 40 m/s, frequency of flux reversals \$ 50 Hz. Current per brush arm \$ 400 A and the armature mmf / pole \$ 7500 A. The mmf Required for ais gap is 50% of armature mmf / pole and gap contraction factors is 1.15.

Given data

Po=bookio, V=500 V, N=9007pm, DC Generator, Bau= 0.6 cob/m², ac = 25000, <u>Pole arc</u> = 0.75, ]=91%. The mmf Required for aig gap is 50% of armature mmf/pole, gap Contraction factor is 1.15.

Design Constraints

pagolacieb

Va \$40 m/s, Frequency Reversals \$ 50 Hz, Current/Brush arm \$400 A, Armature mmf/pole \$ 7500 A

To find

i) Nain dimensions iii) Length of air gap. ii) Number of poles

Solution i) Main dimensions Power developed in Armature (Pa) in kio. Pa= 7 DL n Bas ac xio kw  $D^2 = \frac{Pa}{\pi^2 n B_{av} ac \times 10^3}$ D'L = Pa  $(\hat{\mathbf{n}})$  $\pi^{2} \times (\frac{900}{60}) \times 0.6 \times 35000 \times 10^{3}$ sing Lener 4 For DC generator Log form students Pa (power developed T (Electrical (Mechanical in Armature) Peu Energy) Energy)  $\gamma = \frac{P_0}{\gamma}$ Therefore Pa = Pin - Pmi  $Pa = \frac{P_0}{2} - PmL$ I raise for any carinader for fuctor J  $Pin = \frac{Po}{7}$ For bookw, Pmc Can be neglected, Pa = Poroport sparts then autor / Bouch and & foo .24  $P_a = \frac{600 \times 10^3}{0.91} = \frac{659.8407}{8407}$  ku of the of the of the Lagizavait aich (1 men to reducit

From Equation ()  

$$D_{L}^{2} = \frac{659.3407}{\pi^{2} \times 15 \times 0.6 \times 35000 \times 10^{3}}$$

$$D_{L}^{2} = 0.2121, m^{3} \rightarrow (2)$$
In DC machine, Separation of D and L from  
DL based on the statio  $\frac{1}{-1}$   
Here  $\frac{1}{-1}$  ratio in not given, so we assume  
the pole is square face.  
Therefore  $\frac{1}{-1} = \frac{Pole arc}{Pole pitch}$   
Criven  $\frac{Pole are}{Pole pitch} = 0.75$ , then  
 $\frac{1}{-1} = 0.757$   
 $L = 0.757$   
 $L = 0.757$   
 $L = 0.757$   
 $\frac{1}{-1} = \frac{1}{-1} = \frac{1}{-1}$ 

a) frequency lies b/w 25 to 50 Hz  $f = \frac{Pn}{2}$   $n = \frac{N}{60} = \frac{900}{60} = 15$ P=a,  $f=\frac{2\times 15}{2}=15$  Hz P=4,  $f = \frac{4 \times 15}{2} = 30 H_2$ P=6,  $f=\frac{6\times 15}{2}=45$  Hz P=8,  $f=\frac{8 \times 15}{2}=60$  Hz P=4 and 6, makes the frequency b/w 25 Hz 60 50 H2. b) current / parallel path 7 200 A Current/parallel path (I2) = Ia > 200 A Assume DC generator is DC Shunt generator IF [Load V= 500V, Po = 600 16% Ia = If + IL There is no way to find If. Hence If can be neglected

uns in Ia = Thing instance

$$P_{L} = V_{L} I_{L}$$

$$I_{L} = \frac{P_{L}}{V_{L}} = \frac{600 \times 10^{3}}{500} = 1200 \text{ A}$$

$$I_{a} = I_{L} = 1200 \text{ A}$$
For lap winding A=P  
when P=4, current/parallel path  $(I_{z}) = \frac{I_{a}}{A} = \frac{1200}{4}$ 

$$= \frac{300 \text{ A}}{6}$$
when P=6, current/parallel path  $(I_{z}) = \frac{I_{a}}{6} = \frac{1200}{6}$ 

$$= 200 \text{ A}$$
For usave winding
$$A=2$$
Current/parallel path  $(I_{z}) = \frac{I_{a}}{A} = \frac{1200}{6} = 600 \text{ A}$ 
Therefore for Lap winding, when p=6, current/parallel path  $(I_{z}) = \frac{I_{a}}{A} = \frac{1200}{6} = 600 \text{ A}$ 
Therefore for Lap winding, when p=6, current/parallel path not exceeds 200 \text{ A}
$$\frac{P=6}{P}$$
From Equation (3)
$$L = 0.75 \text{ AD} P$$

$$L = 0.8927 \text{ D} \Rightarrow (3)$$

Phom Equation (2)  

$$D^{2}(0.8927)D = 0.2121$$

$$D^{3} = \frac{0.2121}{0.3927}$$

$$D^{3} = 0.5401$$

$$D = 0.8144, m$$
From Equation (2)  

$$L = 0.8927 \times 0.8144$$

$$L = 0.3927 \times 0.8144$$

$$L = 0.3198, m$$
iii) Air gap length (ATg)  
ATg = 8,00,000 Bg lg kg  

$$lg = \frac{ATg}{8,00,000 Bg kg} \rightarrow (5)$$
Given  $ATg = 50\%$  of animature mmf [pole  $\Rightarrow$ (8)  
 $armature mmf / Pole = \frac{Tz. Z}{2P} = \frac{ac\pi D}{2P}$ 

$$= \frac{35000 \times \pi \times 0.8144}{2 \times 6}$$
Armature mmf [pole = 7462.3297 AT.

La 6.8100 - + 6

From Equation (6)  
AT<sub>q</sub> = 0.5 × 7462.3297  
AT<sub>q</sub> = 3731.1649 AT  
Given Pale are  
pole pitch = 0.75 and we know that  
Form factors 
$$(kg) = \frac{pale arc}{pale pitch} = \frac{Bau}{bg}$$
  
 $\therefore \frac{Bau}{Bg} = 0.75$   
 $\frac{Bau}{0.75} = Bg$   
 $Bg = \frac{0.6}{0.75} = \frac{0.8}{0.8} \text{ cob}/m^2$   
From Equation (6)  
 $lg = \frac{3731.1649}{8,00,000 \times 0.8 \times 1.15}$   
 $lg = 0.0051 \text{ m} = 5.1 \text{ mm}$   
Check for constraints  
 $i) Va \neq 40 \text{ m/s}$   
 $Va = T D n$ 

.

Va = Tx 0.8144 x15 = <u>38.3777</u>, m/s It lies within the limit i) Frequency of Reversal \$\$50 Hz When P=6, f = 45 Hz Hence it is within the limit. ii) Armature mmf |pole not exceeds 7500 A From Equation (6), Armature mmf |pole = <u>7462.3297</u> A7

It lies within the limit.

i) Main dimension

D = 0.8144, m, L = 0.3198, m

ii) <u>No af poles</u> P=6

iii) Air gap length (lg)

leg = 5.1 mm

Design procedure for field winding of De Shunt  
matchines  
i) Determine the Dimensions of the pale  
flux in the pole, 
$$\Phi_p = C_A d$$
  
where  
 $C_A \Rightarrow$  leakage coefficient  
(Assume a suitable value of  $C_A$ )  
Area of the pole,  $A_p = \frac{\Phi_p}{B_p}$   
where  $B_p \Rightarrow$  flux density lies between 1/2 to  
1/7 wb/ 1/2.  
For Cylindrical pole  
Diameter of pole  $d_p = \int \frac{4A_p}{\pi}$   
For Rectangular pole

Length of pole, Lp=L- (0.01 to 0.015) Net ison length of pale 4pi = 0.9 Lp width of the pole, wp = Ap/Lpi 2) Determine the mean length of field coil A suitable value for depth of winding is assumed by knowing the diameter of the armature core. For cylindrical field coil Mean length, Lmt = T (dp + df) where, dp -> depth of pole df -> depth of field winding. For Rectangular field coil Mean length, Lmt = 2 Lp + 2 Wp + 4 df where, Lp > length of the pole wp > width of the pole of a depth of field winding.

1 Lamp

3) Calculate the voltage across each shunt field coil Voitage across field coil,  $E_f = (0.8 \text{ to } 0.85) \text{ v}$ 4) Calculate the area of cross section of field Conductor. Area of cross section of field Conductor af = PLME ATFE 5) Calculate the diameter of field conductors and Copper Space factor Usually circular shaped conductors are used for field winding  $a_f = \pi d_{fc}^2$  $d_{fc} = 4a_{f}$  $d_{fc} = \int \frac{4a_f}{4a_f}$ dfc > diameter of field Conductor Diameter of field conductor with insulation thickness is expressed as difi

6) Determine the number of turns (Tf) and height of field Coll (hf). They can be determined by Solving the following two equations.

$$2 L_{mt} \mathbf{9}_{f}^{c} (h_{f} + d_{f}) = \frac{E_{f}^{2} a_{f}}{e L_{mt} T_{f}}$$

$$T_{f}^{c} a_{f} = s_{f}^{c} h_{f} d_{f}$$

7) Calculate the resistance of the field Coil and field Current

Resistance of field coil,  $R_f = \frac{T_f P L_{mt}}{a_f}$ Field current  $I_f = \frac{E_f}{R_f}$ 

8) Check the Current density in field Coil

$$S_f = \frac{I_f}{a_f}$$

The current density should not exceed 3.5 A/mm<sup>2</sup>. If it exceeds 3.5 A/mm<sup>2</sup>, then increase of by 5°/. and repeat the steps 5 to 8 untill of is less than 3.5 A/mm<sup>2</sup> 9) Check for desired value of mont

Actual value of  $mmf = I_f T_f$ if the actual mmf is less than the desired value, then increasing the depth of field coinding by 5% and repeat step 2 to step 7, ontill the desired mmfis achived.

10) Check for temperature rise.

Temperature rise =  $\frac{\text{Field cu loss}}{(Q_m)} \xrightarrow{\gamma_f S_f}$ 

 $f = \frac{1}{C}$ , where C is the Cooling Coefficient. =  $\frac{0.14}{1+0.1}$  to  $\frac{0.16}{1+0.1}$ 

> Va → Peripheral Speed of Annature Sf → Surface area of field coil.

Fiel cu loss = If Rf

If temperature rise is within limits then the design values are accepted. Otherwise repeat the design procedure by increasing the depth of field winding by 5%

The allowable temperature rise depends on the class of Insulation.

## Design of Commutator

The commutator is constructed by using hard drawn copper and commutator segments can be Separated by thin mica sheet. The following steps are required to design a commutator.

i) Number of commutator sagments

Namber of commentator segments = No: of the

Number of Coils (c) = =  $\frac{1}{2}$  u Sq

 $u \rightarrow coil Sides / slot$ = 2, 4, 6, 8, ---.  $S_a \rightarrow Primature Slots.$ 

The minimum number of Coils is obtained when Voltage between Commutator segments is not exceed 15 at no load.

... Minimum no of coils =  $\frac{EP}{15}$ 

where E → induced enof P→ No:. of poles.

ii) Commutator diameter (Pc)

 $D_c = (0.6 \text{ to } 0.8) D$ , where D is the Armature diameter.

commutator diameter (Dc) should make the phone pheripheral speed b/w 15 m/s and 30 m/s. . . 'Commentator peripheral speed Vc = TDc ns, m/s Also the commutator diameter statisfies the Commentator Segment pitch & 2. 4 mm. Commutator segment pitch = <u>NDC</u> (Pc) C in) Length of commentator (Lc) The length of the commutation is decided by 9) Space required by the brushes. b) Surface area required to dissipate the heat generated by losses in the Commutator. Length of commutator  $L_c = h_b(w_b + c_b) + c_1 + c_2$ to by Number of boushes per spindle. 10 , width of each brush Cb -> Cleavance b/w brushes = 5 mm G -> clearance allowed for staggering the brushes = 10 mm for small machines = 30 mm for large machines.

 $C_2 \rightarrow$  clearance for allowing the end play  $C_2 = 10$  to 25 mm.

If length of the commentator is Small, then surface area available for heat dissipation is less and that leads to temperature rise of commutator which exceeds the permissible limit:

iv) Commutator losses (Pc)

Losses in the commentator are bruch contact loss and bruch friction loss.

 $P_c = P_{bc} + P_{bf}$ 

9) Brush Contact loss (Pbc)

Pbc = Voltage drop per brush ann x Current per brush

#### 07

Pbc = Total Voltage drop x Armature Current in brushes

b) Brush friction loss (Pbf)

The Brush friction lass depends upon the brush pressure, peripheral Speed of the commentator and the coefficient of friction blue brush and commentator. Brush friction lass Pbf Can be calculated by Pbf = UPb P Ab Vc

where M - ) coefficient of friction = 0.1 to 0.3 Pb > Brush Contact pressure, N/m2 and the surface P > No of poles. AB => total Area. of Brushes., m<sup>2</sup> Vc > peripheral speed of Commutator, m/s tal Tal = 1. American and American ( ) - THATE LOAGE = SHIDDLED X And which manually shall got inter priction and a particular approx 100 istration and the property second proved attract Auto. have shared and induced of 35 5. educe of a global and and A A J A T A

#### Design of Brush

The materials used for Brushes are Carbon, Carbon graphile and metal graphile.

The dimensions of the brushes are its thickness and width.

The thickness of brush is selected such that it Covers 2 to 3 Commutator segments. The area of each brush be taken to corry the Current not more than TO A.

Total brush contact area in a spindle  $(A_b) = \frac{T_b}{\delta_b}$   $T_b \rightarrow \text{Current Carried by each Porush = <math>2 I_a$   $\delta_b \rightarrow \text{Current density} = = 0.1 \text{ A / mm}^2$  $\therefore$  Total brush contact area  $(A_b) = \frac{2 I_a}{A \delta_b} \rightarrow (1)$ 

Let A be the parallel path which depends on type of winding selected.

To limit the Current Carried by each brush to TOA, then minimum number of brushes

Pequired is calculated  
minimum number of brusho = 
$$\frac{Current Carried by}{2ach brush}$$
  
To  
No of brushes should be greater than  
minimum no: of brushes.  
Also the total area of brushes in a spindle  
can be expressed as  
 $A_{b} = no : of brushes \times Area of each brush$   
 $A_{b} = n_{b} \times w_{b} t_{b} \rightarrow (2)$   
From Eq. (1) and (2)  
 $\frac{2 Ta}{A S_{b}} = n_{b} w_{b} t_{b}$   
 $w_{b} = \frac{2 Ta}{A S_{b} n_{b} t_{b}}$   
 $t_{b} \Rightarrow$  thickness of brush  
 $= (2 to 3) \times width of Commutators Segments$ 

= (2603) × Bc

Design a Suitable Commutator for a 350 kv, 600 7pm, 440V, 6 pale DC generator having an annature diameter of 0.75 m. The humber of Coile is 288. Assum Suitable values cohereever recessary.

Criven

P= 350 km, N=600 opm, V=440V, P=6, D=0.75m C= 288.

To find

Dimensions of a commutator.

Solution

mitte a f

) Diameter of Commentator

 $P_{c} = (0.6 \text{ to } 0.8) D.$ 

 $D_{c} = 0.7 \times D = 0.7 \times 0.75$  $D_{c} = 0.525 \text{ m}$ 

Check this Commutator diameter Dc that makes peripheral speed of Commentator <15 mm.

$$V_c = \pi P_c n_s = \pi \times 0.525 \times \frac{600}{60}$$
  
 $V_c = 16.4934 m/s$ 

Here the Peripheral Speed of Commutator is greater than 15 m/s. Hence reduce the diameter of Commutator.



 $0.6 \pm 0.0.7$ ,  $P_c$ ,  $V_c = \pi D_c n_s$ 

0.65  $D_{c} = 0.65 \times D$   $V_{c} = 15 \cdot 3153 m/s$ = 0.65 x 0.75 = 0.4875 m

> 0.64  $D_c = 0.64 \times$   $V_c = 15.0796 m/s$ = 0.48 m

> 0.63  $P_c = 0.63 \times 10^{10} V_c = 14.8440 \text{ m/s}.$ = 0.4725 m

So, out of this when  $D_c = 0.48$ , the pheripheral Speed of commutators is well approximately equal to 15 m/s.

Also check that the selected De Statisfies the Commentator Sogment pitch (Bc) is Z 4mm

Be = TDC' No of Commentatory Segments

 $\beta_c = \frac{\pi \times 0.48}{288} = 0.00$  **51** = **5.2** mm

Here potained  $\beta_c$  is greater than 4 mm, Hence we select  $D_c = 0.48$ , m

i) Length of Commutator  $L_{c} = n_{b}(\omega_{b} + c_{b}) + c_{4} + c_{2}$ (b > cleasance b/w boushes = 5 mm C, I clearance allowed for staggering the brushes = 30 mm for large machines. C2 > Clarance for allowing the end play C2 = 20 mm. Therefore  $L_c = n_b(\omega_b + (5x16^3)) + (30x10^3) + (20x10^3)$  $\geq(f)$ Minimum number 2 \_ Current Carried by Brush (Ib) of Brush Required J = To where  $I_{b} = \frac{2I_{a}}{A} \rightarrow (3)$  $\rightarrow$  (2)Let us assume the DC generator is series Generatog. Ja Ise

 $V_{L2} = 440 V, P_{L2} = P_{0} =$  $V_{L2} = 350 kc$ 

PL= PO= VLIL

$$T_{L} = \frac{P_{0}}{V_{L}} = \frac{350 \times 10^{3}}{440} = \frac{795.45}{45}, A$$

From Figure, Ia = Ise = IL

If Cap winding 18 Selected, then Currrent/parallel path  $I_2 = \frac{T_{e}}{A} = \frac{T_{e}}{P} = \frac{795 \cdot 45}{6} = 132 \cdot 575$ , A If wave winding is selected, then current/parallel path  $I_2 = \frac{T_{e}}{A} = \frac{T_{e}}{2} = \frac{795 \cdot 45}{2} = \frac{397 \cdot 725}{2}$ , A

For Lap winding only, the arovent/parallel path (I2) not exceeds 200 A, Hence lap winding is selected.

For Lap winding (A=P=6)

From (3)  $I_{B} = \frac{2 \times 795.45}{6} = \frac{265.15}{6}$  A

From 2

minimum of bough Required =  $\frac{265.15}{70} = 3.7879$ 24

 $h_b > minimum noumber of brushes Required$  $Therefore <math>[n_b = 6]$  Total Contact area of 2 = no: of brushes x Area of each brush in a Spindle ] = no: of brushes x Area of each brush

$$A_p = n_b \times W_b t_b \rightarrow (4)$$

$$A_{p} = \frac{T_{b}}{8_{b}} = \frac{2 T_{a}}{A 8_{b}} = \frac{2 J_{a}}{A 8_{b}} = \frac{2 J_{a}}{A = p = 6}$$

$$= \frac{2 \times 795.45}{6 \times 0.1 \times 10^{6}}$$

$$A_{p} = \frac{0.00(27)}{6 \times 0.1 \times 10^{6}}$$

$$t_{b} = (2 to 3) \text{ width of Commetators Segments}$$

$$t_{b} = (2 to 3) \beta_{a}$$

$$= \frac{2 \times 5.2 \times 10^{3}}{5 }$$

From Eq (A)

$$W_{b} = \frac{Ap}{n_{b}t_{b}} = \frac{0.0027}{6 \times 0.0156} = 0.0283. m$$

From Eq. ()  

$$L_{c} = 6 (0.0283 + (5 \times 10^{-3})) + (30 \times 10^{-3}) + (20 \times 10^{-3})$$

$$L_{c} = 0.2498, m$$

### Answer

$$D_{c} = 0.48 \text{ m}$$

$$P_{b} = 6$$

$$\omega_{b} = 0.0283 \text{ m}$$

$$t_{b} = 0.0156 \text{ m}$$

$$L_{c} = 0.2498 \text{ m}.$$

Sal a lotting

### **EE 8002 DESIGN OF ELECTRICAL APPARATUS**

## UNIT IV

#### **DESIGN OF INDUCTION MACHINES**

Prepared by Dr . T. Dharma Raj Asso.Prof /EEE

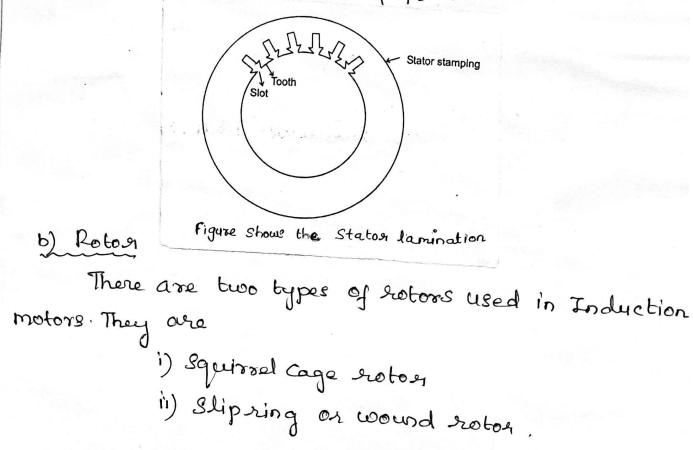
# Construction of 3 & Induction motor

The induction motor consists of two main parts a) staton

b) Roton

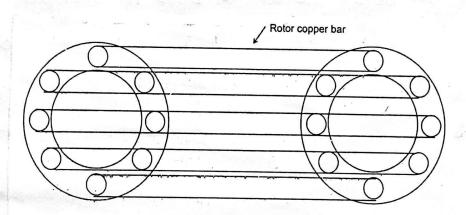
9) Statos

The stator is made up of a number of Stampings with alternate Slot and tooth. Stampings are insulated from each other. Each Stamping is 0.4 to 0.5 mm thick. Number of stampings are Stamped together to build the stator bore. The Stator bore is then fitted in a Casted (or) fabricated Steel frame. The Slots house the three phase winding called Stator winding. It may be connected either in Star or delta. The Stator winding is made for a fixed number of pales.



## i) Squirrel cage robos

This is made up of a Cylindrical Laminated core with slots to carry the rotor Conductors. The rotor Conductors are heavy bens of Copper (or) aluminium short circuited by both ends by end rings. Hence this rotor is also called a short Circuited rotor. The entire rotor resistance is Very Small. External resistance Cannot be Connected in the rotor current. Such motors are extremely rugged in construction. Motors Using Such rotors are Called Squirrel Cage Induction motors.



i) slip ring rotog

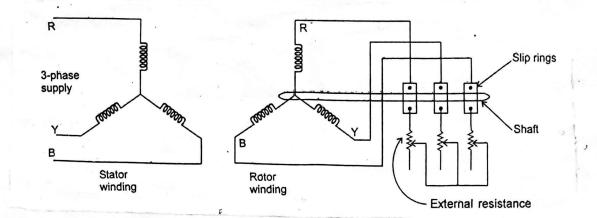


Figure shows the Slip ring (or) wound rotos. In this type of rotos, rotor windings are similiar to the States winding. The rotos winding may be star or delta connected, wound for as many number of pols as the Statos. The three phases are brought out and connected to slip rings mounted on the rotos shaft Variable escternal resistance can be connected in the rotos circuit, with the help of brushes and slip ring arrangements. By Varying the escternal resistance in the rotos Circuit, the motor speed and torque can be controlled. This motor using this type of rotos is called slip ring (or) wound rotos Induction motor.

Comparision of squirrel cage and slip-ring Induction motor

SINO	Squinnel Cage Induction motor	Slip ring Induction Motog
	Low Starting torque	Much Higher Starting torque
<b>२</b> .	No slip rings, brushes gears etc	Extra slip rings, brush gears, etc.
3.	ſ	Comparatively lower efficiency.

Specific Electric and Magnetic Loading of IM

Specific Electric Loading (ac) Ne know that  $ac = \frac{I_Z \cdot Z}{TD} \rightarrow O$ The following factor are to be considered for Selecting the Specific electric Loading a) copper loss and Temperature rise. b) Voltage c) Size and Cost of the machine. d) Overload Capacity

# 2) copper loss and Temperature vie

From Equation (D), the higher the value of ac is Choosen, which increases the current flow through the conductor and number of Conductors

Increase in current in the Conductor, which increases the temperature rise of the machine.

Increase in Conductor, increases the Copper loss, which reduces the efficiency.

b) Voltage.

For high Voltage machine, thickness of Insulation around the Conductor is less, Therefore area of Cross of the Conductor is less.

If large of ac is choosen, will damage the Conductor.

, c) Size and cost of the machine.

From equation (B), specific electoric loading is inversely proportional to the inner diameter of the stator.

If higher Value of ac is choosen, results the Small value of Diameter of the prachine is required Therefore if diameter reduces, the size of the prachine reduces, which inturn reduces the Cost of the machine.

## a) Overload Capacity

If higher value of ac it choosen, the number of conductor used will increases, which inturn increases the leakage reactance.

If leakage reactance increases, the short cut

Reduction in Shoot Cigauit Cyovent reduces the diameter of the circle diagram, which inturn reduces the maximum output. Hence overload Capacity of the motor reduces. From the point of view of

copper loss and temperature rise, 2 allows lower value voltage and overload apacity ) of ac to choose.

Size and cost of the machine of allows higher value of ac to choose.

Because of the above factors, the specific electric loading of an Induction motor usually lies between 25000 to 40000 ampere conductor pretre.

Specific Haghetic loading (Baw)  
We know that 
$$B_{aw} = \frac{P\Phi}{TDL} \rightarrow \textcircled{D}$$

The following factors are to be considered while Selecting the Specific magnetic loading 9) Power factor b) Iron loss c) Overload Capacity

9) Power factor

From Equation @, If higher Value of Bao is Choosen, means flux/pole is high To make flux/pole to high, the induction motor draws more magnetising current

.". Increase in magnetising current results in poor power factor.

b) Iron Loss

There are two types of 1200 losses. Eddy current loss & Bc<sup>2</sup>f<sup>2</sup> Hysterisis loss & Bc<sup>1.6</sup>f

From the above Equations, If we select higher Nalue of Bay, increases the iron loss, which reduces the efficiency.

c) Overload Capacity

From Equation (2), if we select the higher Value of Baw, flux/pole will be large.

If flux pole increases, which inturn increases the emp induced in the winding, because

Esph = 4.44 f & Tsph kus, Voits.

Le as to maintain the constant voltage, we should decrease the turns in the winding, which reduces the larkage reactance.

If leakage reactance reduces, which results in increase in Short circuit current. If Short circuit Current increases, the diameter of the Circle diagram increases which in turn increases the mascimum output of the Induction motor.

From the point of Views of

powerfactor and Iron loss -> low value of Bau is Choosen

overload capacity > high value of Bau is Choosen.

Therefore the Specific magnetic loading lies between 0.3 to 0.6 cob/m<sup>2</sup>. Output Equation of 30 Induction Motor.

Input in kVA = 3× Esph × Isph × 10, kVA >0 Where Esph > states induced emf/phase Isph > States current / phase.

We know that

Synchronous speed 
$$N_g = \frac{120f}{p}$$
, spm

$$f = \frac{N_S P}{120} = \frac{N_S P}{2 \times 60}$$

$$f = \frac{Pn_s}{2}$$
 where  $P \rightarrow Nomber of Poly$   
 $n_s \rightarrow Synchronis Speed in Sps.$ 

From Equation 2 Input Q in LVA = 3x4.44 Pris q Tsph Kwx Isph X10, KVA Input Q in KVA = 6.66 (Pp) ng Tsph kw × Isph ×10, kvA

We know that

$$T_{s} = \frac{z_{s}}{2}$$

$$T_{sph} = \frac{z_{s}}{2x_{3}}$$

$$T_{sph} = \frac{z_{s}}{6}$$

From Equation (3) Input Q in KVA = 6.66 (PQ) ng Zs Kw x Isph x 10, KVA Let Isph be the States current per phase, if there is only one parallel path, then Isph = Iz From Equation (7)  $Triput a in kVA = 6.66 (P$) n_s \frac{Z_s}{L} k_w \times I_Z \times 10, kVA$ Input Q in  $kVA = 1.11(Pq) n_s k_w(I_z, Z_s) \times 10^3$ , kVAhe know that Specific magnetic loading, (Baw) = PA TOL PØ = Bau TOL Specific Electric Loading, (ac)= Iz. Zs  $J_z \cdot Z_s = ac \pi o$ 

From Equation (5) Input a in KVA = 1.11 (BauTOL) NS KO (aCTO) XIO, KVA  $[Input @ in kvn = 1.11 \pi^2 p^2 L ng Bau ac kw x 10^3, kvn]$ 

Note

The output power of Induction motor can be expressed as in kw or hp, then

08

Input & in EVB = Hpx 0:7846 (in Ew) 7 x coso Find the main dimensions of 20 HP, 34, 400v, 50 Hz, 2010 rpm Squirrel Cage IN having 7 of 80% and Pf of 0.9, average flux density is 0.5 Tesla, Specific electric loading is 25000 ac/m, 1f the subtar pheripheral speed is 20 m/s. Assume Kio= 0.955. Cirrendata

 $P_{0=20HP} = 20 \times 0.74b = 14.92 \text{ kW}$ , phase = 3, N = 400 V, f = 50 Hz, N = 2810 Jpm,  $\mathcal{I} = 88^{\circ}/_{\circ}$ , Pf = 0.9, Bav =  $0.5 \text{ testa} = 0.5 \text{ cob}/m^{2}$ , 9c = 25,000, Na = 20 m/s, Kw = 0.955.

To find

Main dimensions.

Solutions

ng = Ng, where Ng -> Synchronous speed in rpm.

The Synchronous speed nearer to motor speed 2810 is 3000 rpm.

$$h_s = \frac{13000}{60} = \frac{50}{7}, rps$$

Substitute as and no value in equation O, we get

$$D^{2}L = 18.8384$$
  
I·II\* $\pi^{2}x 50 \times 0.5 \times 25000 \times 0.955 \times 10^{3}$ 

$$D^{2}L \rightarrow \underbrace{0.0029}{m} \xrightarrow{3} \longrightarrow \textcircled{2}$$

Griven  $V_a = 20 \text{ m/s}$   $\pi D_{ng} = 20$   $\pi \times D \times 50 = 20$   $D = \frac{20}{50 \times \pi} = 0.1274, \text{m}$ From (2)  $(0.1274)^2 L = 0.0029$  L = 0.0029 L = 0.0029 = 0.1787, m $(0.1274)^2$ 

$$D = 0.1274$$
, m  
L= 0.1787, m

Estimate the Staton Corre dimensions, no of Staton Slots and no of Staton Cond Slot for a looke, 9300V, 50 Hz, 12 pole, Star Connected Slipring Induction motor. Assume average gap density = 0.4  $\omega b/m^2$ , ac = 25,000 A/m,  $\gamma = 0.9$ ,  $\cos q = 0.9$ , and  $\omega inding$ factor = 0.96. Choose main dimensions to give best power factor. The Slot Loading Should not exceed 500 A.

Given daba

Po=100 kW,  $V_{L} = E_{L} = 3300 \text{ V}$ , f = 50, p = 12, 3tatogwinding - 3tar connected,  $Bau = 0.4 \text{ cob}/m^{2}$ , ac = 25,00 ac = 25000,  $\gamma = 0.9$ ,  $Pfz \cos \phi = 0.9$ , kw = 0.96Slot loading =  $T_{Z}Z = T_{Sph}Z_{S} \leq 500 \text{ A}$ .

To find

Core dimensions, no: of Stator Elots, number of Stator cond 316t

Solution i) Core dimensions Input Q in KNA = 1.11  $\overline{\Lambda}^2 D^2 L$  ng Bayac Kiyxio, KNA  $D^2 L = \frac{1}{1.11 \times \overline{\Lambda}^2 \times \overline{\Omega}_5 \times$ 

Input Q in KVA = Output power in KN ) × Coso  $= \frac{100}{0.9 \times 0.9} = \frac{123.4568}{568} k \sqrt{3}$  $n_s = \frac{N_s}{60} = \frac{120 f}{P \times 60} = \frac{120 \times 50}{12 \times 60} = \frac{8.3333}{3333}, \pi ps$ Substitute the Q and ng Value in eq B  $D^{2}L = 123.4568$ 1.11× A2× 8-3333×0.4 × 25000×0.96×10  $D^2L = 0.1409, m^3 \rightarrow (2)$ Given for best power fower factor choose the main dimensions. Therefore for Best power factors T= V0.181 2- 0.18L  $\left(\frac{\pi D}{D}\right)^2 = 0.18L$ 

$$\frac{\overline{\Lambda^2 \rho^2}}{P^2} = 0.18 L$$

$$D^{2} = \frac{0.18 L P^{2}}{\pi^{2}}$$

$$D^{2} = \frac{0.18 L \times 12^{2}}{\pi^{2}}$$

$$D^{2} = 2.6262 L \rightarrow (3)$$
From Equation (2)
$$2.6262 L \times L = 0.1409$$

$$L^{-} = \frac{0.1409}{2.6262}$$
$$L = \sqrt{0.0537}$$
$$L = 0.2316$$
m

From

The equation (3)  

$$D^2 = 2.6262 \times 0.2316$$
  
 $D^2 = 0.6083$   
 $D = \sqrt{0.6083}$   
 $D = 0.7799$ , m.

	m - slots/pole/phase	2
m	Ss= mxpalex phase	Yss= <u>TD</u> Ss
a N	72	34 mm
3	108	22.7 mm
ተ	144	17 mm
5	180	13.6 mm
There	Capacity motog, Slot g 5 mm to 25 mm. fore we can select [33 the Advantage of Mec	pitch allowed
There, ue to t Select	fore we can select $[3_{s}]$ the Advantage of the $[3_{s} = 108]$	pitch allowed
There, ue to t Select	fore we can select $[3_{s}]$ the Advantage of the $[3_{s} = 108]$ log turns [phase	= 108 or 144 hanical Strengt
There, ue to t Select iii) Sta	fore we can select $[3_{s}]$ the Advantage of the $[S_{s} = 108]$ log turns [phase Esph = 4:44 f & Tsph	= 108 or 144 hanical Strengt
There, ue to t Select iii) Sta	fore we can select $[3_{s}]$ the Advantage of the $[3_{s} = 108]$	= 108 or 144 hanical Strengt kw, Volts.

Criven States winding is between connected  

$$E_{L} = \sqrt{3} E_{Sph}$$
  
 $E_{Sph} = \frac{3800}{\sqrt{3}} = \frac{1905 \cdot 2559}{70L}$ , Voits  
We know that  $B_{av} = \frac{P\Phi}{TDL}$   
 $\Phi = \frac{B_{av} TDL}{P}$   
 $\Phi = \frac{0.4 \times T \times 0.7199 \times 0.2316}{12}$   
 $\Phi = \frac{0.0189}{9}$ , wb  
Substitute Esph and  $\Phi$  value in Eq. (P)  
 $T_{Sph} = \frac{1905 \cdot 2559}{4.44 \times 50 \times 0.0189 \times 0.96}$   
 $T_{Sph} = 473 \cdot 0067$   
 $T_{Sph} = 473$   
States Conductors (Zs) =  $6 \times 7_{Sph}$   
 $Z_g = 2838$   
States Conductors [Slot =  $\frac{2838}{108} = 26 \cdot 2778$ 

States and slot 
$$(z_3) = 26$$
  
For high capacity motor, double layer winding  
18 used, Hence conductor/slot should be an even  
Integer.  
 $Z_{S}(new) = Cond/slot \times Slot$   
 $Z_{S}(new) = 26 \times 108$   
 $\overline{Z_{S}(new)} = 2808$   
 $\overline{T_{Sph}(new)} = 2808$   
 $\overline{T_{Sph}(new)} = 2808$   
 $\overline{T_{Sph}(new)} = 468$   
Check for slot loading  
Slot loading =  $I_{Z}$ ,  $Z = I_{Sph} \times Z_{SS}$ ,  $A$   
 $Z_{SS} \rightarrow 8tater anductor/slot} \rightarrow (5)$   
Input  $Q$  in  $kvA = 8$  Esph  $T_{Sph} \times 10^{3}$ ,  $kvn$   
 $\overline{T_{Sph}} = \frac{T_{nput} Q \ln kvn}{3 \times Esph \times 10^{3}}$   
 $\overline{T_{Sph}} = \frac{123 \cdot 4568}{3 \times 1905 \cdot 2559 \times 10^{3}}$   
 $\overline{T_{Sph}} = 21.5993$  A

From Equation (5)

Slot loading =  $21.5993 \times 26 = 561.5828$ , A when  $S_S = 108$ , the Slot loading exceeds the limit 500 A,

Therefore we select 
$$S_{s} = 144$$
  
Statog Conductor / slot = 2838  
144  
Statog Conductor / slot = 19.7083  
Statog Conductor / slot = 19.7083

$$Z_{s} = 20 \times 144$$
  
 $Z_{s} = 2880$ 

Stator turns | phase Tsph =  $\frac{Z_s}{6} = \frac{2880}{6}$ (Tsph = 480) Check for slot loading

Slot loading = 21.5993 × 20 = 431.9860, A

when  $g_{g} = 144$ , the slot loading lies within the limit

Answers

D = 0.7799, m L = 0.2316, m  $S_s = 144$ Conductor / 8lot = 20 Total Stator Conductors = 2880 Stator \$urns | phase = 480 Determine the main dimensions, Do: of Ventillating ducts, no: of Stator slots and the no: of turns [phase, area of Cross Section of Statar Conductors of a 37kus, AOOV, 39, 4 pole and 50 Hz Squirrel Cage Induction motor to be Started by a Star-delta Starter. Assume average of lux density in the gap = 0.45 wb/nf, power factor = 0.84. Machine rated at 3.7 kw, 4 pole are soll at a least price and therefore Choose the main dimensions. Assume winding factor = 0.955, Stacking factor = 0.9, Current density = 5 A [mm<sup>2</sup>. Given data

 $P_0 = 3.7 \text{ kw}, E_{L} = V_L = 400 \text{ V}, P = 6, Phase = 3, f = 50 \text{ Hz}, Stain-delta Starter, Baw = 0.45 wb/m², Pf = 0.84, kw = 0.955, Sf = 0.9, 8 = 5 A/mm², efficiency = 85% = 0.85, ac = 23,000$ 

### Pofind

i) main dimensions ii) not of ventillating ducts, iii) Not of Staton Slots iv) not of Staton turns [phase v) area of Cross Section of Staton Conductors.

Solution

i) Main Dimensions

Input a in KVA= 1.110<sup>2</sup> Lng Bauac Kwx10<sup>3</sup>, KVA

L= 1.5 × 0.7854 × D  
L=1.1781 D 
$$\longrightarrow$$
 (3)  
From Eq. (2)  
 $D^2 \times 1.1781 D = 0.0019$   
 $D^3 = \frac{0.0019}{1.1781} = 0.0016$   
 $D = 0.1173$ , m  
From Eq. (3)  
L= 1.1781 × 0.1173  
L= 0.1382, m.  
) Number of ventillating ducks  
when L > 0.1 m of 100 mm, ve

ir)

are provided to Cool the motor.

Given 
$$S_f = 0.9$$
  
 $\therefore S_f = \frac{L_i}{L}$   
 $Li = S_f \times L = 0.9 \times 0.1382$   
 $= 0.1244 \text{ m}$   
If  $Li = 0.1249 \text{ m}$ , It is possible to use 2

ducts with 10 mm wide.

Ventillating Ducts = 2 of 10 mm wide

m	Sg Ss=Sim × pole × phase	$y_{3S} = T_{1}D_{1}S_{3}$
2	24	15 . A mm
3	36	10.21 mm
4	48	7.79: mm
5	60	6-135 mm

in) No. of Statog Blots

for low capacity machine, the Stator slot Pitch Yss is less than 15 mm

iv) No: of terms phase

Fsph= A.44f & Tsph kw, Volts

$$T_{sph} = \frac{E_{sph}}{4.44 \text{ f } 4 \text{ kw}}$$

$$T_{sph} = \frac{E_{sph}}{4.44 \times 50 \times 4 \times 0.955} \rightarrow \textcircled{4}$$

Given, Motor is started Starts by Star- Delta Starter (ie) running time the stator winding is Delta connected.

Ne know that Bau = PO TOL

 $\phi = 0.45 \times \pi \times 0.1173 \times 0.1382$  4  $\phi = 0.0057 \text{ wb}$ 

Substitute Esph and d value in eq. (1)  $T_{sph} = \frac{400}{4.44 \times 50 \times 0.0057 \times 0.955}$ 

Staton Conductors (Zs) = 6× Tsph = 1986

Statog Conductors | Slot =  $\frac{1986}{36} = 55.1668$ Statog Conductog / slot = 55

States Conductors 
$$Z_s$$
 (new) =  $Cond/glot \times Slot$   

$$Z_s = 55 \times 36 = 1980$$
States turns/phase(Tsph(new)) =  $\frac{Z_s}{6} = \frac{1980}{6}$ 
(Tsph (new) = 380  
v) Area of Cross Section of States Conductors  
Current density  $\delta = \frac{Tsph}{a_s}$   
 $a_g = \frac{Tsph}{\delta} = \frac{Tsph}{5} \rightarrow \odot$   
We know that  
Input Q in kun =  $\Im Esph Tsph \times 10^3$ , kun  
 $Tsph = \frac{Tnput Q in kun}{3 \times Esph X 10^3}$   
 $= \frac{5 \cdot 1821}{3 \times 400 \times 10^3}$   
Tsph =  $4 \cdot 3184 \text{ A}$   
From Equation (6)  
 $\left[a_g = 0.8637\right], mm^2$ 

Answers

D = 0.1173 m L = 0.1382 m  $P_d = 2$   $S_g = 36$   $T_{gph} = 330$   $Z_{ge} = 1980$  Statog Conductog / slot = 55Area of Statog Conductor,  $a_g = 0.8637$ , mm<sup>2</sup>.

A 15kN, 440V, 4 pole, 50 Hz, 30 Induction motor is built with a Stator bore = 0.25m and core length of O.16m. The Specific electric loading is 23,000 amp. Cond/m. Using the data of this machine, determine the core dimensions, no: of Stator Slot and no: of Stator Conductors for a 11 ku, 460V, 6 pole, 50 Hz motor. Assume a full load efficiency of 84% and powerfactor of 0.88 for each machine. The winding factor is 0.955. <u>Criven</u> data

Motos.1.	Motor 2
Po= 15 ku	Po=11kh
V= 440 V	V = 460 V
P= 4	p=6
$f = 50 H_2$	$f = 50 H_2$
D = 0.25 m L = 0.16 m ac = 23000 $j = 84^{\circ}/_{\circ} = 0.84$ Pf = 0.87 $k_{w} = 0.955$	2 = 84 % = 0.84 Pf = 0.8 $k_{w} = 0.955$

Tofind

main dimensions, no: of Staton slots and no of staton conductors for a 11 kW motor. Solution The output Equation of 30 Induction motor -2 is Input a in KNA= 1.11 A2 D2 L ng Bawac Keo x10, KNA D<sup>2</sup>L = Input Q in KUA 111 J Dg Bay ac ku x 10 3 D'L = Input ar in Kur 1.11 12 ng Bay 23,000 × 0.955× 10  $\rightarrow G$ Input Qu'in KUA = Output power in kus 7 × caso = <u>11</u> 0.84 × 0.82 Input Q in KVB = 15.9628, KVB  $\frac{n_{g}}{60} = \frac{N_{g}}{60} = \frac{120f}{7 k_{0}} = \frac{120 \times 50}{6 \times 60}$ ng = 16.6667, 8ps

To find Bau, use the data of 15 kW Induction motor Input Q in KVA = 1.11 J2 D2 L ng Baw ac Kwx10 KVF Bau = Input & in kun 1.11 72 D2L ng 96 Kwx 103 Bav = Input a in KUA 1.11  $\pi^2 \times (0.25)^2 \times (0.16) \times n_g \times 23,000, \times 0.955$ ×103  $\rightarrow$  (7) Input & in kun = output power in kw 7 × cos ¢ Input Quin KUB = 15 0.84 × 0.82 Input & in KVA = 21.777, KVA  $n_{g=} \frac{N_{g}}{60} = \frac{120 f}{P \times 60} = \frac{120 \times 50}{4 \times 60} = \frac{25}{25} \times 795$ From Eq 2 Baw = 21.777  $1.11 \times \pi^{2} \times (0.25)^{2} \times 0.16 \times 25 \times 23,000 \times 0.955 \times 10^{3}$  $B_{av} = 0.3620, wb/m^2$ 

Sub  $Bau = 0.3620 \text{ wb}/m^2$ ,  $n_s = 16.6667 \text{ rps}$ , and Input & in kvA = 15.9628 in eq. (D), we get

$$D^{2}L = \frac{15.9628}{1.11 \, \pi^{2} \times 16.6667 \times 0.3620 \times 23,000 \times 0.955 \times 10^{3}}$$

$$D^{2}L = 0.0110, m^{3} \longrightarrow (8)$$

As per given data, the seperation of Dand L from D<sup>2</sup>L 18 based on the reltio <u>L</u> -. To find <u>L</u> gatio, use the data of 15 kW IM

$$\frac{L}{\tau} = \frac{\frac{1}{\sqrt{\pi}}}{\frac{1}{\sqrt{\pi}}} = \frac{0.16}{(\sqrt{\pi}\times0.25)} = \frac{0.8149}{(\sqrt{\pi}\times0.25)}$$

$$\frac{L}{\tau} = 0.8149$$

FOG 11 KN, Induction motog

 $\frac{L}{Z} = 0.8149, \text{ then } L = 0.8149 \text{ T}$   $L = 0.8149 \xrightarrow{\overline{A}} \frac{\overline{P}}{P}$   $L = 0.8149 \times \frac{3.14}{6}$   $L = 0.4265 \text{ D} \longrightarrow \text{ }$ 

From Equation (3)  

$$D_x^2 0.4265D = 0.0110$$
  
 $D_y^3 = \frac{0.011}{0.4265} = \frac{0.0258}{0.4265}$   
 $D = \sqrt[3]{0.0258} = \frac{0.2955}{0.2955}$ , m  
From Equation (4)  
 $L = 0.4265 \times 0.2955$   
 $L = 0.1260$ , m

n	Sz= m x pole x phase	Yes = TD Ss
	$= m \times 6 \times 3$	S
2	36	25.74 mm
3	54	17.16 mm
4	72	12.87 mm

lies

when  $S_S = 54$ , the slot pitch blue the range 15 mm to 25 mm, Hence we select Sg=54

Statog Conductors

$$Tsph = \frac{Esph}{4.44 f 4 kw}$$

$$T_{sph} = \frac{E_{sph}}{4.44 \times 50 \times 4 \times 0.955} \rightarrow 6$$

Let us assume that statog winding is Pella connected

$$E_{sph} = E_{L} = 460 V$$

We know that  $Bav = \frac{P\phi}{TDL}$ 

$$q = \frac{Bav TDL}{P}$$

 $q = \frac{0.3620 \times \pi \times 0.2955 \times 0.126}{6}$ 

$$p = 0.0071, cob$$

Sub Esph and & values in Eq. (5), we get

$$T_{sph} = \frac{460}{4.44 \times 50 \times 0.0071 \times 0.955}$$

We know that

$$T_{sph} = \frac{Z_s}{6}$$

$$T_{s} = \frac{z_{s}}{2}$$

$$T_{sph} = \frac{z_{s}}{2 \times 3}$$

Assewer

D = 0.2955, m L = 0.1260, m Statog Blots  $(S_s) = 54$ Statog turns/phase (Fsph) = 306 Statog Conductors  $(Z_s) = 1836$ Statog Conductor (Slot = 34. Design of Squirrel Cage Robor

It involes the following i) Diameter and length of Rotor i) Number of Rotan Slots iii) Number of Rotor base iv) Rotor bar current and area of rotor conductors N) End ring current and area of end rings. i) Diameter and length of Rotor lg 5 K Lg D= Do + lg + lg  $D_{g} = D - 2lg$ where lg = 0.2 + 2 JDL, mm Dand L are in m Length of rotog = length of Statog Ly=L

# ii) Number of Rotar Slots

with Certain Combination of Staton and noton slots Machine may be

a) referse to start (cogging) b) run at Yyth speed of Synchronous speed (crawling)

c) produce noise and Vibration.

To avoid this,  $(S_s - S_r)$  should not be equal to  $S_s - S_r \neq 0, \pm 1, \pm 2, \pm P, \pm 2P, \pm 3P, \pm (P \pm 1),$  $\pm (P \pm 2).$ 

- 11) Number of robor bars

Number of rotor bars = Nor of rotor slots.

iv) Roton bar current and area of Roton bar Roton bas Current

bet us assume that Rotan mmf is 85% of Staton mmf

Sr x Ib = 0.85 x 3 x Tsph x Isph

Area of Rotor bar

Area of Roton ban is proportional to the Current flows through the ba

$$\delta_{b} = \frac{I_{b}}{a_{b}}$$

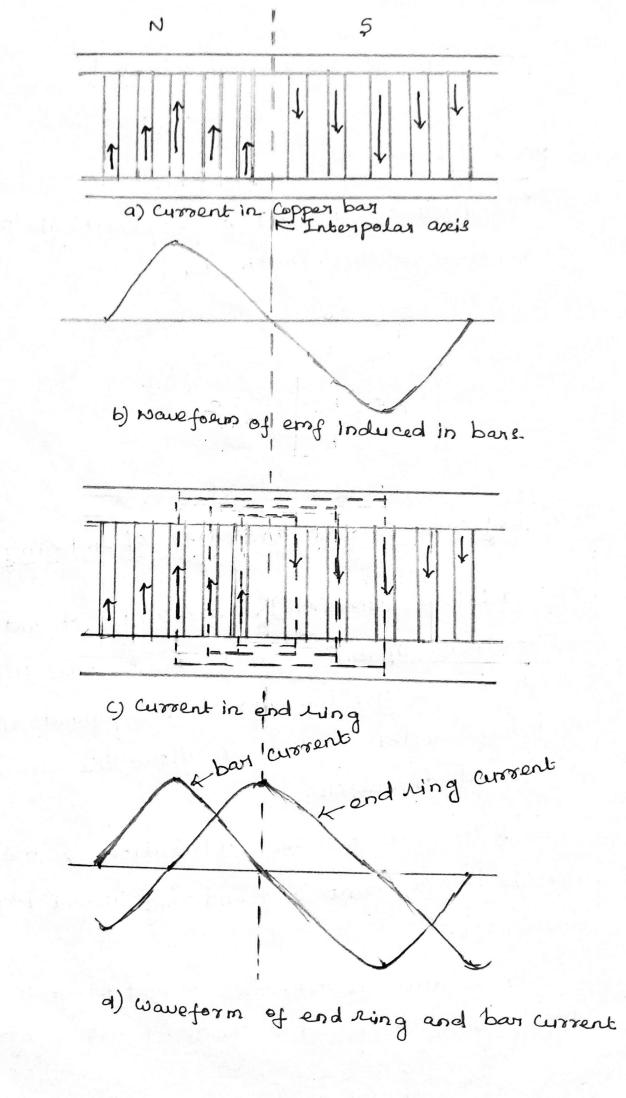
$$\alpha_{b} = \frac{I_{b}}{s_{b}}, mm^{2}$$

v) End ring arrent and Area of end ring

Let us consider the bars under N-and 3 alternate North and South poles. The bars under North pole allows the Current in opward direction and bars under South pole allows the Current in downward direction.

If the flux and emp distribution is Sinusoidal, then the bar current and end ring current is also Sinusoidal.

Normally, the resistance of end rings is negligible as compared to the resistance of bars.



Let us consider the bags under one pole pitch and assume that current in all bass are maximum. Therefore maximum Value of the Current in end ring 18 given by

$$\frac{C(max)}{2p} \times I_{b}(max) \rightarrow (0)$$

Practically all bar currents are not mascimen at same time. Therefore we take the average Value of bar current

$$J_{e}(max) = \frac{s_{T}}{2p} J_{b}(awq) \rightarrow (1)$$

$$J_{b}(awq) = \frac{2}{\pi} J_{b}(maxc)$$

$$Fquation (1) becomes$$

$$I_{e}(maxe) = \frac{s_{T}}{2p} \times \frac{2}{\pi} J_{b}(maxc).$$

The R.m.s value of endring current is griven by  $\sqrt{2I}e = \frac{S_r}{2p} \times \frac{2}{T} \sqrt{2I_b}$ 

$$I_{e} = \frac{S_{\pi} I_{b}}{\pi P}$$

Ne know that I kms = Imanc V2 Imasc = I2. I

Area of the endring  

$$\delta_e = \frac{I_e}{a_e}$$
  
 $a_e = \frac{I_e}{\delta_e}$ 

, cohere de + current density

A II KW, 34, 6 pole, 50 Hz, 220V, Star Connected Induction Motor has 54 Stator Slot, each slot containing 9 Conductors. Calculate the values of bar and end ring Currents, the number of rotor bar is 64. The machine has 7 = 0.86 and  $\cos \phi = 0.85$ , the rotor mmg may be assumed as  $85^{\circ}/_{\circ}$  Stator mmg. Also find the area of bar and end ring Section, if the current density is  $5 \text{ A}[\text{mm}^2]$ .

Given daba

Po=11kW, P=6, 3 phase Y connected Induction motor,  $S_s = 54$ , f=50,  $V_{L} = 220 V_{,z} E_L$ , Cond/Slot = 9  $S_{7} = 64$ ,  $\gamma = 0.86$ , Cosp = 0.86, Rotor mmf = 85% = 0.85of Statos mmf,  $\delta_e = \delta_b = 5A/mm^2$ .

To find

Area of bar and end ring Section.

Solution

1) Area of bar

B<sub>b</sub> = <u>L</u>b Q<sub>1</sub>

ay = <u>Io</u>

$$a_b = \frac{I_b}{5} \rightarrow 0$$

Given rotor mmf = 0.85 × Stator mmf

Motor conductors = rotar slots

$$\frac{\sqrt[3]{2}}{2} \times I_{b} = 0.85 \times 3 \times T_{sph} \times I_{sph}$$

Staton Turns/phase Taph = 
$$\frac{75}{6} = \frac{486}{6} = 81$$
  
Taph = 81

We know that

C

Input Q in 
$$kVR = \frac{Output power in kN}{J \times \cos \phi}$$
  

$$= \frac{11}{0.86 \times 0.85}$$
Input Q in  $kUR = \frac{15 \cdot 0.479}{0.86 \times 0.85}$ , KVR  
Given States winding is star connected, therefore  
 $Fsph = \frac{E_L}{V_3}$ ,  
 $Fsph = \frac{820}{V_3} = \frac{155 \cdot 5635}{5.5635}$ , Volt  
From Equation (3)  
 $Tsph = \frac{15 \cdot 0.479}{3 \times 155 \cdot 5635 \times 10^3}$   
 $Tgph = \frac{32 \cdot 2.439}{3 \times 155 \cdot 5635 \times 10^3}$   
 $T_3ph = \frac{32 \cdot 2.439}{5}$ , A  
Substitute Tsph and Isph Value aquation (2)  
 $\frac{64}{8} \times T_b = 0.85 \times 3 \times 81 \times 32 \cdot 2439$   
 $T_b = \frac{208 \cdot 1240}{64}$ , A

.

From Equation ()

 $a_{b} = \frac{208 \cdot 1240}{5}$  $a_b = 41.6248$ , mm<sup>2</sup>. in) Area of end ring  $\delta_e = \frac{T_e}{q_e}$  $a_e = \frac{I_e}{\delta_o} \rightarrow \textcircled{4}$ Ie = 8, Ib TP  $J_{e} = \frac{64 \times 208.1240}{\pi \times 6}$ Ie = 706.6446, A From Equation (4)  $a_e = \frac{106.6446}{5}$  $q_e = 141 \cdot 32.89$ , mm<sup>2</sup>.

A 90 kW, 500V, 50H2, 3¢, 8 pole, Induction motor has a star connected stator winding accomadated in 53 Slots with 6 conductor [slot. If the slip ring Voltage on open circuit is to be about 400V. Find the Suitable rotor flots, normber of rotor conductors per slot, approximate full load current per phase in rotor. Assume  $\gamma = 0.9$ , cos  $\phi = 0.86$ ,  $\delta = 5 A | mm^2$ , Also find the area of rotor conductors.

Given data

Po=90 kN,  $V_L = E_L = 500 V$ ,  $f = 50 H_2$ , 3 phase, Star connected Statos winding, p=8,  $S_S = 53$ , Statos conductos [slot ( $Z_{SS}$ )= 6,  $E_{TL} = 400 V$ . f = 0.9,  $\cos \phi = 0.86$ ,  $f = 5 A [mm^2]$ .

") Rotog Slots, Number of rotog conductors peg Slots in") area of rotog conductors.

Solution

1) Roton Slots

 $s_g-S_r$  should not be equal to  $\pm 0, \pm 1, \pm 2, \pm P, \pm 2P$ ,  $\pm 3P, \pm (P\pm 1), \pm (P\pm 2)$ 

 $s_{s}-s_{r}$  should not be equal to  $0, \pm 1, \pm 2, \pm 8, \pm 16, \pm 18, \pm 7, \pm 9, \pm 6, \pm 10$ 

 $\Im_{g-S_{3}}$  Should be equal to  $\pm 3, \pm 4, \pm 5, \pm 11, \pm 12, \pm 13, \pm 14, \pm 15 \pm 17$ 

1	Re	The second se	
	q	Sr = gx polex phase.	$S_{S} - S_{\gamma}$
	2	48	63-48=15 1
	3	72	68-72=9 x
	4	96	63-120=53
	5	120	63-120=57 1
m	ii) Lot	$s_r = 4^{g}$ , the di. m, so we selected $sph = 4.44f \phi T_{gp}$ $sph = 4.44f \phi T_{gp}$	n kws.
		<u>sph</u> = <u>Tsph</u> Tsph Trph	
		Toph = Toph Er Esph	

Given Statog Conductor / slot = 6, and Sg = 53

". Staton Conductors Zg = Staton Conductor × Slot

 $Z_{s} = 6 \times 53 = 318$ 

$$T_{sph} = \frac{Z_s}{6} = \frac{318}{6} = 63$$

Ne know that for dlip ring rotor, the coinding 18 Star connected

$$E_{rph} = \frac{E_{rl}}{\sqrt{3}} = \frac{400}{\sqrt{3}} = \frac{230.9401}{\sqrt{3}}, \sqrt{3}$$

and also given Statos winding is Star Connected

$$sph = \frac{-s_{L}}{\sqrt{3}} = \frac{500}{\sqrt{3}} = \frac{288.6751}{\sqrt{3}}$$
 V

From Equation (

 $T_{aph} = \frac{63 \times 230.9401}{288.6751} = 50.4 \simeq 50$   $T_{aph} = 50$ 

 $Z_{rph} = 6x50 = 300$ Rotog Conductog / Slot (Zrs) =  $\frac{300}{48} = \frac{625}{48}$ Rotog Conductog / Slot (Zrs) = 6

( De cause for high Capacity motor conductor / slot should be a Even Integer).

h - mark

S I LANCE

$$Trph(new) = \frac{2r(new)}{6} = 48$$

$$Trph(new) = 48$$

iii) area of rolog conductors  

$$\delta_r = \frac{I_{rph}}{a_r} \left( \delta_r \rightarrow \text{current density} \right)$$

$$a_{r=\frac{1}{\delta_r}} \xrightarrow{\gamma} (2)$$

To find Irph, assume rotor mont is 85% of Statos mont

$$\frac{J_{\text{sph}}}{48} \rightarrow 3$$

we know that

Input @ in KVA = 
$$3 \times Esph \times Isph \times 10^3$$
,  $kvA$   
 $Isph = \frac{Input @ in kvA}{3 \times Esph \times 10^3}$   
 $Isph = \frac{Input @ in kvA}{3 \times 288.6751 \times 10^3} \longrightarrow @$ 

$$= \frac{90}{0.9 \times 0.86}$$

Input Q in kVB = 116.2791, kVAFrom Equation (2)  $I_{sph} = \frac{116.2791}{3x288.6751x10^3}$ 

From Equation (3)  $I_{rph} = \frac{0.85 \times 63 \times 1342675}{48}$ 

From Equation (2)

 $a_r = \frac{149.7922}{5}$  (given  $\delta = 5.4/mm^2$ )  $a_r = 29.9584$ , mm<sup>2</sup>

Answers

Roton Blot Sr = 48 Roton Conductor/Blot = 6 Roton Conductors = 288 Roton turns | phase = 48 Roton turns | phase = 149.7922, A Roton Current/phase = 149.7922, A Area of Roton Conductors = 29.9584, mm <u>Circle Diagram</u>:

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It is an graphical method to determine the performance of an Induction motor

To construct an circle diagram, the following data's are needed.

i) No load test:

→ gives the information about rated voltage (Vo) and no load current (Io) and no load power (Wo) ii) Blocked rotor test data:

and we both

→ gives the information about blocked rotor Voltage (Nb), Blocked rotor current (Ib) and Blocked rotor power (Wb)

iii) Stator Resistance test data:

 $\rightarrow$  gives the information about stator resistance/

convert the above data into per phase values.

From the above datas, calculate the go, gb, Ibn, Wbn.

where, 
$$\varphi_0 = \cos^{-1} \left( \frac{\omega_0}{\sqrt{3} \sqrt{0} I_0} \right)$$
  
 $\varphi_b = \cos^{-1} \left( \frac{\omega_b}{\sqrt{3} \sqrt{b} I_b} \right)$ 

 $I_{bn} \rightarrow current drawn by the blocked rotor motor is at rated voltage.$ 

the total no an the allow

$$T_{bn} = T_{b} \times \left( \frac{V_{o}}{V_{b}} \right)$$

Whn -> power taken by blocked rotor motor at rated voltage.

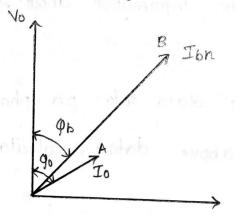
$$Wbh = Wbx \left(\frac{Vo}{Vb}\right)^{2}$$

Procedure to construct circle diagram:

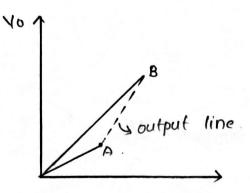
1). Take rated voltage vo as the reference phonor.

B) Choose a suitable current scale and convert To and Ibn in cm.

3). Draw the vector Io lags Vo by an angle to and Ibn lags Vo by an angle 96.

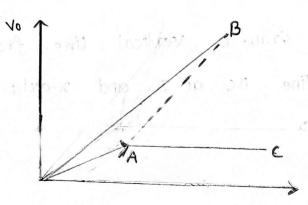


4) Join the points A and B, the Line AB represent the output line.

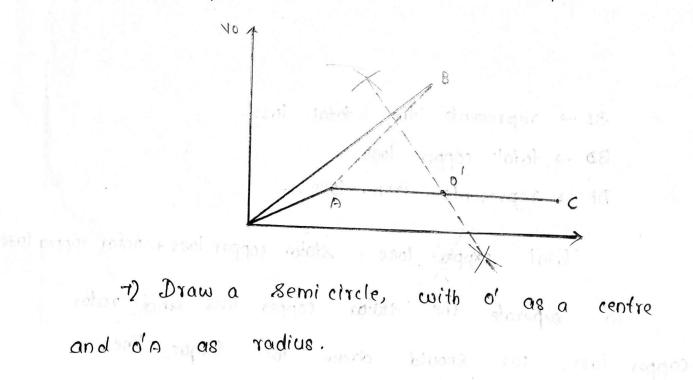


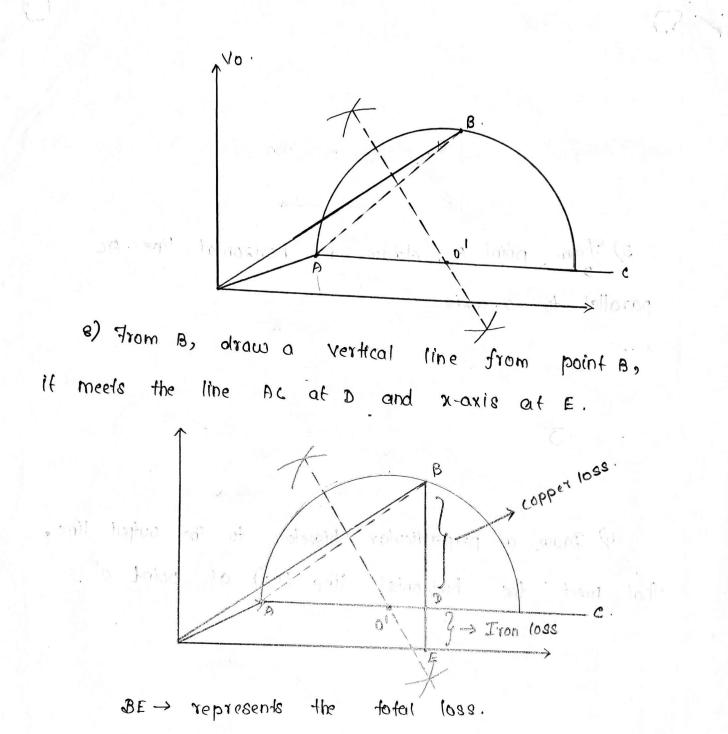
5) From point A, draw a horizontal line Ac parallel to x-axis.

1. Alect



6) Drow a perpendicular bisector to the output line, that meet the horizontal line (Ac) at point o'.

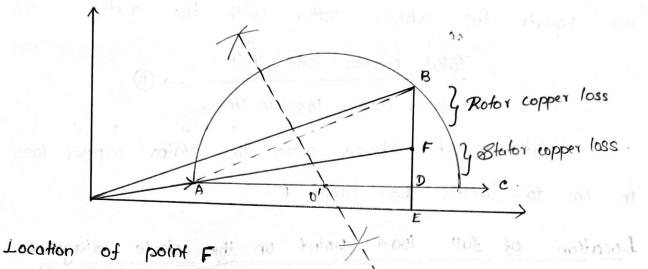




BD -> total copper loss.

 $DE \rightarrow represents$  iron loss.

Total copper loss = stator copper loss + rotor copper loss. To seperate the stator copper loss and rotor Copper loss, we should draw the torque line. 9) The vector AF is the torque line which seperates the stator and rotor copper 1083.



Draw the torque line based on the ratio <u>Stator copper loss</u>, If this rate is not given. Rotor copper loss

In case of cage rotor,

Stator copper loss 3I, RI Rotor copper loss Won-3I, RI

In case of slip ring rotor,

 $\frac{\${\text{Stator copper loss}}}{\texttt{Rotor copper loss}} = \frac{\Im I_1 \overset{?}{R_1}}{\Im I_2 \overset{?}{R_2}} = \frac{\Im I_1 \overset{?}{R_1}}{I_2 \overset{?}{R_2}} \overset{?}{R_2} \overset{?}{R_2}$ 

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From the circle aliagram, the total copper loss is represented by BC and is given in cm. So we equate the above ratio with the ratio

> Total copper loss in cm. -> (1) Stator copper loss in cm

Solution of above gives the stater copper loss in cm to locate the point F.

Location of full load point on the circle diagram. (ase (i) : If full load current is not given,

From point B, draw a vertical line BG.

Length of BG = Rated Output power scale.

hihere, power scale = Whn Length of BE.

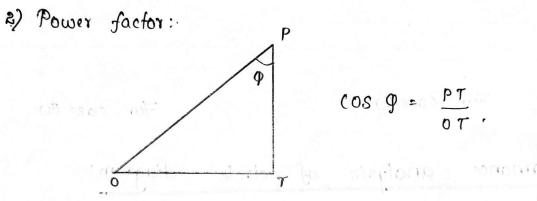
From point Gr, draw a parallel line to the output line. This parallel line meet the semicircle at point P. This point is called full load point.

case(ii) : If full load is not given.

Convert the Ifr in ampere to cm and draw an arc on semicircle by taking o as centre with Ifr as radius. The meeting point P, output line, forque line, horizontal meets line. and x axis at points Q, R, s and T respectively and to meed x-axis at point T

Porformance analysis:

1) Line current = Length of op x current scale.



3) Input power = Length of PT x power scale.

4) Output power = PQ x power scale.

5) Efficiency  $\eta = \frac{PQ}{PT} \times 100\%$ 

6) Rotor copper loss = Length of QPX power scale.

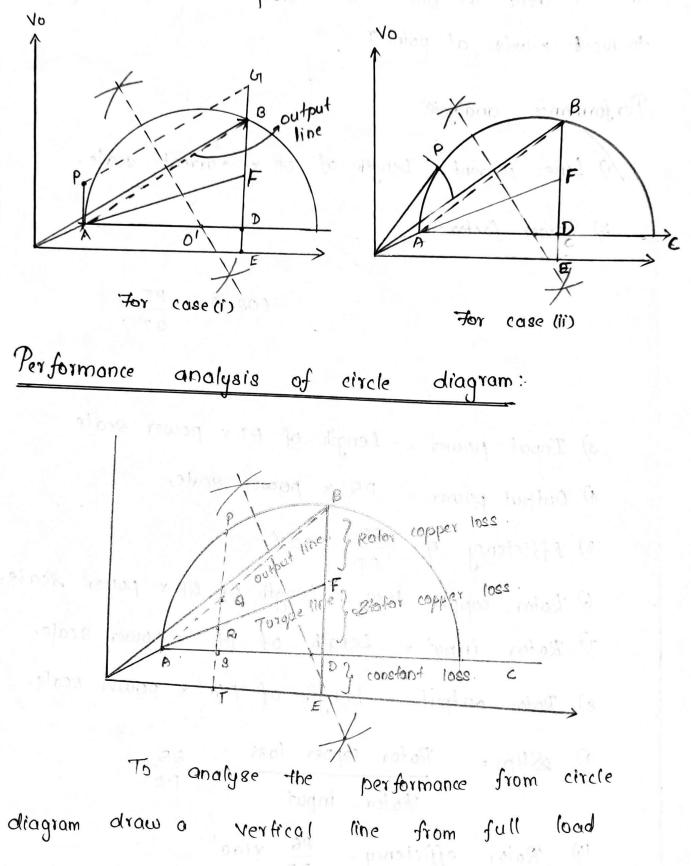
7) Rotor input = Length of PR x power scale.

8) Rotor output = Length of PQ x power scale.

9) 
$$\mathcal{B}$$
 lip = Rotor copper loss  $\frac{QR}{PR}$ .  
Rotor input  $PR$ .

19) Rotor efficiency = 
$$\frac{PQ}{PR} \times 100$$
.

point of arch on the semicircle at point p is called full load point.



1.2

\*\*

11) Maximum Input, Maximum output and Maximum Torque.

#### Maximum Input

Locate a point H on the mid of the semiciencle. From point H, draw a vertical line meets the x axis at Point I

Naximum Input = length of HI × Power scale.

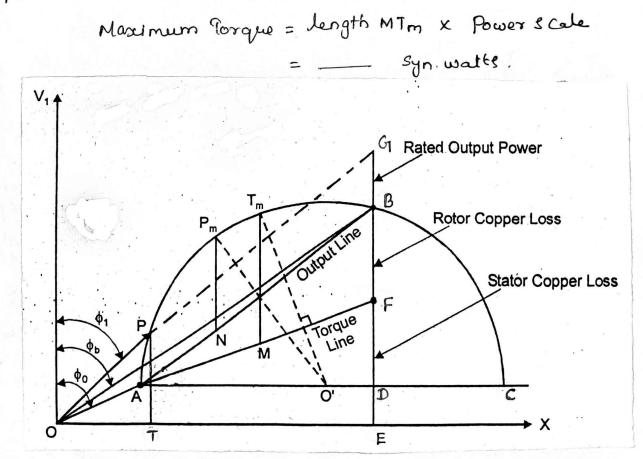
#### Naximum output

Draw a line O'Pm Ir to the output line. From the point Pm draw a vertical line so as to meet the output line at N.

Maximum output = length NPm × Power Scale.

#### Maximum Torque.

Draw a line O'Tm I'r to the torque line. From the point Tm, draw a vertical line so as to meet the torque line at the point M.



## **EE 8002 DESIGN OF ELECTRICAL APPARATUS**

## UNIT V

### **DESIGN OF SYNCHRONOUS MACHINES**

Prepared by Dr . T. Dharma Raj Asso.Prof /EEE

#### Synchronals Generator

A Synchronous machine consists of two major parts.

i) armature and ii) -field System.

Based on the arrangement of the above parts, it is classified into two types.

- i) Revolving armature and
- ii) Revoluing field system.

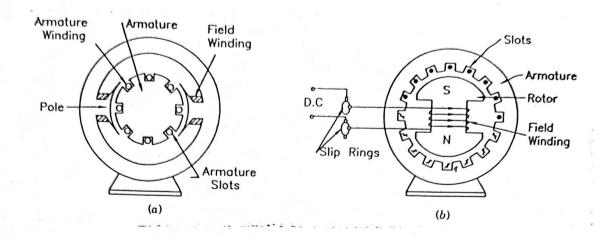
Revolving armature System is Similiar to DC machines This type of Construction is used only for low power Synchronous machines and is unsuitable for medium and high: power machines.

Revolving field System has the following advantages.

i) it permits the use of a stationary armature on which the windings can be easily braced and insulated for high voltage.

2) The use of Slip rings Carrying large currents at high voltages is avoided in the stationary armature

The Revolving field system Synchronous machines are classified into two types. 2) Non Salient pole (07) Cylindrical pole Machiner. 2) Salient pole Machines.



Solient pole machines have projecting poles with concentrated field windings shown in figure (b). The Salient pole construction is used for generators driven by hydraulic turbines (water turbines). Since these turbines at relatively down Speeds and it needs a dange number of poles to produce the desired frequer  $(N = \frac{120f}{p})$ 

cylindrical pole machines have their field winding distributed in slots is shown in figure (9).

The cylindrical rotor construction is used for turbo-alternators which are driven by high speed Steam or gas turbines. Comparision of Salient pole and non salient pole Nachines

Salient pole	non salient pole.
1) Large diameter and axial length	Small diameter and long axial length
ii) Water turbines actals prime movers iii) Has projected poles	steam turbines act as prime movers
iv Needs damper wags	Has no projected poles
	Does not need damper Windings.

# Prime movers for Synchronous Machines.

i) Salient pole -> Water turbines act as promemover, hence it 18 known as whater water wheel generators or hydro generators. The type of turbine to be used depends upon the water head available. The use of hydraulic turbines at various heads is listed below.

> Water heads 400m and above - Pelton wheel Water heads upto 380 m - Francis turbine Water heads upto 50 m - kaplan turbine.

i) non salient pole -> Steam turbines act as primer moxer hence it is called turbo alternators.

#### Run away speed

The run-away speed is defined as the speed Which the prime mover whoat would have, if it is Suddenly unloaded when working at its rated load.

The following are the runaway speeds permissible for Hydro generators

> 1) Pelton wheel - 1.8 times rated speed ii) Francis turbine - 2 to 2.2 times rated speed iii) kaplan turbine - 2.5 to 2.8 times rated speed.

Thus the Salient pole (03) Hydro generators are designed to coithstand mechanical stresses encountered at run away speeds.

The maximum peripheral speed for which. Satient pole machines are designed is about 140 m/s. While for turbo alternators (non satient pole) are designed with maximum peripheral speed of about 175 m/s.

### <u>DMain Dimensions</u>

Dand L'are the main dimensions of salient pole machines, where D is the inner diameter of the stator and L is the length of the stator.

#### 2) Output Equations

The output Q in KVA = 3 Esph Isph X 103 KVA -> () Where Esph = 4:44 f & Tsph kws Volts.

from equation (1)

- Output Q in kun = 3x 4 44f & Tsph Lus Isph x10 kvn -> (2)

In Synchronous generator, the frequency is proportional to the speed of the prime mover, that is given by the expression

$$V = 120f = \frac{2x'60'f}{P}$$

$$f = \frac{NP}{2x'60'} = \frac{Pn}{2}$$

from equation 2

Output Q in KUA = 3x 4 44  $\frac{Pn}{2} \neq T_{sph} kws J_{sph} x 10 kva$  $Ne know that <math>B_{aw} = \frac{P\phi}{\pi p_{L}}$ ,  $P\phi = B_{av} \pi p_{L}$ Output Q in kva =  $3x 4.44 \frac{n}{2}$  Bav  $\pi p_{L}$  Toph kws  $I_{sph} x 10^{3} kva$ 

Output Q in KUA = 666 Baw TOL Kus Tsph Isph XIO KUA > 1

$$T_{sph} = \frac{Z_s}{6}$$
 For n number of parallel path.  
 $T_{sph} = \frac{Z_s}{2s}$ 

Iz = Isph

For n numbers of ll<sup>ell</sup> paths Iz=Isph; Isph=AIz A

From Equation (7)

Output a in kun = 666 n Baw TDL Kuss Zs. Iz X10 KUA

Output Q in KUA = 111 N Bay TOL KWS Zs. Iz X 10 KUA

We know that  $Q_{C} = \frac{I_2 \cdot Z_S}{TD}$ 

$$L_2 \cdot Z_S = ACTD$$

autput Q in KUA = 1.11 n Bay TOL KWS aCTDX10<sup>3</sup> KNA

The Output Q in KVA = 1:11 3 D<sup>2</sup> L n Bau 90 kws x 10 KUA

3) Choice of specific electric and magnetic loading

a) choice of specific Electric Loading

The specific electric loading is defined as the ratio of total ermature ampere conductors to the armature periphery at air gap

 $ac = \frac{\mathbf{J}_z \cdot \mathbf{Z}_s}{\mathbf{T} \mathbf{D}} \rightarrow \mathbf{O}$ 

The following factors are considered for the choice of Specific Loading

i) Copper loss and temperature rise

ii) Voltage

iii) Synchronous reactance

iv) Stray Load loss

1) Copper loss and temperature rise.

From Equation  $ac=\frac{I_Z\cdot Z_S}{\pi D}$ , the higher the Value of at which increases the  $T_Z$  and  $Z_S$ . Therefore increase in Current, increases the temperature rise and increase in area of Conductors, increases the copper loss which resulting low efficiency.

Therefore low value of ac is selected, to reduce the copper loss and temperature rise.

ii) Voltage.

For high voltage machines, the conductor area is Less, because large space is required for insulation. Therefore the low value of ac is used for high voltage machines.

The higher value of ac can be used for low machines, because the space required for insulation is small.

iii) Synchronous Reachance.

From the definition  $ac = I_z \cdot z_s$ , a high value TDof ac leads to high value of leatage reactance. (due to increase in  $z_s$ ) results a high value of Synchronous reactance.

Therefore a machine designed with a high value of ac will have high Xs results

i) poor inherent voltage Regulation

ii) low arrient under short circuit conditions

iii) low value of steady state stablity limit and small synchronizing power and consequently leads to instablity (Pmax = 3 Eph Vph)

iv) Stray Load loss

The stray load loss increases steeply with an increase in ac.

Therefore the typical values of ac used in Synchronous machines

Salient pole machines - 20,000 to 40,000 A/m. Turbo-generators - 50,000 to 75,000 A/m. b) specific magnetic loading (Baw)

It is defined as the ratio of total flux around the air gap to the area of the armature periphery at the air gap

$$Bav = \frac{P q}{\pi DL} \rightarrow @$$

The following factors are to be considered for the Choice of specific magnetic loading

- i) Iron loss
- ii) Voltage
- iii) Transient Short Circuit Current
- iv) stablity
- V) parallel operation

i) Iron Loss

A high value of Bau leads to a high value of flux density in the stator cose and teeth. This increases the roon loss, because the flux density is proportional to mon loss.

Therefore a low value of Bau 18 Selected

#### ii) voltage

From the definition of fur density,  $B_{t} = \frac{\Phi_{t}}{P_{t}}$ 

For high voltage, the space occupied by the insulation is greater and smaller space is left for teeth. Therefore decrease in the area of the teeth, increases the value of flux density in teeth and core.

Therefore to avoid excessive values of flux density, a low value of gap density is used for high vollage machines.

iii) Pransient Short Circuit Current.

from the Emf Equation Es= 4:44 f \$ Ts kwg Upits

By Substituting the f and pp value in above Equation, we get

Es=444 n Bau TOL To kus Volts

Eg & Bau

Higher the value of Bar, increases the Eg which intermediate increases the short Circuit Current [.  $I_s = \frac{E_s}{x_s}$ ]

Therefore to limit the initial electromagnetic force under short circuits condition, a low value of gap deneity should be used.

### iv) Stablity

The maximum power delivered by the Synchronous machine is given by

From the above equation, the maximum power is inversity proportional to its synchronous reactance. If a high value of gap density is used, the flux per pole is large and therefore a smaller number of twons are required for the armature winding. This results in reduction in the value of synchronous reactance, which inturn increases the Synchronising power.

Therefore the use of high gap density improves the Steady state stability limit.

v) parallel operation.

For load Sharing, the synchronous generators are Connected in parallel with other synchronous generators. The statisfactory parallel operators depends on Synchronising power Pmax = 3 Eph Nph Xsph.

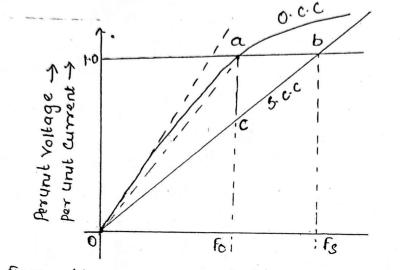
If a high value of Baw is used, results in reduction of Synchronous reactance which in turn increases the Synchronising power. Therefore the machine designed with high value of gap density have the satisfied parallel operation.

Therefore the typical values of Base used in Synchronous machines are

> Salient pole machines - 0.52 to 0.65 wb/m². Turbo generatore - 0.54 to 0.65 wb/m².

### 4, Short Circuit ratio

It is defined as the ratio of field current required to produce raled voltage on open circuit to field current required to circulate rated current at short circuit.



From the por unit diagram

$$SCR = \frac{OFo}{OFs} = \frac{CFo}{bFs} = \frac{cFo}{aFo} = \frac{1}{aFo/cFo}$$

Circuit

SCE = 1 , Thus the short circuit Ratio is the reciprocal of Synchronous reactance Xd.

The following are the typical values of sce used in Synchronous Machines

> Satient pole machines -> 10 to 1.5 Purbo generators -> 0.5 to 0.7.

Effects of sce on machine performance

(i) Voltage Regulation

From the definition of SCR,

SCR = \_\_\_\_\_ Per unit Voltage on open circuit ~> ()

Corresponding per unit current on short circuit

$$sce = \frac{1}{Xd} \rightarrow \textcircled{2}$$

rom eq. (1), A low value of ECR, results a greater changes in Voltage under fluctation of load. Therefore the Voltage regulation of the machine is poor.

# il) stablity

The maximum power delivered by the synchronous

$$P_{\text{max}} = \frac{3 \text{Eph Vph.}}{X_{\text{sph}}} \rightarrow 3$$

From equation (3), the low of scr., high Value of Synchronous reactance.

From Equatio (3), the high value of Xsph reduces the synchronising power.

Reseptive the low value of sce ... lower

iii) parallel operation

The synchronising power required to keep the parallel operated machines is given by

> Pmase = 3 Eph Vph Xsph

A low of value of SCR increases the sychronous reactance Keph from equation @.

This increased Xsph decreases the Synchronising power. These fore the low value of sce gives un Statisfactory parallel operation.

iv) Short circuit averent.

From equation 2, a dow Value of SCE, means that the Synchronous reactance has a large value

The short creat current is given by  $T_{s=\frac{t_s}{X_{s_1}}}$ , larger the value of  $X_s$  results in Smaller the value of short circuit current.

Reseptive the small value of SCR limits the Shoot circuit current.

iv) Self Excitation

Synchronous generator connected with long transmission lines should not be designed with small scr, because this small value of scr lead to long s voltages on open circuit produced by self excitation owing to large capacitive Currents drawon by transmission lines.

# 5) Longth of air gap.

The length of air gap greatly influences the performance of the synchronous machines. The reluctance of the airgap is given by

Increase in airgap length increases the reluctance produced by the armature mmf and reduces the effects of Armature reaction. This results in a small value of Synchronous reactance and high value of SCR.

Thus the machine with a large air gap has

i) a small value of Regulation

ii) a higher value of stablity limit

iii) a higher synchronizing power which makes the machine less sensitive to load variations.

But increase in airgap length, needs a larger field mmf, resulting in increase of cost of the machine. Staton design of salient polo Machines.

1) Main dimensions

The output av in KVA= 111 7202Ln Bawackwx103 KYA D<sup>2</sup>L = WinkVA 1117 n Bowackwxio  $D^2L = - m^3$ ii) Seperation of Dand & from DL In Salient pole machines, the Belection of D depends upon a) the type of pole used b) the permissible peripheral speed. a) Type of polo used i) Round poles  $\rightarrow \frac{L}{c} = 0.6$  to 0.7 ii) Rectangular poles -> 1= 165 b) The permissible peripheral speed i) Bolted construction > 50m/s ii) Dove tailed and Thead Construction -> 80 m/s. The rotor should be designed to with stand Centrifugal foras produced under randway speeds.

- à) statos slot
  - i) Ys ≤25mm for low Vollage machines
    Ys ≤40mm for 6kv or low voltage machines
    Ys ≤60mm for machines upto 15 kv
  - ii) slots pole phase is usually between 2 to 4

### b) Statos Conductors

 $Fs = 4 4 4 f \phi Ts kws Volls \rightarrow 0$   $Fs = \frac{Fs ph}{Ph} \xrightarrow{\Rightarrow 0} where$   $Fs = \frac{Fs ph}{4 4 4 \phi kws} \xrightarrow{\Rightarrow 0} Ts \rightarrow stator tarns/phose$ 

where  $\phi = \frac{Bau TDL}{P}$  $\frac{Z_{S} = 6.7_{sph}}{2}$ 

For n number of parallel paths, the equation  $0 \Rightarrow E_{sph} = 4.44 \text{ f} \oplus \frac{\text{Tsph}}{\text{P}} \text{ kws volts}$ 

check cond/slot should be a integer or not, if it is not a integer means makes it a integer value.

1) Als gap length

Let us assume that mmf required for the airgap is 80%. of no load field mmf (ATto)

$$lg = \frac{0.8 \times AT_{fo}}{8,00,000} \text{ kg Bg}$$

where Bg is calculated, by using the formula  $k_f = \frac{Poleanc}{Polepitch} = \frac{Bau}{Bg}$ . (or)

where kg > gap contraction factor Bg > maximum flux density Determine the dimensions for a 1000 kvA, 50 Hz, 34 375 rpm alternators. The average airgap is 0.55 wb/m?. and the ampere Conductor me per meter are 28000. Use rectangular poles and assume a suitable value for ratio core length to pole pitch in order that bolted pole construction is used, for which maximum permissible speed is 50 m/s. The runaway speed is 18 times the synchronous speed.

Given data

Qr=1000 kVA, f=50 Hz, 34, N= 373 rpm, Baw=0.55 cob/m<sup>2</sup>, Qc = 28000, Rectangular pole, Va not exceed 50 m/s. The runaway speed is 1.8 times the synchronous speed.

Testing

Main dimension.

Solution

The output Equation of Synchronous machine is given by Output QV in  $kVR = 1.11 \pi^2 D^2 L n Bavac kwsx10^3$ , kVR $D^2 L = \frac{Output QV in kVR}{1.11 \pi^2 n Bavac kws x10^3}$ 

$$D^{3} = 5.0591$$

$$D = 1.7167, m$$
From equation (B)
$$L = 0.963 \times 1.7167$$

$$L = 0.3370, m$$
Check for peripheral speed and Furnamony peripheral speed
Peripheral speed  $V_{a} = T D n$ 

$$= \pi \times 1.7167 \times (\frac{315}{50})$$

$$= \frac{33.7073}{50} m/s$$
Permany peripheral speed = 1.8 × Va
$$= 1.8 \times 33.7073$$
Renamony peripheral speed = 60.6132 m/s
The Pernamony peripheral speed = 60.6132 m/s
The Pernamony peripheral speed = 60.6132 m/s
The Pernamony peripheral speed = 60.6132 m/s
$$The Pernamony peripheral speed = 200 m/s. The select
$$\frac{L}{T} = 2$$

$$L = 2T$$

$$L = 2T$$$$

P and have shared from the

$$\begin{array}{l} \downarrow = \frac{2\pi D}{P} \\ \downarrow = 0.3981 \ D \rightarrow \textcircled{0} \end{array} \\ \hline From Equation (D) \\ D^{3} \times 0.3981D = 0.9931 \\ D^{3} \times 0.3981D = 2.5289 \\ \hline D^{3} \times 0.3981 = 2.5289 \\ \hline D = 1.3624, m. \end{array} \\ \hline D = 1.3624, m. \end{array} \\ \hline From Equation (D) \\ \bot = 0.53921 \times 1.3624 \\ \hline \bot = 0.535, m. \end{array} \\ \hline Check for peripheral Speed and ranaway peripheral Speed Speed Va = \pi D n \\ = \pi \times 1.3624 \times (\underbrace{315}_{60}) \\ = \underbrace{26.7501}_{501} \ m/s \\ \hline Penaway peripheral Speed = 1.8 \times V_{6} \\ = \underbrace{48.1513}_{51}, \ m/s \\ \hline The runaway peripheral Speed is within the permissible speed 50 m/s. \end{array} \\ \hline D = 1.3624, m. \\ \hline D = 1.3624, m. \\ \bot = 0.535, m. \end{array}$$

find the main dimensions of a 2500 kVA, 187.5 spm, 50H2, 8 phase, 3KV, Salient pole Synchronous generator. The generator is to vertical, water wheel type. The Specific magnetic loading is 0.6 wb/m<sup>2</sup> and the Specific electric loading is 34000 A/m. Use circular poles with ratio of core length to pole pitch = 0.65. Specify the type of pale construction used. If the runaway speed is about 2 times the normal speed.

## given data

Qr = 2500 KVA, N= 1875 Jpm, 50 Hz, 39, 3# V\_= 3KV=3000 Volt, Salient type gng, Bau = 0.6 cob/m<sup>2</sup>, ac = 24000 A/m, Cigcular poles, <u>L</u> = 0.65. Renaway speed = 2 times the Normal speed.

# Potod

Main dimensions.

Colution

The Output Equation of Synchronour machine is

Quiput Qr in kVA = 1.11 7 D<sup>2</sup> Ln Bau ac Kws x 10<sup>3</sup>, kVA

$$D^{3}L = \frac{2500}{\Gamma 11 \times \pi^{2} \times 1875} \times 0.05 \times 34000 \times 0.955 \times 10^{3}}$$

$$D^{3}L = \frac{3.7483}{\Gamma 1875} \times \frac{1875}{L00} \times 0.05 \times 34000 \times 0.955 \times 10^{3}}$$

$$D^{3}L = \frac{3.7483}{L} \times \frac{1875}{L} \rightarrow 0$$
Given Ciscular poles and  $\frac{1}{L} = 0.65^{\circ}$ 

$$\therefore L = 0.65 \times \frac{\pi}{D}$$

$$L = 0.65 \times \frac{\pi}{D}$$

$$P = \frac{120 \times f}{R}$$

$$P = \frac{120 \times f}{R}$$

$$P = \frac{120 \times 50}{R}$$

$$P = \frac{32}{R}$$
From Equation (1)  

$$D^{3}L = 0.0638 \text{ D} \rightarrow (2)$$

$$D^{3}L = \frac{3.7483}{0.0638}$$

$$D^{3}L = 58.7508$$

$$D = 3.8815, \text{ m}$$
From Equation (2)  

$$L = 0.0638 \times 3.8815$$

$$L = 0.0638 \times 3.8815$$

Check for peripheral Speed and randowy peripheral Speed

pbripheral Speed 
$$a = \pi D_{n}$$
  
=  $\pi \times 3.8875 \times (187.55)$   
=  $38.1654$ , m/s

$$= 2 \times 38 \cdot 1654$$

$$= 76 \cdot 3309 \text{ m} s$$

For this Runaway peripherial speed, Dove tailed type of pole construction is used.

Answers

$$D = 3.8875$$
, m

$$L = 0.2480, m$$

Polie Conconstruction is Dove tailed

Find the main dimensions of a 100 MVA, 11 kV, 50 Hz, 150 spm, 39, water wheel generators. The average gap density 18 0.65 cob/m<sup>2</sup> and ampere conductor per meter are 40,000. The peripheral speed should not exceed 65 m/s at normal running speed inorder to limit the runaway peripheral speed.

Given date = 100×10<sup>3</sup> KVA Ar= 100 MVA, V=11kV, f= 50 Hz, N=150 spm, 3¢, Bau= 0.65 Lob/m<sup>2</sup>, ac= 40,000, Va not exceed 65 m/s. Run away speed = normal running speed To find

Main dimensions.

Solution

The output Equation of Synchronous machine is

Output Qr in 
$$kYP = 1.117^2 D^2 L n Bay Qc x kiosx jo3, kVA
$$D^{2}L = \frac{Output Qr in kVP}{1.117^2 n Bay Qc kios x 10^3}$$

$$D^{2}L = \frac{100 \times 10^3}{1.11 \times 7^2 \times (150) \times 0.65 \times 40,000 \times 0.955 \times 10^3}$$

$$D^{2}L = 1.47.0485 m^3$$$$

I is choosen based on type of pale used in Construction. Here type of pole construction is not Specified. Hence we Check for circular poles and Rectangular poles. a, For circular poles, L = 0.6 to 0.7 i) when  $\frac{1}{7} = 0.6$ N=120f P L=0.6C  $L = 0.6 \frac{\pi 0}{p}$   $P = \frac{120 f}{p}$  $P = 120 \times 50$ L= 0.6 x 7 D 40 150 P= 40 L= 0.0471 D From Equation ( Minute Chines Fines D2×0.04710=147.0485 agistul St duglad off D = 14.6155, m L = 0.6884, m Output here in Peripheral Speed Va = TDn  $V_a = \bar{n} \times 14.6155 \times (150)$ =114.7899 m/s This peripheral speed exceeds the limit. ii) when  $\frac{L}{T} = 0.07$ L=0077

$$L = 0.71 \frac{\pi D}{P}$$

$$L = 0.71 \frac{\pi D}{40}$$

$$L = 0.0550 \text{ p}$$
From Equation (f)
$$D^{2}x_{1}0.0550 \text{ p} = 147.0485$$

$$D = 13.8811, \text{ m}$$

$$L = 0.7635, \text{ m}.$$
Peripheral Speed  $V_{a} = \pi Dn$ 

$$V_{a} = \pi \times 13.8811 \times (\frac{150}{60})$$

$$= 109.0219, \text{ m/s}.$$
This peripheral Speed aloo exceeds the limit.  
Therefore we try it for Pectangular pales.  
**a** for Pectangular Pales,  $\frac{L}{T} = 1 \text{ to } 5$ 
i) when  $\frac{L}{T} = 1$ 

$$L = \pi$$

$$L = \pi D$$

$$L = \pi D$$

$$L = \pi D$$

$$L = \pi D$$

$$L = 0.0785 D$$

From Equation (1)  

$$D^{0}_{x} 0.0785D = 141.0485$$

$$D = \underbrace{12.3251}, m$$

$$L = \underbrace{0.9675}, m$$
Peripheral Speed  $V_{a} = \pi Dn$ 

$$V_{a} = \pi x 12.325 * x (\underbrace{150}_{60})$$

$$V_{a} = \underbrace{96.8011}, m/s$$
This peripheral speed exceeds the limit  
i)  $\frac{L}{C} = \Re$   

$$L = \Re \pi D$$

$$P$$

$$L = \Re \pi D$$

$$D^{2}x 0.15711 D = 147.0485$$

$$D = \underbrace{9.7824}, m$$

$$L = \underbrace{1.5368}, m$$

Peripheral Speed 
$$V_{q} = \pi D n$$
  
 $= \pi \times 9.7824 \times (\frac{150}{60})$   
 $= \frac{76.8308}{6} m/s$   
This peripheral Speed exceeds the limit.  
i)  $\frac{L}{C} = 3$   
 $L = 3 \tau$   
 $L = 3 \tau$   
 $L = 3 \frac{\pi D}{P}$   
 $L = 3 \times \frac{\pi D}{40}$   
 $L = 0.2356 D$   
From Equation (f)  
 $D^{2} \times 0.2356 D = 147.0485$   
 $1 = \frac{2.0134}{40} \text{ y.m.}$   
Peripheral Speed Va =  $\pi D n$   
 $= \pi \times 8.5457 \times (\frac{150}{60})$   
 $= \frac{67.1178}{60} m/s$   
This peripheral Speed also exceeds the limit.  
iv)  $\frac{L}{C} = 4$   
 $L = 4 \tau$   
 $L = 4 \frac{\pi D}{P}$ 

15

.

L= 0.3142 D

From Equation (1)  $D^2 \times 0.3142D = 147.0485$  D = 7.7643, m L = 2.04396, m

peripheral speed 
$$V_a = TDn$$
  
=  $T \times 7.7643 \times (\frac{150}{60})$   
=  $60.9807$ , m/s

This peripheral speed lies within the limit. Therefore we choose

L= 2.4396, m and type of pole used is Rectangular pole with Thead on Dove tailed Construction. Design of Rotary of Salient pole Machine

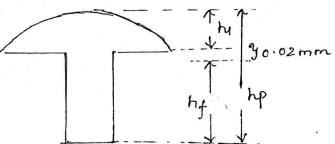
a) Design of Salient pole

1) Area of Cross section of pole

Area of Gross-Section of palebody Ap= 4p Bp

Clux in the pale body p= leakage coefficient x Useful flux/pole,

 $q_p = Q_1 \times q$   $f_p = Q_1 \times q$   $f_p = 0$   $p_p = 0.98$  Lpbp for vactangular pals. Ap = 0.98 Lpbp for vactangular pals.  $Ap = (\frac{T}{4})bp^2$  for Circular pals.  $Ap = (\frac{T}{4})bp^2$  for Circular pals.  $f_p = 18$  taken equal to gross stacking  $f_{action} = f_{action} f_{action}$  $f_{action} = f_{action} f_{action}$ 



where hy is height of pole share

h₂ > height taken by flanges yearly about 20 mm hf > height of field wodg

Height of the pale = hy + 0.02 + hf

Height of the pole shoe (hy) is determined from pole

Estimation of height of the field wdg.

NNIF / metre height of field winding = 1.0 JSFdf 9f

hift = 104 Jsfolf 9f

Height of the field wdg hf =  $\frac{AT_{fl}}{104\sqrt{s_fd_f}}$ 

Where  $\operatorname{PT}_{fl} \rightarrow \operatorname{full} \operatorname{load} field \operatorname{mmf}$   $\operatorname{S}_{f} \rightarrow \operatorname{Copper} \operatorname{Space} \operatorname{factor},$   $\operatorname{Q}_{f} \rightarrow \operatorname{loss} \operatorname{per} \operatorname{unit} \operatorname{Surface}, \operatorname{K}^{9}/\operatorname{m}^{2}$   $\operatorname{d}_{f} \rightarrow \operatorname{depth} \operatorname{of} \operatorname{the} \operatorname{field} \operatorname{wdg}$  $\operatorname{Q}_{f} = \lambda \circ = \operatorname{L} \circ$ 

where cis the cooling coefficients for rotating coils

$$C = 0.08 to 0.12$$

$$9_{f} = \left(\frac{1 \pm 0.1 Va}{0.08 \pm 0.012}\right) @ W/m^{2}$$

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An approximate estimation of full load field mmf can be made by the method given below.

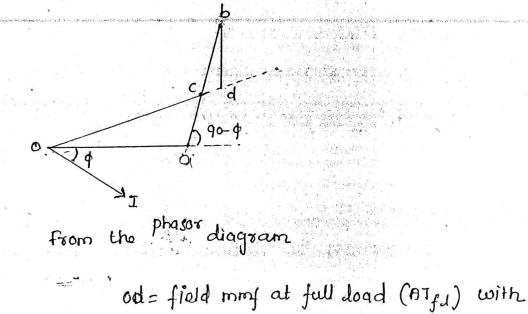
No Load field mmf = ATto = SCR × ATa

where ATa (armature mont/pale) = 2-7 Ts Is Ews

- i) Draw OQ = ATfo
- ii) Drow ab = ATq at angle (90-p) to oa,
- iii) Cut off as such that  $\frac{ac}{ab} = K$ , where kr is called the

Cross reaction coefficient which depends upon the ratio pole arc to pole pitch

iv) Join of and extend it. Drop a perpendicular from b on of extended, culting it at d.



power factor cosq (lagging).

b) Design of field winding.

The following procedures are followed to design the field winding

i. The field winding Should be designed for a voltage from 15 to 20 percent less than the exciter voltage.

Vollage across the field  $\operatorname{coil}(E_f) = (0.8 \text{ to } 0.85) \text{ Ve} p \rightarrow 0$ 2. Determine the height of field winding from the equation hp = hy + 0.02.  $+h_f$ 

$$h_f = h_p - h_i \leftarrow 0.02, \quad m \rightarrow (2)$$

3. Assume suitable depth of the field winding from below

Pole pitch (mm)	winding depth (mm)
0.1	25
0.5	35
٥.4	45

4) Voltage across the field coil

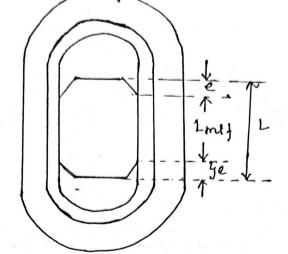
$$F_{f} = I_{f} P_{f} = I_{f} T_{f} P_{I} I_{f} I_{f}$$

where If Tf = ATfl is the full load field mmf.

table

Longth of mean term of field winding.

This can be determined from the below figure



 $Lmtf = 2Lm + \pi (bp + 0.01 + df)$ 

tm=0.9L and e=0.05L

5. Determine the field current Ig by assuming the Current density of is s to 4 A/mm<sup>2</sup>  $I_f = \begin{array}{c} S_f \times a_f \\ \rightarrow \end{array}$ 6. Determine the field turns  $T_f$  from AIf1  $P_fI = I_fI_f$   $T_f = \frac{P_fI_f}{I_f} \longrightarrow$   $T_f$  check the winding space, whether it is possible to accommodate the field turns or not. In case the winding space available is less, increase the depth and if the Space is more, decrease the depth till the winding fits in

vili) The resistance of the winding is calculated by

 $R_{f} = \frac{T_{f} \mathcal{R} \ L_{mulf}}{\alpha_{f}} \rightarrow \textcircled{5}$ 

copper loss in each field coil at 75°C, ay = Ij Rf

$$Q_{ij} = I_{j}^{2} T_{f} \ell L_{mt} \rightarrow 0$$

This loss is to be dissipated by the field Coil and therefore we must check that the temperature rise is within the limits Dissipating surface of the Coil is (S) = 2 LmtfC hft df  $\rightarrow 7$ 

Cooling Coefficient to rotating field coils (Cf)= 0.08 to 0.12 1+0.1 Va

." Temperature rise  $0 = \frac{Q_{1f}C_{f}}{S} \rightarrow 8$ 

If the temperature rise of the Coil exceeds the limits, Increase the depth of field winding. The increase of depth of field winding increases the heat dissipating surface of the Coil from equation (7). and decrease the temperature rise ifrom equation (8) (2) Design of damper winding

The design of damper winding depends upon the purpose for which it is provided.

In Synchronous generators, It is provided to supres the negative sequence field and to damp the oscillations when the machine starts hunting

In Synchronous motor, its Junction is to provide Starting torque and to develop damping power when the machine starts hunting.

The design of damper winding to suppress the inverse.

The amplitude of fundamental of mmy AT, of one phas of a polyphase winding 18 obtained by the equation

$$\frac{\partial T_{1}}{\partial T} \stackrel{f}{\to} \frac{\partial}{\partial T}_{m} k \omega_{1} \rightarrow 0$$

- where ATm = q Zss Isph >3
  - 9 → 8 lots/pole/phase 255 → 8tator Conductors/8 lot Is → Stator Current/phase.

From (2)

$$AT_m = \sqrt{2} \frac{T_{spb}. I_{sph}}{P}$$

$$\begin{array}{rcl} \text{AT}_{1} &= & 4 & \sqrt{2} & \text{Tsplsphkw}_{1} & \rightarrow & \textcircled{3} \\ \hline & & & D \end{array}$$

This pulsating monf is resolved into two rotating monfs, one called the synchronous monf and the other inverse monf each having half the magnitude as above. If the damper winding is to suppress the inverse rotating field, it must develop an equal monf as that of the inverse field.

$$P_{p} = \frac{1}{p} = \frac{1}{p$$

Filso

Equating the equations (5) and (6)

$$acz = 6 I_{sph} I_{sph}$$

$$P$$

$$T_{sph} I_{sph} = acz$$

$$P = 6$$

Equation (1) ->

mmf for damper winding =  $\frac{4J_2}{2\pi} \propto \frac{acz}{b}$  kiu,

= 0.15 acz

Let Ad be the total area of clamper bars/pole and &d be the Current density in the bars

Mmf for damper Winding = Ad &d = 0.15 acz

Ad = 0.15 ac Z

In practice, the area provided for damper winding is greater than the area required. Therefore

$$Ad = 0.2 acc$$

The current density in the damper bars is usually taken as 3 to 4 A/mm<sup>2</sup>.

Inorder to reduce the current induced in damper Windings by tooth ripples, the elamper winding pitch is about 20% less than stator slot pitch

No: of damper bars/pole (Nd) = Pole arc 0.8 × YSS.

The diameter of each damper bar dd is given by

$$d_d = \sqrt{\frac{a_d}{\pi/4}}$$

A 1250 KVA, 30, 60 Hz, 6600V, Salient pole alternator has the following data.

Air gap diameter = 1.6m, length of Core = 0.45m, No of poles = 20, armature ampere conductors per meter = 28,000, tratio of pole arc to pole pitch = 0.68. Stator Slot pitch = 28mm, current density in damper bars = 3 A/mm<sup>2</sup>. Design a suitable damper winding for the machine.

Crives data

 $Q_{r=1250 \text{ KVA}}, 3\phi_{1} bo H_{2}, E_{L} = 6600 \text{ V}, Salient pole$  D = 1.6 m, L = 0.45 m, P = 20, ac = 28,000, Pole are = 0.68 $Y_{ss} = 28 \text{ mm}, \delta = 3 \text{ A}/\text{mm}^{2}$ 

To find

Design of damper winding.

$$A_{d} = \frac{0.2 \times ac \times (\pi D/P)}{\delta}$$

$$A_d = \frac{0.2 \times 28,000 \times (\frac{\overline{\Lambda} \times 1.6}{20})}{20}$$

3

$$a_{d} = 58.6431 \text{ mm}^{2}$$

iv) Length of damper bar  
For large machine, 
$$Ld = L + 0.1$$
, m.  
 $Ld = 0.45 \pm 0.1$   
 $(Ld = 0.55, m)$   
V) Diameter of damper bar  
 $a_d = \frac{T d_d^2}{4}$   
 $d_d^2 = \frac{4 a_d}{T}$   
 $d_d^2 = \frac{4 x 58.6431}{T}$   
 $d_d^2 = 74.6667$   
 $d_d^2 = 8.6410, mm$ 

### Design of turboi alternators

a) Main dimensions.

$$D^{2}L = \frac{Q \ln k v n}{111 \pi^{2} n B a v a c k w x 10^{3}} \rightarrow (1)$$

where kw=kcxkd

$$kc \rightarrow pitch factor = \cos(\frac{\alpha}{2})$$

where d is the angle by which coils are short . Chorded.

Ed → dietribution factor = 
$$Sin(\frac{mB}{2})$$
  
 $m Sin(\frac{B}{2})$   
 $mg = phase Spread$   
 $B \rightarrow Slot angle = \frac{180^{\circ}}{n}$   
 $n \rightarrow Slot/pole$ .  
 $m \rightarrow Slot/pole$ .  
 $m \rightarrow Slot/pole$ ] phase.

After Substituting the values in equation  $\mathbb{O}_{1}$ we get  $D^{2}L = -m^{3}$ .

In turbo generators D is limited by peripheral Speed The For Normal designa a poripheral speed of about 120 m/s is used. The maximum peripheral speed

and the state of the second of the second state of the second state of the second state of the second state of

b) Staton design. i) Staton Blot

i) The slot pitch is normally about 25 to bomm but in the case of large turbo alternators it may even be 75 to 90 mm.

ii) humber of slot/pole/phase is usually b/w 2 to 4. but in the Case of large tarbo-alternatore 8 or 9 slots per pole per phas may be used.

il) Stator Conductors.

Esph = 4:44f & Tskus voits > ()

$$\text{Sph} = \frac{\text{Fsph}}{444 \text{fo} \, \text{kmr}} \rightarrow (2)$$

where q = BauTDL P $Z_s = 6T_{sph}$ 

For n number of parallel paths, the equation () is modified as

4.44 j 4 kws

Zs=6Tsph

Check Cond/Blot should be a integer or not if it is not a integer means makes it a integer value Zscnew) = Cond slot × Slot Te (new) = Ze (new) ph 6. (c) length of air gap ATto = ATa × SCR where armaluse mmf | pale (ATa) = 2.7 Ts Is kus armature mmf/pole (BTa) = acz [ We seen that aimpere conductor per pole =  $I_z Z_s = acz$ -'. ampere turns per pole =  $\frac{acz}{2}$ Let us assume that mmf required for the airgap is 80% of no load field mmf 8,00,000 Bglgkg = 0.8 × act × scr lg= 0.8×act×sce 8,00,000 Bg kg x2

d) Rolog design

i) Full load field mmf can be taken as twice the armatur.

$$AT_{11} = 2AT_{a} \rightarrow \bigcirc$$

where ATa = 2.7 Te Is kus or acc

ii) The field winding should be designed for a Voltage from 15 to 20 percent less than the excepter voltage.

Voltage across the field  $\operatorname{coil}(E_f) = (0.8 \text{ to } 0.85) V_e \rightarrow (2)$ 

